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Validity and limitations of linear analytical models for steel wire strands under axial loading, using a 3D FE model

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Abstract:

In the field of cable modeling, many models have been proposed to describe the mechanical behavior of simple straight strands under axial loading, and the predictions of these models have been compared to experimental data when available. However, the validity domain of these models has not been evaluated yet because the experimental results reported in the literature are very limited. This problem is addressed here, the results from nine linear elastic models of a 6+1 wire single layered strand (simple straight strand) subjected to static axial loads being compared with values from 3D finite element modeling. The analytical models are shown to give satisfactory estimations of the elastic stiffness constants for lay angles below 20°.

Keywords: Cable; Strand; Wire; Analytical model; Finite element

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Nomenclature

 R_c , R_w , R_k : core, wires and wires centerline radius

 k_r : ratio of wire radius to core radius

 α : lay angle

P: pitch length

A: cross section area

I: bending inertia modulus

J: torsion inertia modulus

 u_z , θ_z : overall axial displacement and rotation angle

 $u_{z,z},\ \theta_{z,z}$: overall axial strain and twist angle per unit length

 $u_{t,t}$: wire axial strain

 $\Delta k^{'}$, $\Delta \tau$: changes in curvature and twist per unit length in the wire

 F_z, M_z : overall axial force and torque

 f_z , m_z : non-dimensional axial force and torque

 F_b^i , F_t^i , M_b^i , M_t^i : shear force, traction force, bending moment and torque in the ith

wire

 $k_{\varepsilon\varepsilon}$, $k_{\theta\theta}$, $k_{\varepsilon\theta}$, $k_{\theta\varepsilon}$: axial stiffness coefficients of traction, torsion and couplings

 $\bar{k}_{\varepsilon\varepsilon},\bar{k}_{\theta\theta}$, $\bar{k}_{\varepsilon\theta}$, $\bar{k}_{\theta\varepsilon}$: non-dimensional axial stiffness coefficients

E, G: Young's and shear moduli

v. Poisson's ratio

 $\zeta_i, \eta_i, \lambda_i, \mu_i$: Knapp's parameters [12]

Subscripts c and w referring to core and wire characteristics, respectively

1. Introduction

Helically wound fibers or wires constitute a large class of important engineering components (Cardou and Jolicoeur [1]). There are many kinds of such structure: one is the strand, defined as a structure made up of layers of helical wires wound around a central straight wire core. A wire rope is a structure made up of layers of strands wrapped helically around a central straight strand core. It is then possible to consider wire rope as a basic component to form a new wire rope with more complex cross section.

It is well known that a major advantage of such elements is their ability to support large axial loads with comparatively small bending or torsion stiffness. These structures play an essential role in various civil engineering applications, including the prestressing of concrete, stays for guyed masts, bridging applications such as hangers for suspension bridges as well as for mooring many offshore oil platforms.

Experimental work on large diameter cables requires specific, large and expensive testing devices. Design tools allowing rapid estimation of the overall axial elastic stiffness of stranded structures are therefore essential for designers. As will be detailed below, several analytical models based on different hypotheses (purely tension wires, curved beam theory, Poisson's ratio effects, friction effects and variations in core radius) have been developed and presented in the literature. However, comparisons between these models and corresponding discussion of their validity domain is not available in the literature. Nevertheless, it may be noted that the authors have usually compared their models with the experimental results reported in the earlier works. Jolicoeur and Cardou [2] compared the results obtained by several

mathematical models available in the literature with the experimental 1+6 cables results reported by Utting and Jones [3,4], for six different lay angle (9.2°, 11°, 12.2°, 12.9°, 14° and 17°). Since the experimental results are very limited, obtaining a general conclusion is difficult.

The objective of the present study is to assess the validity domain of several analytical models for the elastic static axial (tension and torsion) behavior of a simple straight metallic strand (single-layered), consisting of six helical wires wrapped around a straight core. The influence of the assumptions of different analytical models will be discussed using comparisons with a complete three-dimensional formulation: a 3D finite element modelling has been used as a reference.

In section 2, a description of the geometry of the structures studied and the corresponding axial elastic overall behavior matrix form are given. Next, the existing mechanical models assumptions and corresponding stiffness coefficient expressions are given in section 3.

In section 4, a three-dimensional finite element model is presented and FE results are compared with experimental data. In order to determine the validity domain, the stiffness matrix components are calculated using the selected analytical models. Then, the analytical results are compared with the FE model results, considered as a reference.

2. Strand overall behavior

Let us consider a single 6+1 straight strand cable made of six helical wires with a circular cross-section wrapped around a straight core as illustrated in Figure 1. The geometry is characterized by the core radius R_c , the wires radius R_w , and the lay angle $^{\alpha}$ measured with respect to the cable z-axis. The wires centerline is then a helical curve of radius R_h :

$$R_h = R_c + R_w \tag{1}$$

It can be noted that the wire cross-sections, due to contact forces with the core, are bean-shaped but can be approximated by ellipses in the plane perpendicular to the structure z-axis (see figure 1). The pitch length denoted by P can be calculated using the following expression:

$$P = \frac{2\pi R_h}{\tan \alpha} \tag{2}$$

The axial behavior of such a structure exhibits coupling between tension and torsion due to the helical design of the wires. Thus, the elastic overall behavior can be expressed in the form:

where $u_{z,z}$ denotes the overall axial strain, $\theta_{z,z}$ the twist angle per unit length, F_z the axial force and M_z the torque. The four stiffness matrix components $k_{\varepsilon\varepsilon}$, $k_{\theta\theta}$, $k_{\theta\varepsilon}$ and $k_{\varepsilon\theta}$ are pure tensile, torsion and coupling terms respectively. Moreover, the stiffness matrix should be symmetric, as can be shown from Betti's reciprocal theorem, cf. [5].

3. Analytical models

3.1 Overview

As previously indicated, several analytical models are available to predict the mechanical behavior of isotropic cables subjected to axial loads, based on knowledge of the component material behavior and geometry of the structure. The first approaches only incorporate effects associated with tension, the bending and torsion stiffness of the wires being neglected. Such analyses have been performed by Hruska [6-8] and by Knapp [9] for a rigid core. McConnell and Zemek [10] modified this model, simply by considering the torsional stiffness of all the individual wires.

More recent and complex analytical models are based on beam theory assumptions: the wires are modelled using Love's curved beam equations [11]. Following this approach, Machida and Durelli [12] have studied the effects of the bending and torsion stiffness of individual wires on the cable stiffness matrix. Knapp [13] studied the effect of variations in core radius (due to the pressure from the layers). Costello and Philips [14] presented a general non-linear theory for a layer of

helical wound wires without core, which included the effects of radius and lay angle variations (due to Poisson's ratio effect). This model was later extended by Philips and Costello [15] and Velinsky [16] to apply to wire rope with internal wire rope cores. Then, Costello [17] presented a linearized theory including the effects of curvature and twist variations. Kumar and Cochran [18] have developed a linearized form of this theory, leading to a closed-form expression for stiffness matrix components. Kumar and Botsis [19] have extended this model to obtain the analytical expression for the maximum contact stresses induced in the multilayered strands with metallic wire core. Huang [20] studied the contact mode conditions (radial or lateral) for 1+6 cable. Local contact deformation is neglected whilst the Poisson's effect is included. It is found that radial contact seems to be the prevailing case, even when no initial gap exists between wires in the layers. Utting and Jones [3, 4] have extended the Costello's analysis to include wire flattening (contact deformation) and friction effects. The results show that these phenomena have very little effect on estimates of the global cable response.

Sathikh et al. [21] presented a closed form symmetric linear elastic model for a cable with a rigid core, using discrete thin rod theory. Labrosse [22] presented a new analytical approach to predict the overall behavior of 1+6 cables subjected to bending, tension and torsion. In this model, Poisson's ratio effect is neglected while relative motions between core and wires are considered.

Elata et al. [23] presented a new model for simulating the mechanical behavior of a wire rope with an independent wire rope core under axial loads. In contrast with previous models that consider the effective response of wound strands, this model considers the complete double-helix configuration of individual wires within the wound strand and directly relates the wire level stress to the overall load applied at the rope level. Bending and torsion stiffness of the individual wires are neglected, but the wires are subjected to loads on their lateral surface, applied by adjacent wires. Therefore, the accuracy of this model increases when the number of wires in the wire rope increases. It should be noted that its major interest is to analyze the effect of a double helix configuration which is not the case here (in the 7 wire cable geometry). Moreover, this model does not lead to closed-form equations, and therefore will not be applied in the present paper.

Another approach for multi-layered cables consists of modeling each layer as an equivalent orthotropic sheet. This homogenization method was first applied to cable modelling by Hobbs and Raoof [24], and it has been described in detail by Raoof [25] and then extended by Raoof and his associates over two decades [24-29]. The same approach can also be used by replacing each layer with a cylinder of orthotropic, transversely isotropic material [30-34]. Such homogenization approaches can be applied when the number of wires in the layer is large, but this is not the case for 1+6 structures.

In addition, several models are also available for the analysis of synthetic cables. Leech et al. [35], presented a quasi-static analysis of fiber ropes and included it in commercial software: Fiber Rope Modeller or FRM. Another model was developed by Rungamornrat et al. [36], and later extended by Beltran et al. [37] and Beltran and Williamson [38]. These models are very similar to that of Leech but they have concentrated on a damage model to take into account the degradation of rope

properties as a function of loading history. These models are also implemented in a computer program. Very recently, Ghoreishi et al. [39, 40] have developed two closed-form analytical models which can be used in sequence to analyze the synthetic cables.

In this work, the case of a simple straight strand, with linear elastic behavior, isotropic material and axial loading, is considered.

3.2 Descriptions of the analytical models

As shown previously, there are several mathematical models that are based on different assumptions. In this section, nine analytical models have been selected and are described in more detail. These models are based on different assumptions, such as:

- Axial stiffness of wire is only considered,
- Bending and torsion stiffness of wire are included,
- Poisson's ratio effect is considered,
- Inter-wire contact (relative motion between core/wire) is considered.

The equations for each model are then briefly presented in a standardized form, using the same notations as those of Ghoreishi et al. [41].

3.2.1 Hruska's model

In the models based on pure tension wire, bending and torsion stiffness of the individual wires are neglected. Using purely geometrical relations, wire axial strain, u_{tt} , can be expressed as follows:

$$u_{t,t} = u_{z,z} \cos^2 \alpha + R_h \theta_{z,z} \sin \alpha \cos \alpha$$
 (4)

where $u_{z,z}$ and $\theta_{z,z}$ represent axial strain of cable and angle of twist per unit length, respectively. The stiffness matrix components are:

$$\begin{cases} k_{\varepsilon\varepsilon} = (AE)_c + 6(AE)_w \cos^3 \alpha \\ k_{\varepsilon\theta} = k_{\theta\varepsilon} = 6(AE)_w R_h \cos^2 \alpha \sin \alpha \\ k_{\theta\theta} = (GJ)_c + 6(AE)_w R_h^2 \sin^2 \alpha \cos \alpha \end{cases}$$
(5)

where (AE) and (GJ) represent axial and torsional stiffness respectively. The core and wires characteristics are denoted using subscripts c and w respectively.

This stiffness matrix has been obtained by Hruska [6-8] and re-derived by Knapp [9] for a rigid core.

3.2.2 Model of McConnell and Zemek

McConnell and Zemek [10] modified Hruska's model by adding the sum of the torsion stiffness of all the individual wires into Equation (5)₃. The equation of $k_{\theta\theta}$ then becomes:

$$k_{\theta\theta} = (GJ)_c + 6(GJ)_w + 6(AE)_w R_h^2 \sin^2 \alpha \cos \alpha$$
 (6)

The relations for $k_{\varepsilon\varepsilon}$, $k_{\varepsilon\theta}$ and $k_{\theta\varepsilon}$ in Eqs. (5) remain identical.

3.2.3 Model of Machida and Durelli

Machida and Durelli [12] have studied the effects of the bending and torsion stiffness of individual wires on the cable stiffness matrix. They obtained the bending and twisting moments of each individual wire, then by projecting on the cable axis, and summing for all the wires, one obtains:

$$\begin{cases} k_{\theta\varepsilon} = 6 \left((AE)_{w} R_{h} \cos^{2} \alpha \sin \alpha - 2 \frac{(EI)_{w}}{R_{h}} \cos^{2} \alpha \sin^{3} \alpha + \frac{(GJ)_{w}}{R_{h}} \cos^{2} \alpha \sin \alpha (1 - 2\cos^{2} \alpha) \right) \\ k_{\theta\theta} = (GJ)_{c} + 6 \left((AE)_{w} R_{h}^{2} \sin^{2} \alpha \cos \alpha + 2 (EI)_{w} \cos^{3} \alpha \sin^{2} \alpha + (GJ) \cos^{3} \alpha (1 - 2\sin^{2} \alpha) \right) \end{cases}$$
(7)

where (EI) is the bending stiffness of the wire. The relations for $k_{\varepsilon\varepsilon}$ and $k_{\varepsilon\theta}$ in Equation (5) remain identical. Thus stiffness matrix symmetry is lost.

3.2.4 Knapp's model

Knapp [13] presented a new stiffness matrix for straight cables subjected to tension and torsion. He studied the compressibility of the central core. The equilibrium equations first include geometric non-linearity and are then linearized to give a linear stiffness matrix. This approach was primarily aimed at soft core cables.

However, thanks to a general derivation, Knapp's results can also be applied to the more rigid core cables considered in the present work.

In the present notations, stiffness matrix components can be written as:

$$\begin{cases} k_{\varepsilon\varepsilon} = (AE)_c + 6(AE)_w \cos^3 \alpha \zeta \\ k_{\varepsilon\theta} = 6(AE)_w R_h \sin \alpha \cos^2 \alpha \eta \\ k_{\theta\varepsilon} = 6 \left[(AE)_w R_h \sin \alpha \zeta - \frac{(EI)}{R_h} (\lambda + 2\zeta) (\sin^3 \alpha + \frac{(GJ)}{R_h} (1 - 2\zeta) \sin \alpha \cos^2 \alpha \right] \\ k_{\theta\theta} = (GJ)_c + 6 \left[(AE)_w R_h \eta \sin \alpha - \frac{(EI_w)}{R_h} (\mu + 2\eta - \frac{2R_h}{\tan \alpha}) \sin^3 \alpha \right] \\ + \frac{(GJ_w)}{R_h} (\frac{R_h}{\tan \alpha} - 2\mu) \sin \alpha \cos^2 \alpha \end{cases}$$

$$(8)$$

where ζ_i , η_i , λ_i and μ_i are parameters defined by Knapp in his paper, related to the geometric and elastic properties of the cables. In the present work, for steel core cable the compressibility of the central core is negligible, ζ_i and η_i are very close to $\cos^2\alpha$ and $R_h \sin\alpha\cos\alpha$, respectively, while λ_i and μ_i are nearly equal to zero (see Ref. [13]). If these limiting values are substituted into equation (8), these reduce exactly to Machida and Durelli's results.

3.2.5 Costello's model

Costello's theory [17], based on Love's curved beam equations, takes into account the effects of radius and lay angle variations, as well as wire bending and torsion moments. The equilibrium is expressed using a set of non-linear equations which is also valid for large deflection. In the case of simple straight strands, it is

assumed that change in lay angle is small. It follows that the cable axial strain $u_{z,z}$ and twist $\theta_{z,z}$, as well as changes in curvature and twist per unit length, $\Delta k'$ and $\Delta \tau$, in each wire are linearized with respect to $\Delta \alpha$. Considering the development detailed in Ref. [17], the relationships for each wire may be written as:

$$\begin{cases} u_{z,z} = u_{t,t} + \Delta\alpha \tan\alpha \\ R_h \theta_{z,z} = u_{t,t} \tan\alpha - \Delta\alpha + v \tan\alpha \frac{(R_c u_{z,z} + R_w u_{t,t})}{R_h} \\ R_w \Delta k' = -\frac{2\sin\alpha \cos\alpha}{R_h/R_w} \Delta\alpha + v \frac{R_c u_{z,z} + R_w u_{t,t}}{R_h} \frac{\sin^2\alpha}{R_h/R_w} \\ R_w \Delta \tau = \frac{1 - 2\cos^2\alpha}{R_h/R_w} \Delta\alpha + v \frac{R_c u_{z,z} + R_w u_{t,t}}{R_h} \frac{\sin\alpha \cos\alpha}{R_h/R_w} \end{cases}$$

$$(9)$$

and the local mechanical behavior of each wire is given by:

$$\begin{cases} M_t^i = (GJ)_w \Delta \tau \\ M_b^i = (EI)_w \Delta k' \\ F_t^i = (EA)_w u_{t,t}^w \end{cases}$$

$$F_b^i = M_t^i \frac{\sin^2 \alpha}{R_b} - M_b^i \frac{\sin \alpha \cos \alpha}{R_b}$$

$$(10)$$

where F_b^i , F_t^i , M_b^i and M_t^i denote shear force, traction force, bending moment and torque in the ith wire, respectively; moreover, $\Delta k'$ and $\Delta \tau$ represent changes in curvature and twist per unit length, respectively and $u_{t,t}$ is the axial strain in an outer wire. Projecting on the cable axis and summing for all the wires, one gets:

$$\begin{cases} F_z = (EA)_c \ u_{z,z} + 6 \left(F_t^i \cos \alpha + F_b^i \sin \alpha \right) \\ \\ M_z = (GJ)_c \ \theta_{z,z} + 6 \left(M_t^i \cos \alpha + M_b^i \sin \alpha + R_h F_t^i \sin \alpha + R_h F_b^i \cos \alpha \right) \end{cases}$$

$$(11)$$

To obtain the stiffness matrix components, the first two equations $(9)_{1-2}$ allow the axial strain of the wire, $u_{t,t}$, and the variation of lay angle, $\Delta \alpha$, to be expressed as a function of global strain of the strand, $u_{z,z}$ and $\theta_{z,z}$. It follows that the changes in curvature and twist per unit length, $\Delta k'$ and $\Delta \tau$, are also expressed as a function of these strains. Then, the local mechanical behavior of each wire can be expressed as a function of global strain of the strand, $u_{z,z}$ and $\theta_{z,z}$. Finally, by substituting equations (10) into equations (11), the global behavior of the strand can be obtained.

It should be noted that, recently, a closed-form expression of the cable stiffness matrix coefficients, has been given by Ghoreishi [42]. Moreover, in this model, the stiffness matrix symmetry is lost.

3.2.6 Model of Kumar and Cochran

Kumar and Cochran [18] have linearized the equations of Costello's model, and arrived at a closed-form expression for the stiffness matrix. In the present notation, the stiffness matrix components may be written as:

$$\begin{cases} k_{ss} = (AE)_c + 6(AE)_w \cos \alpha & (1 - (1 + v) p \sin^2 \alpha) \\ k_{s\theta} = 6(AE)_w R_h p \cos^2 \alpha & \sin \alpha \end{cases}$$

$$\begin{cases} k_{\theta s} = 6 \left((AE)_w R_h \sin \alpha - \frac{(EI)_w}{R_h} \sin \alpha (1 - v_f \cos^4 \alpha + (1 + v) (q - 1 + v_f \cos^4 \alpha) \right) \\ k_{\theta \theta} = (GJ)_c + 6(EI)_w q \cos \alpha \end{cases}$$

$$(12)$$

where p and q are given by:

$$\begin{cases} p = (1 - v_1) \left[1 - (1/4) \left(\frac{R_w}{R_h} \right)^2 (1 + v_f \cos 2\alpha) \sin^2 \alpha \right] \\ q = (1 - v_1) \left[4 \left(\frac{R_h}{R_w} \right)^2 \sin^2 \alpha + 1 - \sin^4 \alpha - v_f \cos^4 \alpha \cos 2\alpha \right] \end{cases}$$
(13)

in this relation v_1 and v_f are obtained as follows:

$$\begin{cases} v_1 = v(R_w/R_h)\sin^2\alpha \\ v_f = v/(1+v) \end{cases}$$
 (14)

The stiffness matrix symmetry is also lost.

3.2.7 Ramsey's model

Ramsey [43, 44] has proposed a mathematically rigorous approach based on differential geometry. Constitutive equations (10)₁₋₂, in which internal moments are linearly related to the curvature and twist variations $\Delta k'$ and $\Delta \tau$, are replaced by:

$$\begin{cases}
M_t^i = (GJ)_w (\Delta \tau + \tau u_{t,t}) \\
M_b^i = (EI)_w (\Delta k' + k' u_{t,t})
\end{cases}$$
(15)

For axial loads, as indicated by Jolicoeur and Cardou [2], most equations reduce to Costello's model except for the foregoing differences. To evaluate their influence, equations (15)₁₋₂, have been used instead of equations (10)₁₋₂ in Costello's model.

3.2.8 Sathikh's model

Sathikh et al. [21] concentrated on the lack of symmetry in the earlier models. They developed a symmetric linear elastic model for simple straight strands, using Ramsey's theory and discrete thin rod theory. A simple straight strand with a rigid core and having only core to wire contact has been analyzed taking into account the wire tension, twist and bending together, for its response due to axisymmetric loads (traction-torsion). Following modification to the present notation, Sathikh's relations may be written as:

$$\begin{cases} k_{\varepsilon\varepsilon} = (AE)_{c} + 6 \left((AE)_{w} \cos^{3} \alpha + \left[(GJ)_{w} \sin^{2} \alpha + (EI)_{w} \cos^{2} \alpha \right] \frac{\sin^{4} \alpha \cos \alpha}{R_{h}^{2}} \right) \\ k_{\varepsilon\theta} = k_{\theta\varepsilon} = 6 \left((AE)_{w} R_{h} \cos^{2} \alpha \sin \alpha \right) + \left[(GJ)_{w} \cos^{2} \alpha - (EI)_{w} (1 + \cos^{2} \alpha) \right] \frac{\sin^{3} \alpha \cos^{2} \alpha}{R_{h}}$$

$$k_{\theta\theta} = (GJ)_{c} + 6 \left[(AE)_{w} R_{h}^{2} \sin^{2} \alpha \cos \alpha \right] + (GJ)_{w} \cos^{7} \alpha + (EI)_{w} \sin^{2} \alpha \cos \alpha \left(1 + \cos^{2} \alpha \right)^{2}$$

$$(16)$$

3.2.9 Labrosse's model

Recently Labrosse [22] presented a new analytical approach to predict the global response of a cable subjected to bending, tension and torsion. In this theory the wires are considered as curved beams as presented by Love [11]. The integrated

stresses over the cable cross section and the inter-wire efforts have been presented as a function of the generalized strains of the cable and derivation of the inter-wire slippage.

For the axial loads (tension and torsion), the stiffness matrix components become:

$$\begin{cases} k_{\varepsilon\varepsilon} = (AE)_{c} + 6(AE)_{w} \cos^{3} \alpha \\ k_{\varepsilon\theta} = k_{\theta\varepsilon} = 6(EA)_{w} R_{h} \sin \alpha \cos^{2} \alpha \\ k_{\theta\theta} = (GJ)_{c} + 6\left((EA)_{w} R_{h}^{2} \sin^{2} \alpha \cos \alpha + (GJ)_{w} \cos^{5} \alpha + (EI)_{w} \sin^{2} \alpha \cos \alpha (1 + \cos^{2} \alpha)\right) \end{cases}$$
(20)

The selected analytical models with their principal features (i.e., behavior of wire, change in geometry due to Poisson's effect, symmetric matrix and closed form solution), are summarized in table 1.

4. Numerical study

Emphasis is placed on the linear elastic global behavior of a simple isotropic straight steel strand under small strain. To evaluate their validity domains, the matrix stiffness coefficients obtained from the analytical models presented in the previous section are compared with those calculated from a complete 3D formulation: the overall behavior of the strand has also been calculated using a 3D FE model detailed hereafter. The results are expressed in a general non-dimensionalized form and for strand lay angle α varying between 2.5 and 35°, which cover the usual practical values. The investigation of local effects (i.e. stress distribution in the cross section, contact deformation, local plastic yielding) is not included in this paper.

4.1 FE approach

4.1.1 Background

With the development of finite element methods during the last decades, together with the development of computer capacity, certain authors have used the finite element approach to analyze the mechanical behavior of cables. Carlson [45], modeled the wires by bar elements as well as the connections between the wires. Cutchins [46], used solid elements (6 node) to model the wires, their connections were simulated with springs. Nawrocki [47] presented a one dimensional curved beam element for analyzing the simple straight strand (1+6) and later Nawrocki and Labrosse [48] studied the relative motions between core and the wires, using this element. They have shown that the inter-wire pivoting and the inter-wire sliding govern the cable global response, for axial and bending loads, respectively. It should also be underlined that some FE studies have focussed on local phenomena: Jiang et al. [49, 50] investigated the stress distribution within the wires, in a simple straight strand as well as in a three layered straight strand, using a concise 3D finite element model with prescribed displacement field. One can mention also the work of Wehking and Ziegler [51] devoted to the detailed computation of rope stresses. In the same way, Messager and Cartraud [52] have developed an homogenization procedure: applying the rigorous asymptotic development theory, a slice of a strand can be studied using a 3D FE model free of any displacement assumption. In the present paper, a complete 3D FE model is presented as a reference to determine the validity domain of existing analytical models.

4.1.2 Description of the model

The simple straight strand cable structure has been computed using the COSMOS FE code, Version 2.0. The geometry of the core has been obtained by a linear z-axis extrusion. Each wire has been generated by the extrusion of a circular surface along an helical curve corresponding to the centroidal line of the wire. As shown in figure 2, each wire section consists of twelve finite elements (six 15-node and six 20-node solid elements). It should be noted that the number of elements along the strand z-axis varies with lay angle because of the pitch length variations in this range of lay angle, for example, pitch length of 769.4 mm and 41.9 mm for lay angle of 2.5° and 35° respectively.

The boundary conditions are defined as follows:

- One end-section of the cable is fully clamped.
- At the other end-section, the wires and the core nodes are linked using rigid body elements connected to a master node located at the cross-section center.
- On this master node, the transverse loads and bending moments are prescribed to be zero.

The "6+1" cable structure considered is illustrated in figure 2. It is assumed that the wires do not touch each other (no wire/wire contact) in the undeformed state so contact is only present radially (core/wire contact). Moreover, Huang [20] has shown that even if the wires are in radial contact in the undeformed state, they tend to separate while loading.

In addition, in a preliminary study, the influence of contact conditions between the wires and the core (wire/core contact) has been examined for two limit cases: sliding without friction and merging. These contact conditions have been applied for nodes situated on the helical lines of contact between core and wires. The results obtained show that the static overall behavior is not sensitive to these modeling hypotheses. Indeed, the relative difference between these limit cases, for stiffness matrix components, is less than 1.5% (for lay angle varying between 2.5 and 35). Therefore, for the results detailed thereafter in section 4.2, the contact nodes have been merged. However, it should be mentioned that while the friction effect plays a small role in global stiffness behavior of such structures, the effect of friction on the long-term performance and durability of a structure under cyclic loading can be significant.

The stiffness coefficients are computed in four successive steps corresponding to different loading conditions in tension and torsion on the master node:

 $k_{\varepsilon\varepsilon}$: an axial load F_z is applied while the rotation angle θ_z is prescribed to zero.

 $k_{\theta\theta}$: a torque M_z is applied whereas the axial displacement u_z is locked

 $k_{\varepsilon\theta}$: axial load F_z with a free rotation angle θ_z .

 $k_{\theta\varepsilon}$: torque M_z with a free axial displacement u_z .

The cable strains are calculated from the master node axial displacement and rotation. Preliminary tests, performed for models of lengths between two and ten pitches, have demonstrated that the overall axial response is not influenced by end

effects. The results detailed in next paragraphs have been obtained for an FE model consisting (for each value of the lay angle α) of only two pitch lengths. It should be underlined that the discrepancies observed on the coupling terms $k_{\mathcal{E}\theta}$ and $k_{\theta\mathcal{E}}$ have been found to be negligible for all FE analyses performed, the differences were always less than 0.4%.

It should be mentioned that, the numerical convergence has been verified with respect to:

- Number of elements per pitch
- Axial length of the model (number of pitch in the model)

4.1.3 Comparisons with experiments

To demonstrate the accuracy of FEM results, the results of the 3D FE model are compared with the experimental data, reported by Utting and Jones [3, 4]. In the experimental study material modulus determined from tests on wires and core has been measured to be 197.9 GPa, cf. [3]. In the FEM model isotropic material is considered, the Young modulus E and the Poisson's ratio have been set to 197.9 GPa and 0.3, respectively.

First, the stiffness matrix components are computed using a 3D FE model for three strands which were studied by Utting and Jones in Refs. [3, 4]. Then, numerical results are compared with the experimental data. It should be noted that the experimental results do not allow a comparison of the stiffness matrix components directly, so different parameters will be compared for different tests as will be explained below.

Utting and Jones [4] have performed an experimental study on simple straight steel strands with six different lay angles (for all of them: $R_c = 1.97$ and $R_w = 1.865$ mm). Nominally fixed-end tests, as well as the tests with free and partially restrained ends were carried out. The rotations were measured during a load range of 40 kN for free end tests. This rotation can also be calculated by the following expression:

$$\theta_{z,z} = \frac{F_z}{k_{\varepsilon\theta} - \frac{k_{\varepsilon\varepsilon} k_{\theta\theta}}{k_{\theta\varepsilon}}}$$
, for free end condition.

For nominally fixed-end tests, the torque generated was recorded. Table 2 compares measured values with FE results for nominally fixed end and free end conditions.

It should be noted that, for nominally fixed end tests in which the rotation, theoretically, should be zero, but during the tests, some small rotations, $\theta_{z,z}$, were reported and attributed by the authors to resin deformation at the end fittings of the cables, namely, 0.064, 0 and 0.058 rad/m for specimen number 1, 4 and 6, respectively, see Utting and Jones [3, 4]. Thus, numerical values (in table 2) have been obtained by solving equation (3) for M_z , with the given values of F_z and $\theta_{z,z}$, using the stiffness matrix components which were obtained from the 3D FE model. For the fixed end condition, the twisting moment, M_z , can be expressed as follows:

$$M_z = \frac{k_{\theta\varepsilon}}{k_{\varepsilon\varepsilon}} (F_z - k_{\varepsilon\theta} \theta_{z,z}) + k_{\theta\theta} \theta_{z,z}$$
, for fixed end condition.

The intermediate case where both an axial load and twisting moment are imposed was also considered. Axial strain, $u_{z,z}$, and twist per unit length, $\theta_{z,z}$, of the strand are then measured. If the axial load is fixed, say 40 kN, and the torque is given two values M_z^1 and M_z^2 , one gets two sets of results $(u_{z,z}^1, \theta_{z,z}^1)$ and $(u_{z,z}^2, \theta_{z,z}^2)$. Defining $\Delta M_z = M_z^2 - M_z^1$, $\Delta u_{z,z} = u_{z,z}^2 - u_{z,z}^1$, $\Delta \theta_{z,z} = \theta_{z,z}^2 - \theta_{z,z}^1$, one can calculate slopes $\Delta M_z / \Delta u_{z,z}$ and $\Delta M_z / \Delta \theta_{z,z}$. These slopes can also be given by the following expressions:

$$\begin{cases} \frac{\Delta M_z}{\Delta u_{z,z}} = k_{\theta \varepsilon} - \frac{k_{\varepsilon \varepsilon} k_{\theta \theta}}{k_{\varepsilon \theta}} \\ \frac{\Delta M_z}{\Delta \theta_{z,z}} = k_{\theta \theta} - \frac{k_{\theta \varepsilon} k_{\varepsilon \theta}}{k_{\varepsilon \varepsilon}} \end{cases}$$

Table 3 presents such slopes measured by tests and prediction by the FE model.

In general, tables 2 and 3, indicate a good correlation between the FE model and experiment, at least for the range of lay angles considered here.

So, once the numerical reliability of the FEM model was checked - convergence has been reached on the cable behavior - the modeling assumptions - elliptical shape approximation for the wire cross-section, sticking contact between the wires and the core – have been validated through the comparison with experimental results. Experimental results are available only for lay angle values of 9.2, 12.2 and 17°, but the FEM model is not restricted to those lay angles. Due to the lack of published experimental data (particularly for larger lay angles), hereafter the 3D FE results will be considered as a reference solution.

4.2 Results and discussion

In order to perform general comparisons for lay angle varying between 2.5 and 35°, the stiffness matrix component values obtained both using the analytical formulations and the 3D FE model have been expressed using a non-dimensional form. Considering Eq. (3), this may be written as:

where the reduced loading components are defined by:

$$\begin{cases} f_z = \frac{F_z}{E \cdot \pi \cdot R_h^2} \\ m_z = \frac{M_z}{E \cdot \pi \cdot R_h^3} \end{cases}$$
 (8)

The non-dimensional components of the stiffness matrix are then expressed as follows:

$$\begin{cases}
\bar{k}_{\varepsilon\varepsilon} = \frac{k_{\varepsilon\varepsilon}}{E \cdot \pi \cdot R_h^2} \\
\bar{k}_{\theta\theta} = \frac{k_{\theta\theta}}{E \cdot \pi \cdot R_h^4} \\
\bar{k}_{\varepsilon\theta} = \frac{k_{\varepsilon\theta}}{E \cdot \pi \cdot R_h^3} \\
\bar{k}_{\theta\varepsilon} = \frac{k_{\theta\varepsilon}}{E \cdot \pi \cdot R_h^3}
\end{cases}$$

$$(9)$$

As can be shown from any analytical model, the post-processed \bar{k}_{ij} axial stiffness coefficients turn out to be dependent on only two geometrical characteristic values: lay angle α and ratio of wire radius to core radius $k_r = R_w/R_c$. Moreover, these dimensionless coefficients \bar{k}_{ij} , are independent of the elasticity modulus E of the constitutive material; the only material term characterizing the isotropic material behavior is the Poisson's ratio value. For the present numerical preliminary study, this coefficient has been taken to be 0.3. Ghoreishi [42] performed a sensitivity study for these geometrical characteristics (α and k_r), which demonstrates that the accuracy of the models is not sensitive to k_r for the usual practical values. Therefore, the results obtained using the different models are compared only for lay angle variation.

Figure 3 details the evolutions of the non-dimensional axial stiffness, $\bar{k}_{\varepsilon\varepsilon}$, obtained by different analytical models as well as 3D finite element model, versus lay angle.

These curves show that all the analytical models studied here produce only two distinct series of results: there is a coincidence of the results for the models of Hruska [8], Machida and Durelli [12], McConnell and Zemek [10], Sathikh et al. [21] and Labrosse [22] (the differences are negligible and only one curve has been plotted for these 5 models), and another series of results for Kumar and Cochran [18], Ramsey [44] and Costello [17]. It may be noted that for the second group of models, the change of geometry due to the Poisson's effect is taken into account. However, the non-dimensional axial stiffness remains a little higher than those of FE model. In addition, for the small lay angles ($\alpha \le 15^{\circ}$), the results of all the models are very close (within 1.7% of each other), and a good agreement is obtained with the FE model

(always less than 5% difference). The difference between these groups of models increases slightly with increasing lay angle. Moreover, the models which take into account the change of geometry due to the Poisson's effect, give the results closest to FE model values, see figures 3, 4 and 5.

Figure 4 shows the non-dimensional torsion term versus lay angle. These results were obtained by selected analytical models and the 3D finite element model. As shown in figure 4, the model of Hruska [8] gives results appreciably lower than the others, because the torsional stiffness of the wires is neglected. This effect is more important when the lay angle is small. This can be easily explained by the fact that an increase in lay angle also increases considerably the torsional stiffness, while having little or no effect on the $(GJ)_{w}$ or $(EI)_{w}$ terms. This effect is especially noticeable for a single layered strand because in a multi-layered strand, the torsional stiffness of the wire is negligible when compared to the term contributed from axial load.

All the other analytical models provide very similar results, the difference increasing slightly with increasing lay angle. In addition, for $\alpha \le 20^{\circ}$, the agreement between the analytical models (except Hruska's model) and FE model are encouraging. Beyond this value, a significant difference appears.

Let us note finally that the simple modification to the model of Hruska made by McConnell and Zemek [10], gives results which are quite comparable to those of much more complex models, as has already been noted by Jolicoeur and Cardou [2].

The evaluation of the non-dimensional coupling terms $\bar{k}_{\varepsilon\theta}$ and $\bar{k}_{\theta\varepsilon}$, versus lay angle, are illustrated in figures 5 a) and b) respectively. As underlined previously, only the models of Hruska, Sathikh and Labrosse lead to symmetrical matrix terms $(\bar{k}_{\varepsilon\theta} = \bar{k}_{\theta\varepsilon})$.

For $\bar{k}_{\varepsilon\theta}$, we can make the same qualitative analysis as in the case of the axial stiffness. The results of the analytical models are grouped in two series, whose difference increases slightly with the lay angle. For a small angle ($\alpha \le 15^{\circ}$), the agreement with the FE model is very good. Beyond this value, a significant difference appears (up to 45%). With regard to $\bar{k}_{\theta\varepsilon}$, (see figure 5b), the differences between the results of the analytical models are more significant. In addition, when the lay angle increases, it appears that the models which take into account the change of geometry due to the Poisson's effect, also give the results closest to those of the FE model.

5. Concluding remarks

The objective of this paper is to determine the validity domain of analytical models of steel strands. The elastic behavior of a simple 6+1 geometry having lay angles up to 35° and subjected to axial static load has been studied. First, the equations of nine selected models have been rewritten with the same notation, and their principal assumptions have been briefly presented. It was shown that the 3D FE model can be used as a reference for the cable behavior since its results are very close to existing experimental data.

Then, the results of selected models are compared with those of 3D FE model calculations, using general dimensionless stiffness coefficients.

Considering these results, the \bar{k}_{ij} values are in good agreement, (except for the model of Hruska which clearly underestimates the torsional stiffness), for lay angle values below 20°. The relative differences between analytical and FEM results are then always less than 10 %. Nevertheless, the discrepancies grow rapidly and significantly, except in the case of the axial stiffness, for α values beyond 20°. These differences then become significant (up to 45 %) especially for the coupling terms $\bar{k}_{e\theta}$ and $\bar{k}_{\theta e}$. It is possible that in this range of lay angle curved beam theory is too approximate for the helical wire analysis. From the point of view of accuracy, it appears that for lay angles below 15°, there is a negligible difference between analytical models. For lay angles beyond 15°, the analytical models which take into account the change of geometry due to the Poisson's effect, Kumar and Cochran [18], Ramsey [44] and Costello [17], provide the results closest to those of 3D FE model. However, these models present the disadvantage of having a nonsymmetrical stiffness matrix.

In conclusion, the analytical models presented here appear quite satisfactory for the estimation of the elastic stiffness coefficients of 6+1 metallic strand structures with lay angles less than 20°. This is an important conclusion for designers and has allowed the domain of validity of models for these structures to be quantified in terms of finite element results, instead of relying on experimental data. For larger lay angles the use of 3D FE modeling would be recommended.

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Figures

- Figure 1: Geometry of a simple straight strand or 1+6 structure.
- Figure 2: Mesh examples.
- Figure 3: axial stiffness, $\bar{k}_{\it \epsilon\it \epsilon}$, versus lay angle $\it lpha$.
- Figure 4: non-dimensional torsion term, $\bar{k}_{\theta\theta}$, versus lay angle α .
- Figure 5: dimensionless coupling terms versus α , a) $\bar{k}_{\,\varepsilon\theta}$ and b) $\bar{k}_{\,\theta\varepsilon}$

Tables

Table 1: selected analytical models with their principal features.

Models	Behavior of wires			Poisson's	Symmetric	Closed-form	
1,10,0018	Te. To. Be.		Be.	effect	matrix	solution	
HRU	X				X	X	
MAC	X	X	X			X	
McC	X	X				X	
KNA	X	X	X			X	
KUM	X	X	X	X		X	
RAM	X	X	X	X			
SAT	X	X	X		X	X	
COS	X	X	X	X		X	
LAB	X	X	X		X	X	
HRU: Hruska [7]			RAM: Ramsey [43]				
MAC: Machida and Durelli [11]				SAT: Sathikh [20]			
McC: McConnell and Zemek [9]				COS: Costello [16]			
KNA: Knapp [12]				LAB: Labrosse [21]			
KUM: Kumar	and Coch	ran [17]					

Table 2: comparison between 3D FE model and experimental data (reported in Refs. [3, 4]), for nominally fixed and completely free end conditions.

Specimen	Stiffness	matrix co	mponents	M_z (N.m) (fixed end)		$\theta_{z,z}$ (rad/m)	
	obtained	by 3D FE 1	model			(free end)	
	$k_{\varepsilon\varepsilon}$	$k_{sg} = k_{gs}$	$k_{ heta heta}$	Test	FEM	Test	FEM
	(kN)	(N.m)	(N.m ²)				
$1 (\alpha = 17^{\circ})$	1.32e3	12.4e3	26.74	34.4	36.7	2.4929	2.50889
$4 (\alpha = 12.2^{\circ})$	1.43e3	9.72e3	19.46	26	27.2	2.2049	2.12266
$6 (\alpha = 9.2^{\circ})$	1.48e3	7.64e3	15.86	18.8	19.9	1.62025	1.73139

Table 3: comparison between 3D FE model and experimental data reported in Refs. [3, 4] for partially restrained ends.

Specimen	Stiffness	matrix co	mponents	$\Delta M_z / \Delta u_{z,z}$		$\Delta M_z / \Delta \theta_{z,z}$		
	obtained by 3D FE model			$(10^3 \mathrm{N.m})$		(N.m ²)		
	$k_{\varepsilon\varepsilon}$	$k_{sg} = k_{gs}$	$k_{\theta \theta}$	Test*	FEM	Test*	FEM	
	(kN)	(N.m)	$(N.m^2)$					
$1 (\alpha = 17^{\circ})$	1.32e3	12.4e3	26.74	16.2	15.94	13.7	15.03	
$4 (\alpha = 12.2^{\circ})$	1.43e3	9.72e3	19.46	20.1	18.84	12.8	12.84	
$6 (\alpha = 9.2^{\circ})$	1.48e3	7.64e3	15.86	22.9	23.10	11.6	11.92	
* these results were given by Utting and Jones [3, 4], in graphical form.								

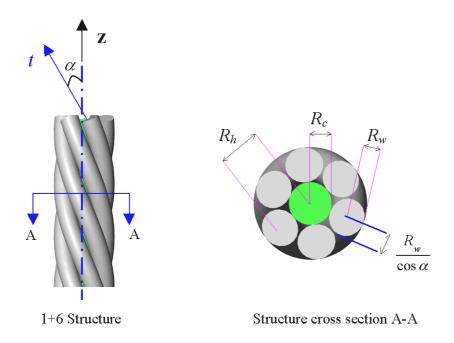


Figure 1: Geometry of a simple straight strand or 1+6 structure.

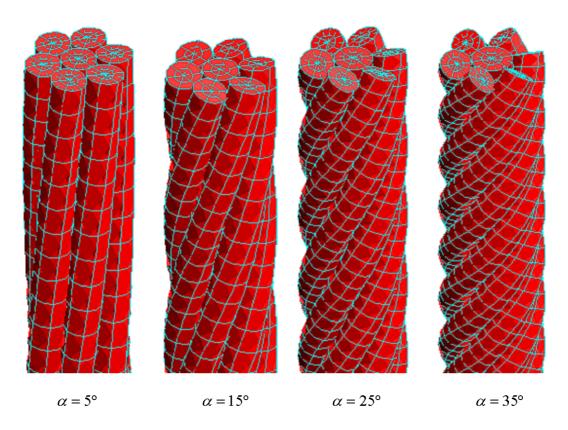


Figure 2: Mesh examples.

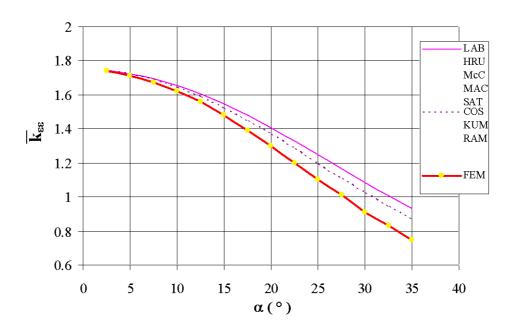


Figure 3: axial stiffness, $\bar{k}_{\it ee}$, versus lay angle α .

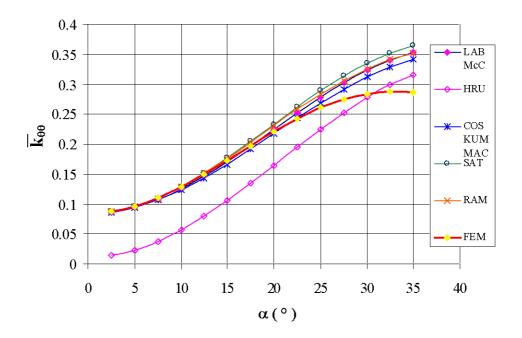
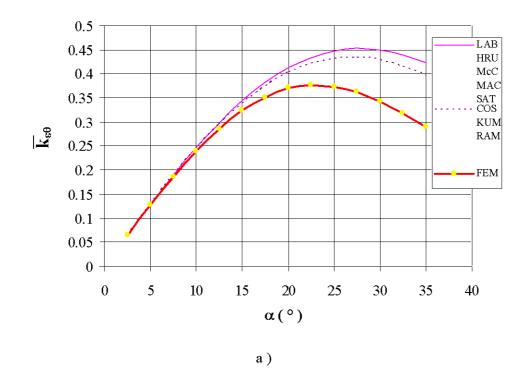


Figure 4: non-dimensional torsion term, $\,\bar{k}_{\,\theta\theta}\,,$ versus lay angle $\,\alpha$.



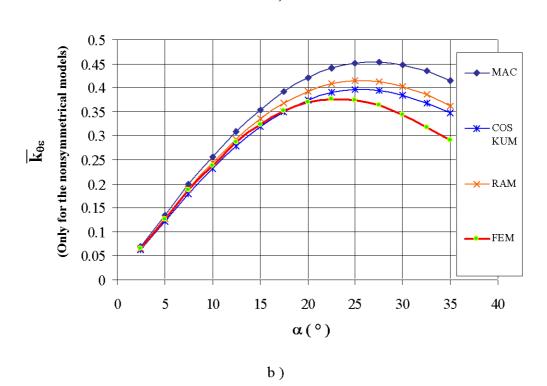


Figure 5: dimensionless coupling terms versus α , a) $\bar{k}_{\varepsilon\theta}$ and b) $\bar{k}_{\theta\varepsilon}$

