

## Photochemistry on the Space Station-Aptamer Resistance to Space Conditions: Particles Exposure from Irradiation Facilities and Real Exposure Outside the International Space Station

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### Abstract :

Some microarray-based instruments that use bioaffinity receptors such as antibodies or aptamers are under development to detect signatures of past or present life on planetary bodies. Studying the resistance of such instruments against space constraints and cosmic rays in particular is a prerequisite. We used several ground-based facilities to study the resistance of aptamers to various types of particles (protons, electrons, neutrons, and carbon ions) at different energies and fluences. We also tested the resistance of aptamers during the EXPOSE-R2 mission outside the International Space Station (ISS). The accumulated dose measured after the 588 days of this mission (220 mGy) corresponds to the accumulated dose that can be expected during a mission to Mars. We found that the recognition ability of fluorescently labeled aptamers was not significantly affected during short-term exposure experiments taking into account only one type of radiation at a time. However, we demonstrated that the same fluorescent dye was significantly affected by temperature variations (-21 degrees C to +58 degrees C) and storage throughout the entirety of the ISS experiment (60% of signal loss). This induced a large

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variability of aptamer signal in our analysis. However, we found that >50% of aptamers were still functional after the whole EXPOSE-R2 mission. We conclude that aptamer-based instruments are well suited for in situ analysis on planetary bodies, but the detection step requires additional investigations.

**Keywords** : Astrobiology, Cosmic rays, Biochip, Aptamers

## 56 **I. Introduction**

57 In the context of planetary exploration, space agencies call for the development of miniaturized  
58 techniques to search for traces of extant or extinct life. In particular, the use of instruments  
59 based on bio-affinity receptors, like antibodies or aptamers, might be very efficient. In this  
60 technology, bio-affinity receptors are fixed to a solid surface to specifically capture their target.  
61 A signal is displayed when receptor-target recognition events occur.

62 Antibody-based biochips have been extensively studied in proteomic and functional genomics  
63 and their scope has been extended to the field of astrobiology. Indeed, due to the benefits of  
64 high throughput and miniaturization, and small volume consumption, antibody-based biochips  
65 have been proposed to detect biomarkers, especially organic ones, in the search for  
66 extraterrestrial life (Parro *et al.*, 2005; Le Postollec *et al.*, 2007; Parro *et al.*, 2008; Martins *et*  
67 *al.*, 2011; Parro *et al.*, 2011b; Sims *et al.*, 2012; McKay *et al.*, 2013; Smith *et al.*, 2014). Space  
68 is a hazardous environment, with high fluxes of ionizing radiations, from primary particles of  
69 galactic cosmic radiation (GCR) and solar energetic particles (SEP), to secondary particles  
70 produced by the interaction of primary particles with environment materials. One main concern  
71 relies on the resistance of biochips to the cumulative effects of these radiations. In recent years,  
72 studies have been performed to evaluate the resistance of antibody reagents to space constraints,  
73 especially regarding the effect of specific particles on antibody binding performances (Le  
74 Postollec *et al.*, 2009a; Le Postollec *et al.*, 2009b; Baqué *et al.*, 2011a; de Diego-Castilla *et al.*,  
75 2011; Baqué *et al.*, 2017; Coussot *et al.*, 2017). To complement ground-based radiation studies,  
76 a short term mission was performed on the BIOPAN-6 low-earth orbit platform to demonstrate  
77 the effects of cumulative radiations on the antibodies' ability to bind to their respective antigens  
78 (Derveni *et al.*, 2012; Derveni *et al.*, 2013). For the absorbed radiation doses during the 12 days  
79 of exposure, the freeze-dried antibodies packaged into laser-cut glass fiber pads were unaffected

80 with regard to their recognition performances in the condition of the experiment. Furthermore,  
81 in the frame of the BiOMAS project (Biochip for Organic Matter Analysis in Space) and part  
82 of the Photochemistry on the Space Station (PSS) experiment, biochip models, in which  
83 antibodies were immobilized onto a surface, have been installed outside the International Space  
84 Station (ISS) on the EXPOSE-R2 platform, for a real long-term exposure (more than 18  
85 months) to spatial constraints (Vigier *et al.*, 2013; Cottin *et al.* 2015, Cottin *et al.* 2017). Our  
86 recent results show that our biochip models resist to an 18 months extra-vehicular mission  
87 (Coussot *et al.*, 2018b). This work, in combination with the ground-based radiation studies,  
88 permits to assess the antibodies' ability to bind to their respective antigens even after a long-  
89 term exposure to the real space environment. These experiments indicated that antibody-based  
90 instruments seem well suited for the search for organic matter in *in situ* planetary bodies  
91 samples.

92 Other target-binding reagents with specificity and binding affinity rivalling with those of  
93 antibodies are aptamers. Aptamers are short, single-stranded nucleic acids, able to bind from  
94 small molecular weight molecules up to proteins. Aptamers have demonstrated some interesting  
95 advantages over antibodies: production at low cost, specificity against small and non-  
96 immunogenic molecules, thermal stability, and ability to maintain their structures over repeated  
97 cycles of denaturation/renaturation (Song *et al.* 2012). From diagnostics to food safety or  
98 therapeutics, aptamers have shown their interests in arrays and biosensors development (Song  
99 *et al.* 2012; Dong *et al.*, 2014; Rozenblum *et al.*, 2016; Gotrik *et al.*, 2016). The first  
100 investigation for nucleic acid aptamers ability to maintain their binding performances under  
101 simulated cosmic radiations, in particular 2 MeV protons effects, was carried out by Baqué *et*  
102 *al.* (2011b). The results revealed that, under the protons flux and energy tested, there were no  
103 deleterious effects on the binding affinity of the irradiated DNA aptamer. So far, no more

104 investigations on the effect of high-energy particles on aptamer performances have been  
105 reported.

106 In the present study, we first present a set of experiments carried out these last seven years in  
107 ground-based particles accelerator facilities to study the effect of different types of incident  
108 particles (protons, electrons, neutrons, and carbon ions) at different energies and fluences, on  
109 the ability of a model aptamer to recognize its target. The second part of the paper discusses the  
110 results obtained after a long time exposure to real space constraints of aptamer samples in the  
111 framework of the PSS project (Cottin *et al.*, 2015).

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## 114 II. Materials and methods

### 115 II.1 Chemical, reagents and materials

116 *L*-tyrosinamide (*L*-Tym), fluorescein dye (F), sodium bicarbonate (NaHCO<sub>3</sub>, × 99,5%), sodium  
117 carbonate (Na<sub>2</sub>CO<sub>3</sub>, × 99,5%), and Tris(hydroxymethyl) aminomethane (Tris× 99,8%) were  
118 purchased from Sigma Aldrich (Saint-Quentin Fallavier, France). NaCl and MgCl<sub>2</sub> of analytical  
119 grade were obtained from Chimie-Plus Laboratoires (Bruyères de Pouilly, France) and Panreac  
120 Quimica (Barcelona, Spain), respectively. The aptamer anti-*L*-Tym containing 49 nucleotides  
121 and modified in its 3′ end by fluorescein (#4551724, MW 15 860.5 g/mol) was provided by  
122 Eurogentec (Angers, France). This aptamer was chosen as an aptamer model, it has been  
123 reported as 49merYm3′F in the text, and had the following sequence

124 5′-AATTCGCTAGCTGGAGCTTGGATTGATGTGGTGTGTGAGTGCGGTGCCC-F-3′ (4  
125 different batches were used: batch 1 for ground-based neutrons, electrons, and protons  
126 experiments, batch 2 for ground-based <sup>12</sup>C experiments, batch 3 for ISS mission; batch 4 for  
127 post-flight experiments). The ultrapure water (18.2 MΩ) was obtained from a Purite Still Plus  
128 water purification system (Thame, UK). Other chemicals are analytical grade and used as  
129 received. Nunc MaxiSorp<sup>®</sup> polystyrene 96 well plates were obtained from VWR (France), and  
130 were used as sample containers during the freeze-drying step, and throughout the laboratory  
131 radiation exposures. In case of ISS experiment, Corning DNA Bind<sup>®</sup> 8-well strip plate with  
132 N-hydroxysuccinimide modified surface (NHS-wells) were provided by Sigma Aldrich (Saint-  
133 Quentin Fallavier, France). NHS-wells were manufactured by Air Liquide (Sassenage, France)  
134 to fit perfectly with the shape and size of EXPOSE-R2 closed cells provided by the French  
135 Space Agency (CNES, Toulouse) (Vigier *et al.*, 2013). The home-designed NHS-wells had a  
136 diameter of 8.4 mm with a 7.1 mm height. Teflon cap were specially manufactured by Air  
137 Liquide to close the well. (**Figure 1a**)

138

139 ***II.2 Freeze-drying and sample preparation***

140 Particle irradiation effects were evaluated on both freeze-dried 49merYm3øF aptamer and  
141 fluorescein dye. One hundred microliters of 49merYm3øF aptamer or fluorescein solution at  
142 500 nM in ultrapure water (dilution by 1/40 from a stock solution freshly prepared at 20 µM  
143 from aptamer commercial vial) were put into MaxiSorp<sup>®</sup> wells (commercial or home-designed  
144 well format) and freeze-dried according to a protocol optimized for the freeze-drying of small  
145 volume of reactants (Coussot *et al.*, 2018a). Briefly, MaxiSorp<sup>®</sup> wells were put into a home-  
146 made aluminium holder, and liquid nitrogen was added to freeze the samples. The aluminium  
147 holder was then placed into the central part of a freeze-dryer with a top-press device (Christ  
148 Alpha 264 from Martin Christ GmbH, Germany). Freeze-drying was performed overnight  
149 (condenser temperature -85°C, vacuum 0.05 mbar). The chamber of the freeze-dryer was filled  
150 with nitrogen gas before closing the sample-containing aluminium holder, which kept sealed  
151 hermetically the samples and sheltered them from the moisture and light until its opening. The  
152 opening of the aluminium holder was done in a glove box under a controlled atmosphere of  
153 helium (10% He) in argon (Ar) provided by Air Liquide (Sassenage, France) to maintain a  
154 relative humidity (RH) level of 10 to 15%. During our experiments, the RH and air temperature  
155 were controlled using a thermohygrometer Testo 605-H1 (Testo, France). RH was 12±3% and  
156 air temperature was 22.8±2.3°C. After opening the aluminium holder, freeze-dried samples  
157 were sealed within a plastic bag using a vacuum sealing device FoodSaver<sup>®</sup> (Fischer  
158 Scientific, France) and stored at 4°C in the dark until running laboratory irradiation  
159 experiments.

160 In case of ISS experiments, 150 µL of carbonate buffer (pH 9.4, 0.1M) were added into the  
161 home-designed NHS-wells to deactivate reactive sites. The incubation took place overnight at

162 room temperature. After successive water washings, 100  $\mu\text{L}$  of 49merYm3 $\phi$ F  
163 aptamer/fluorescein solution at 500 nM, freshly prepared in ultrapure water, were pipetted into  
164 each well. The freeze-drying procedure was run as described above. After opening the  
165 aluminium holder in the above controlled atmosphere, freeze-dried samples were capped and  
166 directly transferred into the CNES closed cells using a tool vacuum suction pen (FFQ939 from  
167 Mayf's Online Shop, China). The CNES closed cells were then screwed with a final tightening  
168 of 0.7Nm using a torque screwdriver TorqueVario $\text{\textcircled{R}}$ -S (Wiha, Germany) (**Figure 1b**) (Vigier  
169 *et al.*, 2013). Soldering of the CNES cells and their integration to the sample carriers were done  
170 according to CNES internal procedures.

### 171 ***II.3 Experiments on ground-based particles accelerator facilities***

172 The composition of cosmic rays is dominated by protons, electrons, alpha particles, and a small  
173 fraction of heavy ions (like carbon, oxygen and up to iron ions). Interactions of these primary  
174 particles with matter produce secondary particles like electrons and neutrons. So, experiments  
175 on ground-based facilities (producing protons, electrons, neutrons and carbon ions) have been  
176 performed to study the impact of GCR and SEPs on the affinity of the 49merYm3 $\phi$ F aptamer  
177 towards its *L*-Tym target. The set of experimental conditions is summarized in **Table 1**, and  
178 detailed in Baqué *et al.*, 2017. These are considered as short-term exposure experiments taking  
179 into account only one type of radiation at a time to study the potential effects on the  
180 49merYm3 $\phi$ F aptamer-*L*-Tym bindings. All experiments were repeated at least 3 times per  
181 irradiation conditions in order to have 3 titration curves per analytical conditions. 6 replicates  
182 were prepared for fluorescein dye. Freeze-dried samples were irradiated directly in their sealed  
183 bags. The influence of the bag during irradiation experiments was simulated with GEANT4  
184 Monte Carlo toolkit and demonstrated to be negligible (Baqué *et al.*, 2017). Controls are the  
185 reference samples that have been freshly prepared the day of analysis: they did not undergo

186 freeze-drying process, neither transportation nor storage at 4°C. We previously verified that  
187 freeze-drying does not induced any effects on binding curves nor in  $K_d$  value. So, controls  
188 (freshly prepared references) can be used in place of lyophilized-non-irradiated samples for  
189 ground-based experiment interpretations. Analyses were done by using fluorescence  
190 polarization (FP) assay for aptamers as described in section II.7, and in direct fluorescence for  
191 fluorescein dyes as described in section II.6.

#### 192 ***II.4 Experiment outside the International Space Station (ISS)***

193 Simulating radiation environment in Earth facilities is highly complicated, since it is almost  
194 impossible to include all constraints encountered during an entire space mission all together  
195 (long duration storage, transportation, take-off/landing shocks, thermal constraints, and  
196 cumulative effects of cosmic rays particles). For that purpose, 11 samples (7 ground controls,  
197 4 exposed samples also denoted as ISS samples) took part in the  $\delta$ Biochip in PSS experiment $\delta$   
198 (Vigier *et al.*, 2013; Cottin *et al.*, 2015) during the EXPOSE-R2 mission. These samples were  
199 simultaneously prepared, within the same freeze-drying batch, as that described in the  
200 paragraph II.2. The 4 exposed CNES cells were spread on the 2 exposure levels of the Tray 3  
201 of the EXPOSE carrier dedicated to the PSS experiment. Disposition of the 2 CNES cells on  
202 the sample carrier were the same for the upper and the lower level of the sample carrier. Launch  
203 to ISS (July 23<sup>th</sup>, 2014) and storage inside ISS was at ambient temperature (22-25°C on average)  
204 (Rabbow *et al.*, 2017). Then, EXPOSE-R2 platform was placed outside the ISS on the Universal  
205 platform D on August 18<sup>th</sup> with opening of the valves on August 20<sup>th</sup>, and removal of the UV  
206 shield in October 22<sup>nd</sup>, 2014. On February 3<sup>rd</sup> 2016 trays were covered and brought back inside  
207 the ISS. Thus, the exposed samples spent 566 days outside the ISS over the 588 days of the  
208 mission (1 year, 8 months).

209 Into each CNES cell, two thermoluminescent detectors (TLDs) were placed at the bottom of  
210 the cell, below the samples. These passive dosimeters were analyzed at the end of the mission  
211 (for detailed methods of analysis see Berger *et al.*;2012; 2015) and revealed that the absorbed  
212 dose by our aptamer samples was about 220 mGy for the 588 days of the mission. This absorbed  
213 dose is consistent with doses expected for a mission to Mars. Indeed, considering Geant4  
214 simulations in Le Postollec *et al.* (2009a), this absorbed dose (220mGy) corresponds to 8  
215 months travel and 6 months at Mars surface. Considering doses measured by Curiosity rover  
216 (Hassler *et al.*, 2014), this dose corresponds to a mission to Mars with 8 months travel and 18  
217 months at the surface.

218 During their extravehicular exposition, the samples on Tray 3 were submitted to temperatures  
219 varying between -20.9°C and 57.98°C (Rabbow *et al.*, 2017). Accurate measurements of the  
220 CNES exposed cells temperature were provided by the temperature control interface of the  
221 Planetary and Space Simulation Facilities (Microgravity User Support Center (MUSC),  
222 <http://www.musc.dlr.de/expose-r-2/>; **Figure 2a**). On March 2<sup>nd</sup> 2016, EXPOSE-R2 Tray 3  
223 landed on Earth. During transit from Baïkonur cosmodrome to Moscow, and from Moscow to  
224 German Aerospace Center (DLR, Germany) a recording of temperatures was done, the  
225 temperatures oscillated between 20°C and 24°C. Then from DLR to LISA (Laboratoire  
226 Interuniversitaire des Systèmes Atmosphériques, Paris Est-Créteil) (Rabbow *et al.*, 2017), and  
227 Paris to Montpellier a controlled chamber at 4°C was used during transportation in the summer  
228 period. De-integration of the exposed CNES cells from the sample carriers was done according  
229 to MUSC/DLR/ESA/CNES internal procedures. On June 24<sup>th</sup> 2016, the soldered joint of all the  
230 CNES cells (including ground references) was removed, using a mechanical lathe and a  
231 manufactured CNES tool, in the mechanics department of the Montpellier University (**Figure**

232 **1c)**. Samples were immediately sealed in a FoodSaver<sup>®</sup> bag (Fischer Scientific, France) and  
233 stored in the dark at 4°C until analysis with FP assays.

### 234 *II.5 EXPOSE-R2 mission ground references*

235 During the same period, experiments have been performed on ground to control the long-term  
236 behaviour of the 49merYm3øF aptamer upon storage but not exposed to space constraints. Two  
237 ground CNES cells were stored at the French Space Agency (CNES, Toulouse) by maintaining  
238 an accurate temperature control at 3.9±0.8°C over that period. These samples were named as  
239 øCNES 5°Cö samples. Three ground CNES cells were kept in DLR (Cologne, Germany): 2  
240 ground CNES cells were stored at 5°C (øDLR 5°Cö samples) and 1 ground CNES cell  
241 underwent the same thermal history as ISS samples (øDLR ê Tö sample) (see above, **Figure**  
242 **2a**). In addition, the behaviour of 2 additional ground samples was studied at DLR combining  
243 long time storage, varying thermal environment as ISS ones, and UV radiations (øDLR  
244 ê T+UVö samples) (Rabbow *et al.*, 2017). Calibrated Bentham 150 double monochromators  
245 instrument equipped with a SOL2000 lamp (Dr. K.Hönle GmbH, Martinsried, Germany) was  
246 used to provide a spectrum similar to that of sunlight. All of these 7 ground CNES cells were  
247 brought back to Montpellier to be handled simultaneously with the exposed CNES cells during  
248 all the desoldering process and 4°C storage (section *II.4*) before running FP assays (section  
249 *II.7*). However, data of DLR ê T and DLR ê T+UV samples have not been added to our analysis  
250 due to the lack of precision of the applied temperatures.

### 251 *II.6 Additional ground thermal cycling experiments*

252 The 49merYm3øF aptamer and fluorescein dye were freeze-dried at 500 nM in ultrapure water  
253 as described in section *II.2*. Thermal cycling experiments were carried out on freeze-dried  
254 samples in their sealed bags in a temperature test chamber (Vötsch VT 4004). Two independent

255 thermal variation experiments were done: one with a sharp rise in temperature (5°C/min) until  
256 a one hour plateau at 80°C (referred as "80°C peak" in the text), and a long cycling period  
257 (178.8 hours or 7.45 days) mimicking thermal variations of exposed cells (**Figure 2b**) with an  
258 amplitude of about 70°C (referred as "long cycle" in the text). After reconstitution of freeze-  
259 dried product with 120 µL of deionized water, 100 µL were pipetted into black 96-well half-  
260 area microplates from Greiner Bio-One (Courtaboeuf, France) to measure fluorescence  
261 intensity of the signal (RFS). Fluorescence measurements were done on a Tecan® Infinite®  
262 F500 microplate reader (Mannedorf, Switzerland) having excitation and emission filters at 488  
263 nm and 525 nm, respectively. A blank signal with neither aptamer nor fluorescein dye was used  
264 to evaluate noise value (RFN). Results are presented as signal to noise (S/N) ratio equals to  
265 RFS/ RFN.

## 266 *II.7 Aptamer-based fluorescence polarization (FP) assay*

267 For all experiments carried out in ground-based particles accelerator facilities and the overall  
268 ISS mission, a structure-switching aptamer assay based on a fluorescence polarization (FP)  
269 approach (**Figure 3**) was used to evaluate the binding ability of the 49merYm3øF aptamer  
270 towards its *L*-Tym target. The approach, described by Ruta *et al.* (2009) and Perrier *et al.*  
271 (2010), is based on monitoring the fluorescence anisotropy change ( $r$ ) of an aptamer labelled  
272 by a single fluorescent dye at its 3øend. Fluorescence anisotropy change  $r$  was calculated as  
273 the difference between the measured anisotropy ( $r$ ) of the labelled aptamer-target complex, and  
274 the anisotropy in absence of target ( $r_f$ ), which was prepared by replacing the *L*-Tym target  
275 solution by deionized water. The optimized binding buffer for FP assay consisted of 10 mM  
276 TrisøHCl, pH 7.5, 10 mM MgCl<sub>2</sub> and 50 mM NaCl (Zhu *et al.*, 2011). The aptamer solutions  
277 (500 nM) were heated at 80°C for 5min for denaturation then left to stand at room temperature  
278 in the dark for 30min. 100 µL of the renatured aptamer solutions were deposited into black 96-

279 well half-area microplates containing the increasing concentrations of *L*-Tym  
280 (0.5/1/2.5/5/7.5/10or12.5/25.0/40.0or50.0/80.0/200.0 $\mu$ M) for measurements on Tecan's  
281 Infinite® M1000 PRO (Tecan, France). Excitation was set at 485 $\pm$ 20 nm, and emission was  
282 collected with 535 $\pm$ 25nm bandpass filters.

## 283 *II.8. Data interpretation and statistical analysis*

284 For a 1:1 stoichiometry, the measured anisotropy ( $r$ ) can be linked to the apparent dissociation  
285 constant  $K_d$  via the following relation:

$$286 \quad r = \frac{r_f K_d + r_b c}{K_d + c} \quad \text{Eq (1)}$$

287 where  $r_f$  is the anisotropy in the absence of target,  $r_b$  the anisotropy of maximally target-  
288 associated aptamer, and  $c$  the concentration of free target. For a limiting aptamer probe  
289 concentration, the total concentration of target ( $c_T$ ) in the reaction system approximates the free  
290 target concentration  $c$ . The nonlinear regression of the  $r$  versus  $c$  ( $\acute{e}$   $c_T$ ) plots, where  $r_b$  and  $K_d$   
291 constituted the adjustable parameters, was achieved using the Table curve 2D software (Systat  
292 Software GmbH, Erkrath, Germany).

293 To evaluate cosmic rays effects on freeze-dried exposed aptamers, control samples were  
294 prepared with an aptamer probe that did not undergo freeze-drying process or irradiation  
295 (reference sample). Binding affinity data for exposed samples and control samples were  
296 compared using the relative dissociation constant value (relative  $K_d$ ) expressed as follow:

$$297 \quad \frac{K_{d(\text{irradiated})}}{K_{d(\text{control})}} \quad \text{Eq (2)}$$

298 Relative  $K_d$  values are reported in **Table 2**.

299 For the ISS experiments, relative  $K_d$  were estimated from freeze-dried ground samples stored  
300 at 5°C in DLR facility during all the duration of the mission. öFlight samplesö corresponded

301 to ISS samples that undergo all cumulative effects of the mission (flights, extravehicular  
302 exposure, and transportation). Relative  $K_d$  values are reported in **Table 3**.

303 All data are reported as the mean  $\pm$  the standard deviation (SD) from at least four replicate  
304 experiments. Statistical significance of the assays was determined using Student's  $t$ -test  
305 ( $p=0.05$ ).

306

### 307 **III. Results**

308 We first determined a reference value of the affinity of the aptamer used to its target. In the  
309 various FP assays we performed, the concentration of 49merYm3øF aptamer was constant  
310 (500 nM) and its target *L*-Tym increased in concentrations from 0.5 to 200 µM (10 levels of  
311 concentrations, see experimental part). Binding data of 49merYm3øF aptamer to *L*-Tym  
312 were globally fit to four parameter logistic regressions permitting the calculation of  $K_d$   
313 ( $1.54 \pm 0.14$  µM, aptamer batches 1, 2, 3, n=14), which is comparable with previous published  
314 data (Ruta *et al.*, 2009).

315

#### 316 ***III.1. Irradiation effects on aptamers during ground-based experiments***

317 To minimize experimental error, all assays were performed at least in triplicate. Affinity  
318 curves do not reveal significant differences in anisotropy change  $r$  (**Figure 4**). The  
319 corresponding relative  $K_d$  values were calculated (*Eq. 2*), and reported in **Table 2**. The aim  
320 of the present study, according to the particle types, fluences, energies and exposure  
321 durations, is to determine whether irradiated samples undergo damaging effects on the  
322 aptamer leading to a decrease in its ability to bind to its target or leading to a decrease of the  
323 detection signal. Although no effect was detected with  $^{12}\text{C}$  ions at high energy (**Figure 4a**),  
324 (statistically identical  $K_d$  values), differences in  $\hat{e}r$  were observed with other particles. For  
325 instance, in **Figure 4b**, a slight difference was observed in  $\hat{e}r$  between the two irradiated  
326 samples with electrons. However, analysis of the relative  $K_d$  for both irradiated samples  
327 shows that there were statistically identical ( $p > 0.05$ ) compared to the corresponding  
328 reference. This suggests that the recognition capabilities of the 49merYm3øF aptamer were  
329 not affected by the irradiation process even when using higher fluence and irradiation  
330 duration. The same phenomenon was observed with neutrons: the higher fluence  $F_{n1}$  induced

331 a larger decrease in anisotropy change  $r$  (**Figure 4c**). The relative  $K_d$  value corresponding  
332 to irradiated sample with  $(F_{n1}; E_n; D_{n1})$  radiation conditions (which presents higher  
333 variability), stayed statistically identical to the reference one. Consequently, we  
334 hypothesized that high fluences of electrons and neutrons might have deleterious effects on  
335 the fluorescein dye. To verify that point, we run a Student's  $t$ -test on fluorescence intensities  
336 obtained for fluorescein dye before and after irradiations. It revealed that significant  
337 differences could be observed with up to 40% loss in fluorescence signal for high fluence  
338  $F_{n1}$  for neutrons and fluence  $F_{p1}$  at both energies  $E_{p1}$  and  $E_{p2}$  for protons. So, potential  
339 degradation of fluorescein dye might occur with higher fluence. Thus, after exposure to  $(F_{e1};$   
340  $E_e; D_{e1})$  radiation with electrons beam and  $(F_{p1}; E_{p1})$  et  $(F_{p1}; E_{p2})$  radiation with protons  
341 beam, fluorescence anisotropy changes might not be due to alteration of binding but rather  
342 suggest some alterations of the fluorescein dye emitting properties. In addition, with protons,  
343 cumulative effects were observed. Indeed, the results of FP measurements after protons  
344 irradiation showed a small discrepancy between the reference sample and the 4 irradiated  
345 ones (**Figure 4d**). Although  $K_d$  values were slightly smaller than that of the reference (**Table**  
346 **2**), they remained statistically identical. All radiation configurations tested led to a similar  
347 anisotropy change  $r$  that could suggest there was no important damage in the structure of  
348 the 49merYm3øF aptamer that consequently could alter the binding to its target  $L$ -Tym.  
349 Compared to our previous published data (Baqué *et al.*, 2011b), showing that 2MeV protons  
350 had no effect on the binding affinity of irradiated 49merYm3øF aptamer neither on the  
351 fluorescein dye, the present data might suggest that higher energies of protons (25 MeV/50  
352 MeV) alter the fluorophore.

353

354

355

### 356 *III.2. ISS experiments*

357 Since radiation gradients were observed on the EXPOSE platform in previous ISS missions  
358 (Berger *et al.*, 2012; 2015), the 4 exposed aptamers (flight samples) were spread on the 2  
359 exposure levels of the Tray 3 of the EXPOSE carrier dedicated to the PSS experiment, but with  
360 the same disposition on the upper and lower level of the sample carrier. However,  
361 measurements performed with passive dosimeters during the EXPOSE-R2 mission revealed, a  
362 posteriori, that the absorbed dose difference between cells at upper and lower levels was  
363 extremely low (about 3 mGy for the whole mission corresponding to 1.4% of the total). Some  
364 TLDs stacks composed of eleven TLDs included into Tray 3 structure bring explanations on  
365 this very low difference. As can be seen in **Figure 5**, absorbed dose decreases when shielding  
366 increases but after  $\sim 0.75 \text{ g/cm}^2$  of shielding, the absorbed dose measured remains quasi stable  
367 and the dose decrease evolves very slowly even if shielding is significantly increased. As the  
368 estimated shielding of TLDs placed into cells on the upper level is about  $0.82 \text{ g/cm}^2$ , it explains  
369 the very low absorbed dose difference observed between both levels.

370 Calculation of  $K_d$  from independently and freshly prepared reference samples was  $1.58 \pm 0.12$   
371  $\mu\text{M}$  (aptamer batch 4), which is comparable with the above reference value ( $1.54 \pm 0.14$ ).  
372 However,  $K_d$  values from the 8 mission samples appeared to be on average 1.5-2 times higher.  
373 We previously verified that freeze-drying induced no effects on binding curves nor in  $K_d$  value.  
374 However, storage stability analyses of freeze-dried 49mer Ym30F aptamer (batches 1&2),  
375 carried out at  $4^\circ\text{C}$  for 20 days and 7 months, showed differences in anisotropy change  $r$  for  
376 the different storage times and aptamer batches (data not shown). Consequently, to only take  
377 into account the effects encountered during the ISS mission (flights, extravehicular exposure,  
378 and all transportations during the mission) relative  $K_d$  were calculated using the freeze-dried  
379 ground samples stored at  $5^\circ\text{C}$  in DLR facility during all the duration of the mission. The binding

380 curves and the corresponding relative  $K_d$  values are reported in **Figure 6** and **Table 3**,  
381 respectively. We distinguished two groups of affinity curves. Group1 concerns the highest  
382 measured anisotropy change  $r$  with both ground controls stored at 5°C (DLR and CNES)  
383 during the mission. Group 2 represents the flight samples. We note that the ðlower trayð samples  
384 seem to have a different behaviour than the flight samples exposed in the ðupper trayð.

385 Based on these observations, an effect due to the temperature variations during the mission can  
386 be invoked. To test this hypothesis, post-flight ground thermal cycling experiments were carried  
387 out on the 49merYm3øF aptamer and the fluorescein dye over a 80°C short peak of temperature  
388 and a long cycle mimicking thermal variations of flight samples (see *section II.6* of the  
389 experimental part and **Figure 2b**). S/Ref values were  $74.5\pm 15.1$ ,  $62.1\pm 23.5$ , and  $59.0\pm 14.9$  for  
390 the freeze-dried fluorescein reference, the 80°C peak, and long cycle, respectively. The aptamer  
391 S/Ref values were  $23.3\pm 5.9$ ,  $23.5\pm 10.4$ , and  $70.6\pm 20.8$  for the reference, the 80°C peak, and  
392 long cycle, respectively. Consequently, there is no great impact of thermal variation upon the  
393 fluorescence properties of the freeze-dried dye. On the contrary, significant differences were  
394 observed with the 49merYm3øF aptamer, with the same order of S/Ref for the long cycle  
395 ( $70.6\pm 20.8$ ) as for the reference of the free dye ( $74.5\pm 15.1$ ). These both values are statistically  
396 identical ( $p>0.05$ ). One possible explanation could be that a long cycle affected the photo-  
397 chemical features of the fluorescein dye when attached to aptamer and consequently induced  
398 both a decrease in the fluorescence anisotropy and a higher variability in all signals (Perrier *et*  
399 *al.*, 2009). Dye played a preponderant role in the performance of the FP assays (Perrier *et al.*,  
400 2018).

401  $K_d$  values of flight ðupper & lower trayð samples are not statistically different from ground  
402 ðDLR 5°Cð and ðCNES 5°Cð controls. Consequently, based on the above considerations, our

403 results showed that aptamer recognition capabilities were not affected during the overall flight  
404 mission or that the degradations are too small to be detected by our analysis protocol.

#### 405 **IV. Conclusions**

406 During the last years, we used various facilities producing protons, electrons, neutrons and  
407 heavy ions to test whether cosmic particles could induce changes in aptamer binding events.  
408 However, simulating space radiation environment in Earth facilities remains impossible due to  
409 the great number of variables. So, we also carried out a real space exposure of aptamer samples  
410 outside the ISS during the EXPOSE-R2 mission.

411 The results presented in this paper suggest that cosmic radiation has no significant effect on the  
412 aptamers recognition ability. On the contrary, repeated temperature cycling seems to alter the  
413 mobility of its fluorescein dye. This effect results in a lowering of the recognition step signal  
414 that could be interpreted as a loss of recognition, or as the absence of the target, or as an  
415 alteration of the fluorescent dye properties. In all cases, it alters the limit of detection of the  
416 biochip.

417 In our context, since we controlled the concentration of the target and we obtained  $K_d$  values  
418 similar to the reference value, we attribute the lowering of signal as a modification of  
419 fluorescent dye mobility. So, we suggest conducting further studies to improve the detection  
420 step using another fluorescent dye or another detection method.

421

#### 422 **Acknowledgments**

423 The authors would like to thank the French national space agency (CNES) for financial support  
424 (05/2182/00-DCT094). The authors also thank all staff from the Louvain-la-Neuve cyclotron  
425 facility, from the Bergonié Institut, and the staff from the Istituto Nazionale di Fisica Nucleare  
426 of Catania for their assistance during irradiation experiments. The authors thank Jean-Louis  
427 Kergueme from the Mechanics Department of the Montpellier University for his assistance

428 during the desoldering of the CNES cells, and Dr. Sonia Khier for her assistance that has led to  
429 good progress during the analysis of samples.

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601 **Figures**

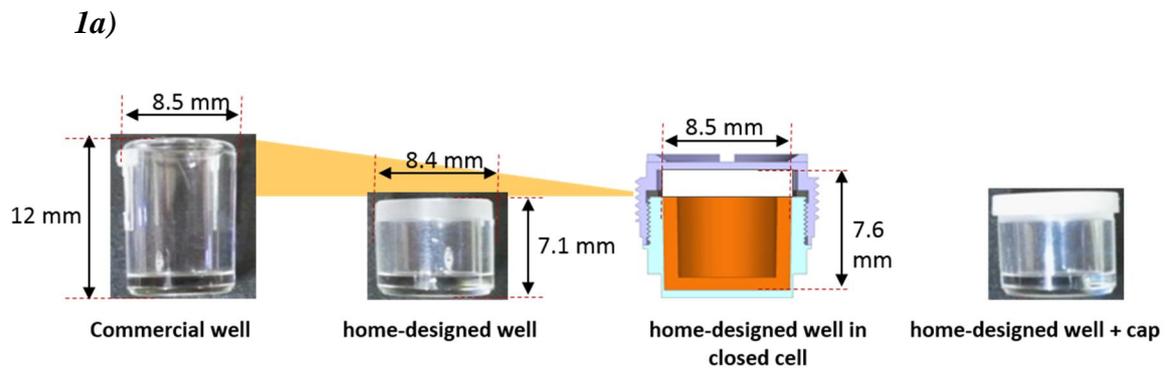
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603 **Figure 1**

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608 *1b)*

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620 *1c)*

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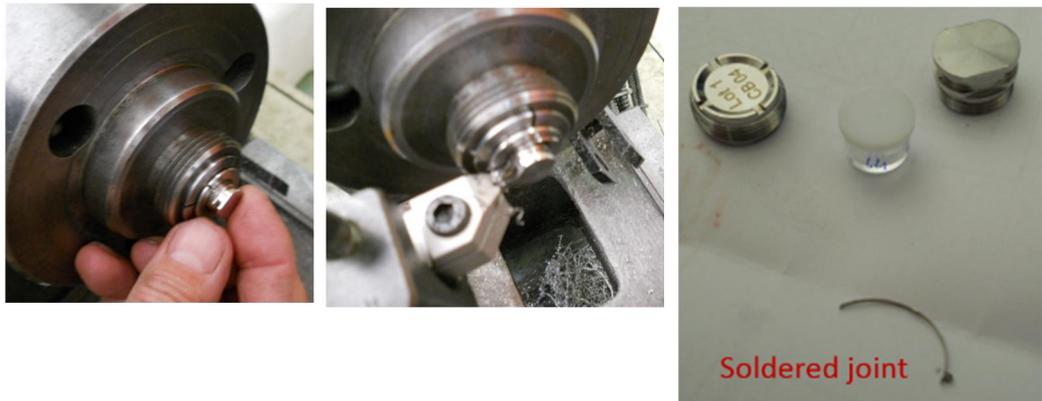
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**Figure 1 caption:**

*1a)* Design and dimension of the home-designed well and its cap.

*1b)* Incorporation of capped well into CNES cells under controlled atmosphere and final screwing of closed CNES cells using a torque screwdriver.

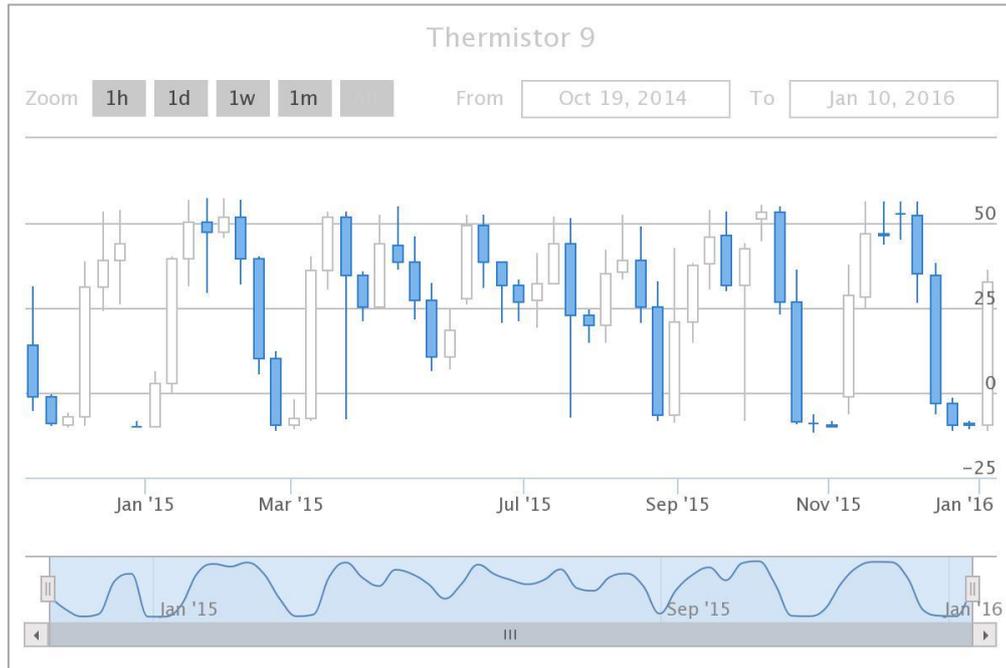
*1c)* Post-flight opening of the CNES cells to remove the soldered joint.

640 **Figure 2**

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2a)

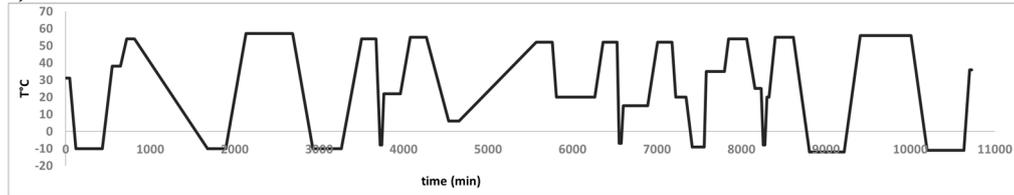


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2b)



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649 **Figure 2 caption:**

650 2a) Overview of the EXPOSE-R2 temperature profiles for tray 3 on thermistor 9 sensor

651 during the EXPOSE-R2 mission (from October 19, 2014 to January 10, 2016,

652 <http://www.musc.dlr.de/expose-r-2/>).

653 2b) Ground thermal cycling experiments mimicking thermal CNES cells exposure outside the

654 ISS but in accelerated time: one hour of EXPOSE-R2 mission was converted into one minute

655 to carry out this experiment in lab, and temperature amplitude was similar (referred as long

656 cycle in the text).

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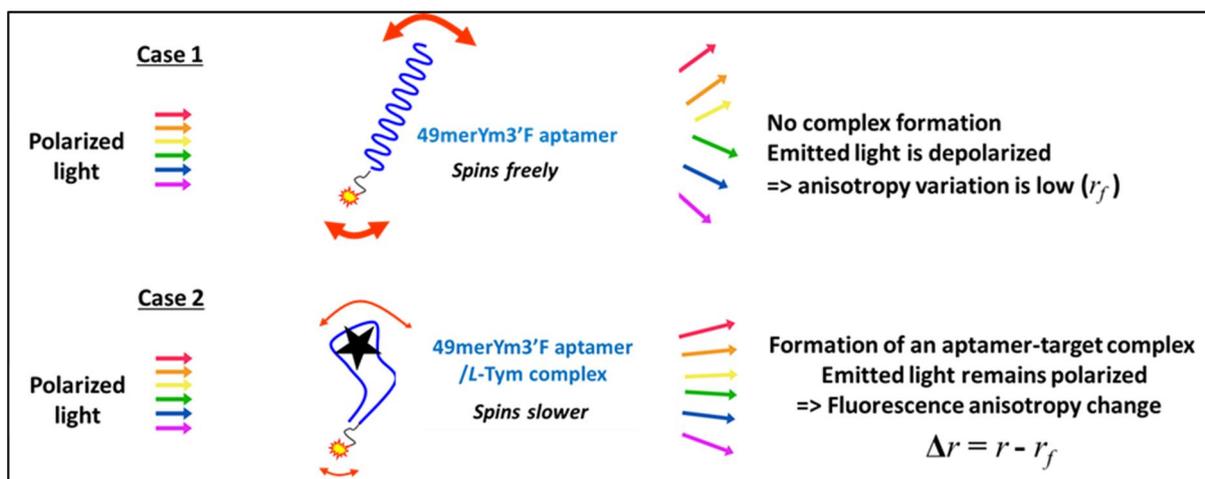
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665 **Figure 3**



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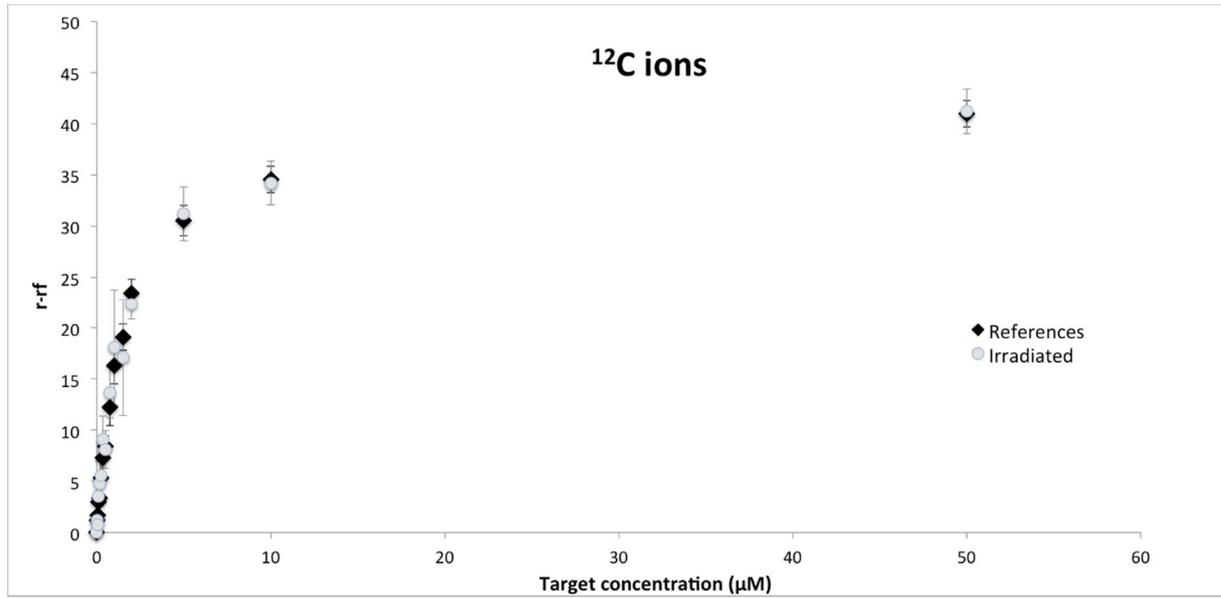
668 **Figure 3 caption:** Principle of the fluorescence polarization (FP) assays. A fluorescence  
669 anisotropy increase is observed upon the formation of a labelled aptamer-target complex.

670 **Figure 4**

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672 **4a)**

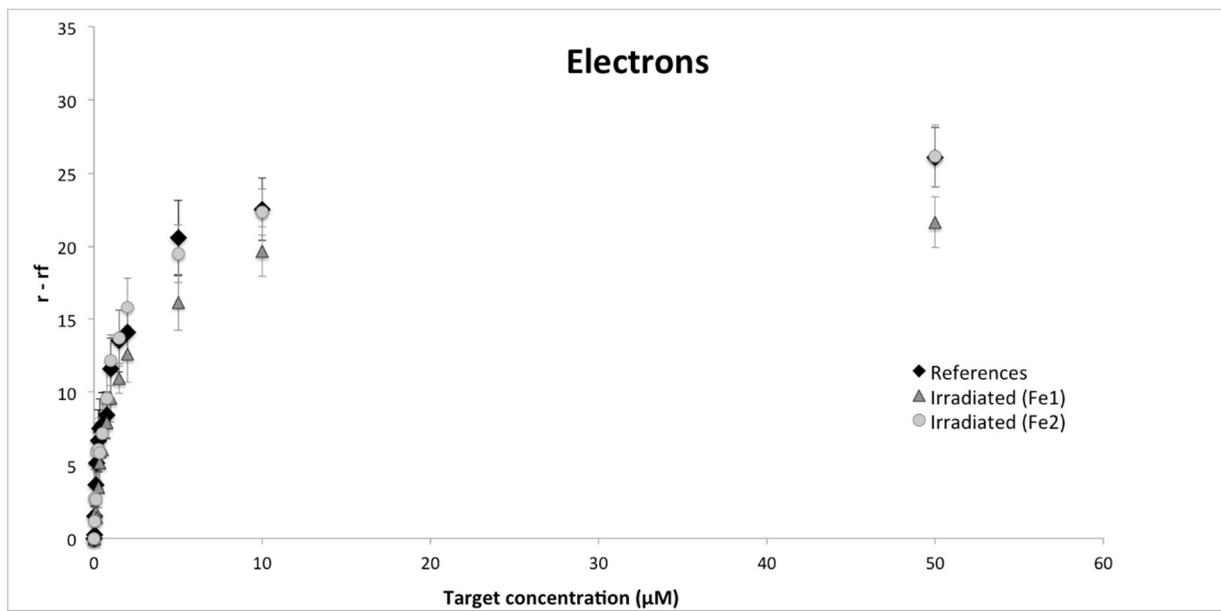
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676 **4b)**



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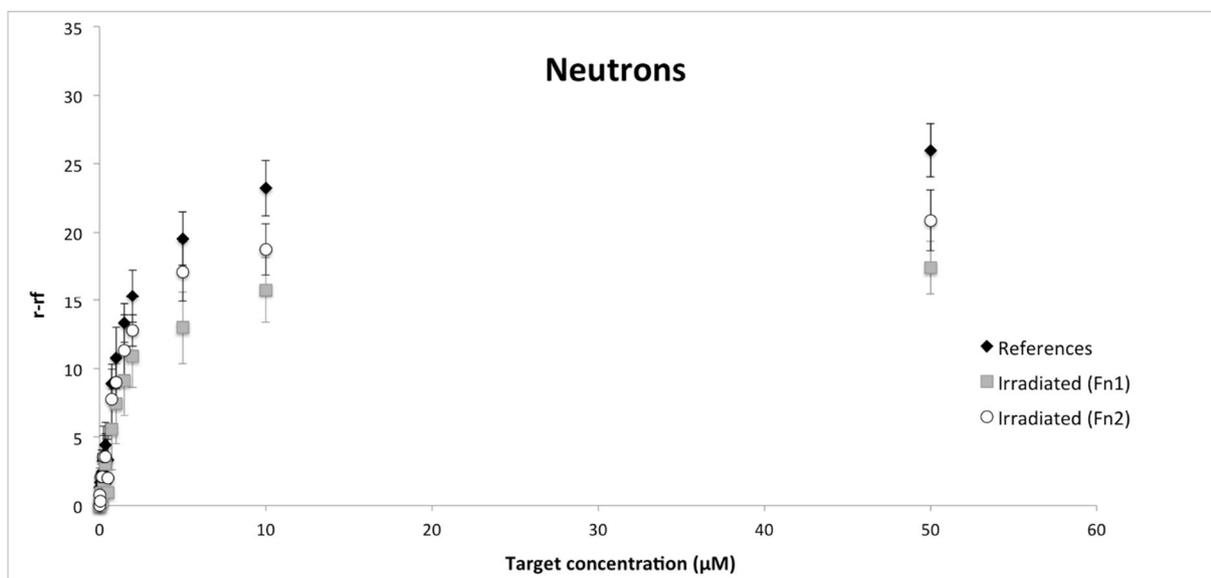
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4c)

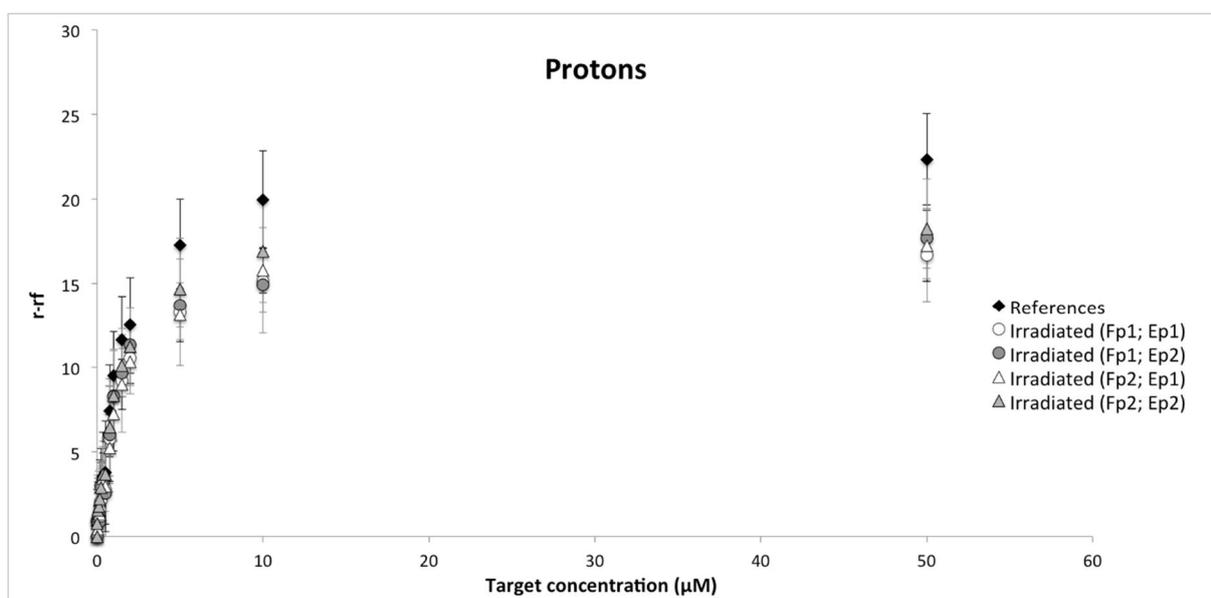


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4d)



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687 **Figure 4 caption:** Affinity curves for 49merYm30F aptamer with additions of *L*-Tym  
 688 obtained for reference (non irradiated sample), and irradiated samples with particles at various  
 689 fluence (F), energy (E), and irradiation duration (D), (for details see Table 1). The saturation  
 690 curves were used for the determination of dissociation constant values ( $K_d$ ) (Eq. 2) by non-  
 691 linear regression analysis. The calculated relative  $K_d$  are listed in Table 2. Reference samples  
 692 ( $n=3$  or  $6$ ) are labelled in black diamond.

693 4a) Irradiated samples with  $^{12}\text{C}$  ions ( $F_c; E_c; D_c$ ) are illustrated in grey circle ( $n=3$ ). 4b)  
694 Irradiated samples are represented in dark grey triangle with electron fluences of  $3.1011$   
695  $\text{p}/\text{cm}^2$  ( $F_{e1}; n=4$ ) and with a grey circle for  $3.1010$   $\text{p}/\text{cm}^2$  fluence ( $F_{e2}; n=4$ ).

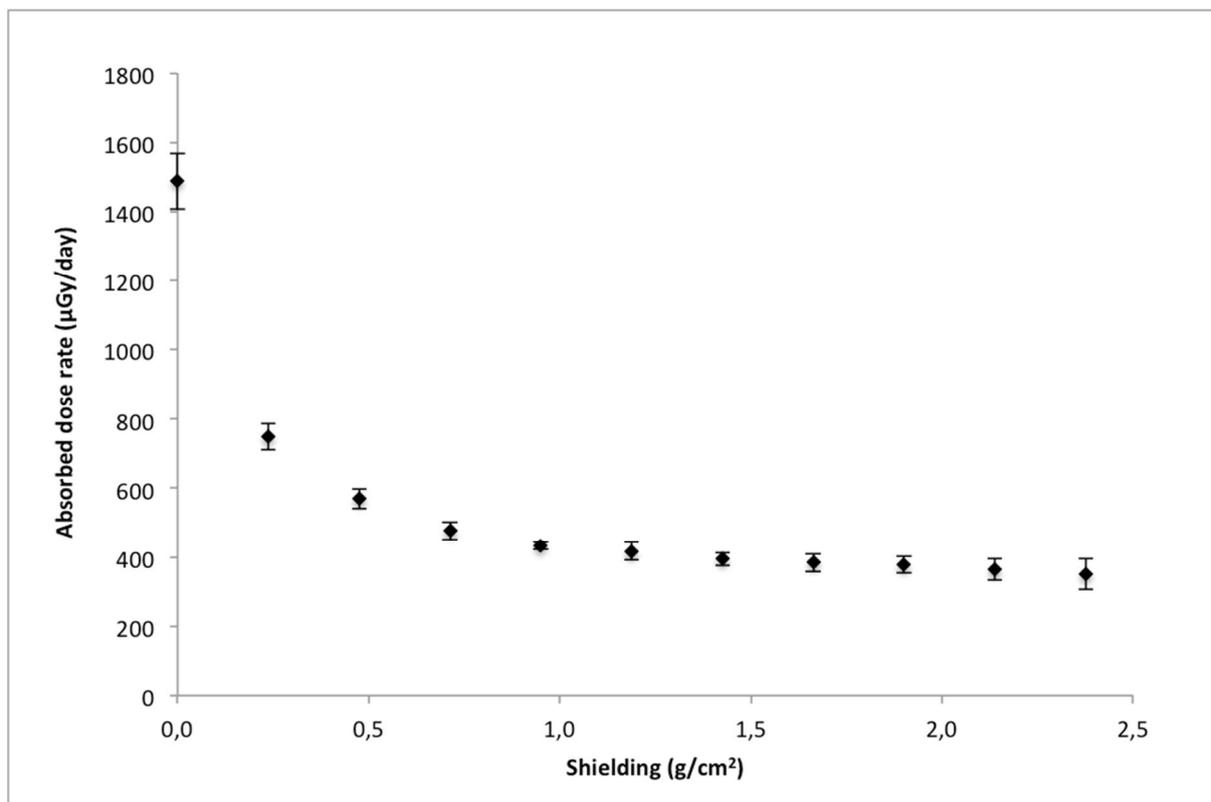
696 4c) Irradiated samples with neutron fluences of  $3.1013$   $\text{p}/\text{cm}^2$  ( $F_{n1}; n=3$ ) are illustrated in  
697 grey square, and  $3.1012$   $\text{p}/\text{cm}^2$  ( $F_{n2}; n=3$ ) in empty circle.

698 4d) Irradiated sample with protons fluence of  $3.1012$   $\text{p}/\text{cm}^2$  with energy of  $25$  MeV ( $F_{p1};$   
699  $E_{p1}; n=3$ ) are represented in empty circle, and in grey circles for an energy of  $50$  MeV ( $F_{p1};$   
700  $E_{p2}; n=3$ ). Fluence of  $3.1011$   $\text{p}/\text{cm}^2$  with energy of  $25$  MeV ( $F_{p2}; E_{p1}; n=3$ ) are plotted with  
701 an empty triangle, and with an energy of  $50$  MeV ( $F_{p2}; E_{p2}; n=3$ ) with a grey triangle.

702

703

704 **Figure 5**



705

706

707 **Figure 5 caption:** Evolution of the absorbed dose rate as a function of the shielding.

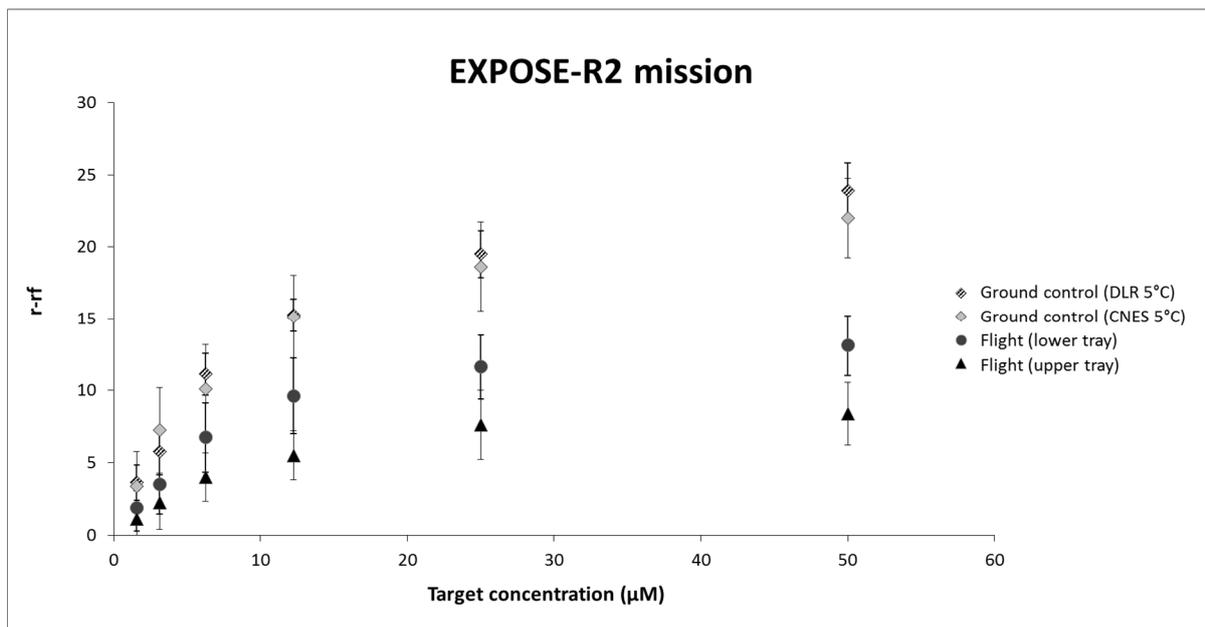
708 Measurements were performed on 5 stacks of 11 TLDs placed into the Tray 3 during

709 EXPOSE-R2 mission.

710

711

712 **Figure 6**



713  
714

715 **Figure 6 caption:** Affinity curves for 49merYm30F aptamer with additions of L-Tym  
716 obtained for controls and flight samples. The corresponding relative Kd values are given in  
717 Table 3.

718

719 **Table 1**  
720

Date	Place		Particule type	Fluence (particules/cm <sup>2</sup> )	Energy	Irradiation duration
04/27/2012	Istituto Nazionale di Fisica Nucleare	Catania, Italy	<sup>12</sup> C	$F_c = 2 \cdot 10^6$	$E_c = 62 \text{ MeV/nuc}$	$D_c = 7 \text{ min}$
07/07/2011	Institut Bergonié	Bordeaux, France	Electrons	$F_{e1} = 3 \cdot 10^{11}$	$E_e = 9 \text{ MeV}$	$D_{e1} = 70 \text{ min}$
				$F_{e2} = 3 \cdot 10^{10}$		$D_{e2} = 7 \text{ min}$
06/01/2010	Centre de Recherche du Cyclotron	Louvain-la-Neuve, Belgium	Neutrons	$F_{n1} = 3 \cdot 10^{13}$	$E_n = 17 \text{ MeV (mean)}$	$D_n = 22 \text{ min}$
03/06/2010			Protons	$F_{p1} = 3 \cdot 10^{12}$		$E_{p1} = 25 \text{ MeV}$
				$F_{p2} = 3 \cdot 10^{11}$	$E_{p2} = 50 \text{ MeV}$	$D_{p2} = 10 \text{ min}$

732

733 **Table 1 caption** Main experimental conditions for neutron, proton, electron and carbon ions irradiations. Subscript letters indicate the particle type  
734 and subscript numbers are used to differentiate the applied conditions. Units for fluences are in particles per cm<sup>2</sup>. Technical details have already  
735 been reported in Baqué *et al.* studies (Baqué *et al.*, 2017).

736

737

738

739 **Table 2**

Experiments on ground-based particles accelerator facilities	
Particle type	relative $K_d$ for irradiated samples (fluence, energy, duration, replicates)
$^{12}\text{C}$	$1.01 \pm 0.15$ ( $F_c; E_c; D_c; 3$ ) 742
Electrons	$1.09 \pm 0.22$ ( $F_{e1}; E_e; D_{e1}; 4$ )
	$0.99 \pm 0.20$ ( $F_{e2}; E_e; D_{e2}; 4$ ) 743
Neutrons	$1.19 \pm 0.35$ ( $F_{n1}; E_n; D_n; 3$ )
	$0.99 \pm 0.16$ ( $F_{n2}; E_n; D_n; 3$ ) 744
Protons	$0.88 \pm 0.21$ ( $F_{p1}; E_p; D_{p1}; 3$ )
	$0.80 \pm 0.15$ ( $F_{p1}; E_{p2}; D_{p1}; 3$ ) 745
	$0.93 \pm 0.18$ ( $F_{p2}; E_{p1}; D_{p2}; 3$ )
	$0.84 \pm 0.16$ ( $F_{p2}; E_{p2}; D_{p2}; 3$ ) 746

747

748 **Table 2 caption:** Relative dissociation constant values (relative  $K_d$ ) (Eq. 2) derived from the  
749 titration curves obtained for irradiated samples (n=3 or 4) after ground-based irradiation  
750 experiments with at least (n=3) reference samples.

751

752 **Table 3**

Biochip in PSS experiment during the EXPOSE-R2 mission		
Sample type	relative $K_d$ (replicates)	
Ground control (CNES 5°C)	0.82± 0.10 (2)	755
Ground control (DLR 5°C)	1.00± 0.10 (2)	756
Ground control (DLR ΔT)	1.29± 0.28 (1)	757
Ground control (DLR ΔT + UV)	0.93± 0.20 (2)	758
Flight (upper tray)	0.94± 0.14 (2)	759
Flight (lower tray)	1.07± 0.21 (2)	760

761

762

763 **Table 3 caption:** Relative dissociation constant values (relative  $K_d$ ) (Eq. 2) derived from the  
764 titration curves issued from the "Biochip in PSS experiment" during the EXPOSE-R2  
765 mission.