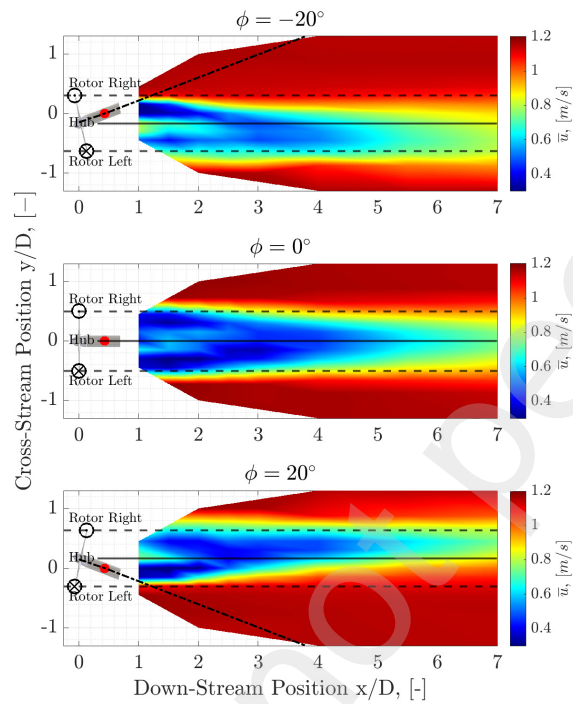


Graphical Abstract

The effect of combined yawed and turbulence intensity on the wake development and performance of a tidal stream turbine.

Matthew Allmark, Allan Mason-Jones, Jean-Valery Facq, Benoît Gaurier, Gregory Germain, Tim O'Doherty



Highlights

The effect of combined yawed and turbulence intensity on the wake development and performance of a tidal stream turbine.

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- Detailed description of wake skew under yawed inflow.
- Detailed description of turbulent structures in the near-mid wake.
- Initial study into wake steering for HATTs.

The effect of combined yawed and turbulence intensity on the wake development and performance of a tidal stream turbine.

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Abstract

The development of wakes downstream of horizontal-axis tidal stream turbines is of interest because these devices, when installed in arrays, can generate predictable renewable energy. Specifically, wake development impacts both the performance of individual turbines and the overall turbine arrays, where upstream turbines can reduce the power output of downstream devices. Additionally, tidal flows are often not perfectly symmetric and can exhibit short-term variations in the predominant incoming flow angle.

This work presents the methods and findings of a lab-scale experimental campaign designed to characterize the wake structure under combined yaw and turbulent flow conditions. A 0.9 m lab-scale tidal turbine was subjected to low and high turbulent inflow characteristics, two yaw conditions ($\pm 20^\circ$), and a no-yaw case. The wake downstream of the device was recorded using a 3 component laser Doppler velocimeter, and these measurements were used to characterize the wake structure.

Under low turbulent conditions, yawing marginally improved wake recovery. In most yaw cases, wake skew was observed; however, the center-line progression of the wake was complex and influenced by the in-flow characteristics. Some degree of self-similarity in the flow was observed, which would presumably improve with downstream distance. Analysis of turbulence in the wake revealed a complex picture — with particularly high levels of turbulent kinetic energy at the blade tips under yawed flow conditions. The length scale of turbulence in the near to mid-wake was smaller than the rotor plane

length scales, and the return to isotropic conditions, which is assumed in many computational models, is complex and dependent on the anisotropy of the inflow turbulence.

Keywords: Tidal Energy, Horizontal Axis Tidal Turbine, Wake Measurement, Flume testing

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2000 MSC: 0000, 1111

1. Introduction

The wakes of Horizontal Axis Tidal Turbines (HATTs) have been an active area of research for over a decade. Tidal flows are generally bi-directional with approximately 180° between ebb and flood flows, however some studies have shown misalignment of ebb and flood directions can be greater than $\pm 20^\circ$ but tend to fall within this range [1]. This can be due to local bathymetry, or the presence of headlands, tidal channels or larger features. Devices such as the SeaGen device pitched their turbine blades through 180° , but do not change the orientation of the turbine axis, and therefore the impact of flow misalignment on the turbine and the wake development downstream of devices needs to be assessed. Misalignment was identified early on by Bahaj et al. [2] as an issue which needed to be addressed by tidal turbine manufacturers. Furthermore, as tidal flows are predictable and periodic means there is ample scope for optimising the layout of tidal turbine arrays. This extends not only to a minimum distance between turbines in the main flow direction, but also allows more complex interactions to be considered, such as cross-stream spacing to take advantage of blockage effects, or alternating rotation directions to take advantages of swirl in the wake, or in the case of the research presented through utilising yaw misalignment to take advantage of wake meander.

A growing body of research has provided flow velocity measurements in the wakes of actuator discs [3], behind individual turbines [4], [5], [6] and in the wake of multiple devices [7] [8]. A central aspect of the experimental work produced so far have focused fully or in part on the effect of the turbulence levels in the on-coming flow on the device performance and wake development. The work of Maganga et al.[4] showed that an increase in turbulence intensity causes faster wake recovery, which was confirmed by [5], who concluded that, whilst an increase in ambient turbulence intensity only

29 had a limited effect on the mean C_P and C_T , the wake was “deeply influ-
30 enced” by the upstream turbulence, and that the “wake shape, length and
31 strength largely depend on the upstream turbulence conditions”. The work
32 of Ebdon et al. [9] further developed this notion and considered the effect of
33 operating tip-speed-ratio (λ) on the wake structure. This work also consid-
34 ered the effect of the method of characterising the wake width and length on
35 the reported wake structure - i.e. showing that the wake width reported was
36 highly dependent on the method of describing the wake boundary. Ebdon’s
37 work further corroborated the finding that ambient turbulence upstream of
38 the device has a large effect on the wake recovery and further concluded that
39 the manner in which one defines the wake width has a profound impact on
40 the reported wake shape, with the later finding having a significant impact
41 on the potential for the development of wake models for array optimisation
42 [9]. A study with single turbines has also considered the effects of wide,
43 shallow channels, to try and match possible geometric conditions in ocean
44 channels, which found only small levels of asymmetry in the wake expansion
45 [6]. A similar, high fidelity set of measurements using an Acoustic Doppler
46 Velocimeter (ADV) in the wake of a turbine was published recently by Chen
47 et al.[10], who measured to 20 diameters downstream of the rotor. This low
48 ambient turbulence study with a relatively high blockage ratio of 16% by
49 turbine swept area found that the centre-line velocity recovered to 90% of
50 the free stream velocity approximately 11 diameters downstream of the rotor.
51 As seen in a previous Large Eddy Simulation (LES) study [11], it was found
52 that the stanchion had a significant influence over the wake in its immediate
53 vicinity, but this was limited to the near wake region.

54 Another aspect of wake development which has received a good deal of
55 attention are the development of turbulence characteristics within the wake.
56 Turbulence characteristics such as turbulence intensity, decay rate and level
57 of anisotropy are thought to play a central role in the wake recovery process
58 as well as in the application of HATTs. Investigations into the turbulence
59 produced in the near wake (between 1.5 and 7 diameters downstream of
60 rotor) were made by Tedds et al. [12] who noted that the rotation of the tur-
61 bine blades induced significant anisotropy into the turbulence, and suggested
62 therefore that numerical models which rely on the assumption of isotropic
63 turbulence (e.g. 2-equation RANS models) may struggle to accurately re-
64 produce the flow in this region. It was also noted that the rate of decay
65 of turbulence kinetic energy in the wake region was “significantly different
66 to that observed downstream of grids, meshes or perforated disks, suggest-

67 ing that previous modelling approaches, which neglected swirl effects and
68 modelled the turbine by absorption discs, may significantly over predict the
69 turbulent kinetic energy (k) decay rate of HATT wakes.” [12]. This would
70 suggest that the quantitative results of porous discs may not be directly ap-
71 plicable to tidal turbines, however, qualitative effects such as those noted by
72 Blackmore et al. [13] could well still be applicable. Zhang et al. [14] also
73 studied the homogeneity of the turbulence in the wake of a lab-scale tidal
74 turbine.

75 A final aspect of wake development which has warranted attention are the
76 characteristics of the near wake region, particularly with respect to measure-
77 ments made in a phase locked manner to the turbine rotation. To this end,
78 Morandi et al.[6] conducted a flume study where measurements were made
79 immediately behind the turbine rotor. These measurements were phase-
80 locked with the turbine’s rotational frequency in order to identify how wake
81 features relate to the position of blades. The study demonstrated the com-
82 plexity of the near-wake structure, and also indicated that some features of
83 the near-wake such as the strength of the tangential velocity component, are
84 dependent on the turbine operating condition (tip-speed ratio). This was
85 also found in the work presented by Ebdon et al [9].

86 One approach which has been pursued since the wide scale introduction of
87 wind turbines (and subsequently as the field of tidal turbines has developed)
88 is to attempt to find an analytical or empirical method for the prediction of
89 the wake. The attraction of this approach is that a prediction of the effects
90 of the wake without the need to resolve the flow-field around the turbine
91 would potentially provide a quick method to predict the effects of a turbine
92 on its surrounding environment. Early work in this area was conducted by
93 Lissamann [15], who attempted to develop a “functional and dimensionally
94 correct” analytical model of turbine wakes, based on known profiles of jets
95 and plumes. Whilst no attempt was made to take into account the com-
96 plex physical interactions in the wake (presumably due to the limitations of
97 computational power at the time), Lissamann does include two turbulence
98 terms – one for the ambient turbulence and one for the turbulence generated
99 by the rotor itself, which are considered important to the development of
100 the wake. These terms are tuned to the limiting cases of a plume and jet
101 flow, and the model applied to wind farms to investigate farm power output
102 for different configurations and wind directions, with agreement to real wind
103 farms found to be reasonable considering uncertainties due to the instability
104 of meteorological conditions and the limited physics contained within the

105 model. Jensen [16] developed an analytical model based on mass conserva-
106 tion considerations to produce a ‘top-hat’ lateral profile of velocity deficit
107 downstream of a wind turbine. The theoretical basis of this was adapted
108 by Frandsen et al.[17] to include both momentum and mass conservation,
109 and the resulting model applied to arrays of offshore wind turbines. The
110 ‘top-hat’ profile, which underestimates velocity deficit at the centre of the
111 wake, and overestimates it at the wake edges, was replaced by Bastankhah
112 and Port´eAgel [18] with a Gaussian profile. This was shown to produce a
113 better match to downstream velocity profiles behind a wind turbine, when
114 compared to LES and experimental data.

115 Building on this work with wind turbine wakes, Lam and Chen [19] com-
116 bined axial momentum theory across an actuator disc with flume measure-
117 ments from Maganga et al. [4] to develop analytical equations for the pre-
118 diction of the axial velocity deficit and its lateral distribution downstream of
119 a tidal turbine. The equations for the wake prediction are based on work for
120 ship propellers. These equations were then improved upon to take into ac-
121 count the effects of the turbine hub in order to predict the region of “double-
122 dip” wake deficit seen in the near-wake region[70] in flows with low ambient
123 turbulence intensity. The improved equations were compared to wake mea-
124 surements made by Pinon et al. [20] in high ambient turbulence conditions,
125 and showed good agreement for the far wake. This model, however, relies on
126 using empirical data from the centreline velocity deficit of each turbine in or-
127 der to calibrate it. The disadvantage of these empirical and analytical models
128 lies in the fact that they are attempting to reproduce a wake using a rela-
129 tively small number of input variables. Real wakes are dependent on many
130 different factors and the complex physical interaction between these factors,
131 and therefore, whilst these empirical and analytical models can produce wake
132 predictions very quickly, their accuracy and ability to provide detailed infor-
133 mation about the wake will necessarily be limited due to the simplicity of
134 the underlying model. The accuracy can only be expected to suffer further
135 as turbines are grouped together into arrays, and further complexities are
136 added to the incoming flow such as vertical velocity shear profiles and ocean
137 bathymetry.

138 This paper seeks to add to the body of knowledge by considering both
139 wake structure and turbulence characteristics under combined inflow and
140 yawed turbine conditions. This work seeks to add experimental data and
141 findings to the research which will aid improved modelling and the develop-
142 ment of low-cost wake optimisation approaches which maybe able to consider

143 yaw. To date such a study has, to the best of the authors knowledge, not
144 been completed.

145 The paper is organised as follows, in Section 2 the experimental design
146 and apparatus is detailed, we then start the results by presenting the time
147 averaged wake velocities (stream-wise and vertical) in Section 3.1. This is
148 followed by presentation of an analysis of how the turbine wake, if consid-
149 ered as an axis-symmetric wake, adheres to self similarity laws in Section
150 3.2. Turbulence characteristics measured in the wake are then considered in
151 Sections 3.3, 3.4 and 3.5, which inspect the turbulence kinetic energy gener-
152 ation and dissipation, the length scale of the turbulence generated and the
153 isotropy and structure of the turbulence generated in the wake, respectively.
154 Finally Section 4 summarises the main findings of the research.

155 2. Methodology

156 2.1. Experimental Setup

157 Figure 1 shows a photograph of the experimental apparatus used to study
158 the impact of inflow turbulence and device yaw on the development of the
159 flow structures in the wake of a lab-scale HATT. Figure 2 shows the setup
160 as a series of plan-view schematics.

161 A lab-scale turbine was mounted on a 0.105 m diameter stanchion con-
162 nected to a supporting structure above the $4 \times 2 \times 18$ m wave and current
163 flume at IFREMER, Boulogne-sur-Mer, as shown in Figure 1. The turbine
164 was mounted at three different yaw angles as designated in Figure 2a, the
165 yaw angles tested were 0° , 20° and -20° . The hub height was set to 1 m
166 below the water surface and centred in the cross-stream direction, at 2 m
167 from each side wall, as seen in Figure 2.

168 The origin of the coordinate system used throughout the paper is shown
169 in Figure 2 and sets the origin at the centre of the hub for the 0° yaw case.
170 Yawing the device anti-clockwise was given a negative designation where the
171 turbine hub moves in the negative y-direction. Finally, the z-direction is
172 aligned with the vertical position in the water column, again with the origin
173 at hub height for the 0° yaw case, downwards is designated as positive and
174 upwards as negative.

175 For all cases the turbine stanchion was kept in a fixed position leading
176 to movement of the rotor position laterally (y-direction), the position of
177 the turbine hub and blade extremes will be presented in the charts where
178 necessary, as illustrated in Figure 2. Finally, the points on the charts show

179 where the flow velocity in the wake was measured via a 3-Dimension Laser
180 Doppler Velocimeter (LDV).

181 Two broad inflow conditions were generated with similar stream-wise ve-
182 locity magnitude but with significantly different levels of inflow turbulence
183 intensity. This was achieved by repeating the tests twice with and without
184 the flow straightener installed in the flume. The flow straighteners are used
185 to straighten the flow direction, break down larger flow structures and re-
186 duce the overall turbulence level in the flow. The test cases with the flow
187 straighteners are designated the Low Turbulence Intensity (LTI) cases and
188 the tests without the flow straighteners are designated High Turbulence In-
189 tensity (HTI) cases.

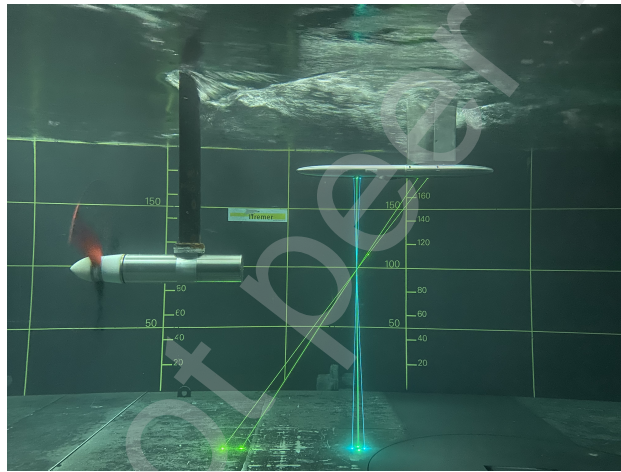


Figure 1: A photograph of the test setup at the IFREMER test facility, the photo shows the lab-scale tidal turbine and the 3-D LDV measurement apparatus used - the picture shows a measurement being taken in the near wake $x/D = 2$.

190 2.2. Lab-Scale Tidal Device

191 The lab-scale turbine utilised in the experiments presented herein was de-
192 veloped by Allmark et al [21] and has been tested thoroughly under a variety
193 of flow conditions, including wave flows [22] [23] [24], profiled flows [25], wake
194 flows [9] and turbulent flows [8]. The device used is one of three produced
195 and is of 0.9 m diameter, utilising blades created with a modified Wortmann
196 FX63-137 airfoil, the exact blade details can be found in [26]. The device
197 is controlled by a Permanent Magnet Synchronous Machine (PMSM) which
198 can be operated in speed or torque (load) control - for these experiments

199 speed control was used to maintain a stable rotational velocity within ± 1
200 RPM of the target. The Permanent Magnet Synchronous Machine (PMSM)
201 was mounted in a direct-drive configuration where the rotational velocity was
202 measured via an encoder which was utilised for vector oriented control [24].
203 A torque/thrust transducer developed by Applied Measurements is mounted
204 within the drive-train directly behind the turbine rotor, upstream of any
205 seals or bearings. Blade root bending moments are measured on each blade
206 using a series of full-bridge strain gauges, installed by Applied Measurements
207 - the signals for the bridge are amplified in the turbine nose cone. The data
208 was collected by various data acquisition cards mounted within a National
209 Instruments Compact Rio and sampled at 200 Hz.

210 *2.3. Wake Measurement*

211 A 3-Dimension Laser Doppler Velocimeter (LDV) system was used to
212 measure the fluid velocity at the turbine rotor plane and throughout the
213 near-to-mid wake. The 3D LDV system used 6 laser beams to measure three
214 components of the fluid velocity, u , v and w , aligned with the x , y and z
215 axes, respectively - see Figure 2. The lasers used to measure the flow had
216 wavelengths of 514 nm, 466 nm and 532 nm with the raw data requiring
217 projection onto the x , y and z axis. The 3D LDV requires that the flume
218 tank is seeded with reflective particles which in this case are silver coated glass
219 particles of 10 μm diameter - the particles are small enough to have minimal
220 impact on the flow but large enough to reflect enough light to achieve a good
221 signal-to-noise ratio. The 3D LDV has a variable sample rate, as a seeding
222 particle must pass through two converging lasers to make a measurement.
223 The sample rate is therefore related to the number of seeding particles used
224 and for these experiments the level of seeding resulted in an average sample
225 rate of 182 Hz. However, to process the data consistently the readings were
226 re-sampled to a common sample rate of 170 Hz.

227 The flow velocity at the rotor plane (without the turbine installed) was
228 measured prior to the experiments with 3D LDV mounted inline with the
229 eventual rotor location, measuring various points in a cross formation span-
230 ning the device rotor. The locations of the measurement points are also
231 shown in Figure 2b and 2b.

232 For each yaw angle and turbulence intensity combination the near to mid
233 wake of the turbine was measured via a 3D LDV. The near-to-mid wake was
234 measured, with the turbine set to a fixed rotational velocity, and the points
235 measured are highlighted in Figures 2b and 2c. For each wake map the

236 turbine was operated at approximately 75 RPM or 7.85 rad/s which resulted
237 in an average tip-speed ratio (λ) of $\lambda = 4$ which has been previously found to
238 give peak power output for the lab-scale turbine [21]. During the LTI cases
239 the 3D LDV data was captured for 150 s whereas for the HTI cases each
240 wake point was recorded for 300 s.

241 2.4. Rotor Plane Conditions

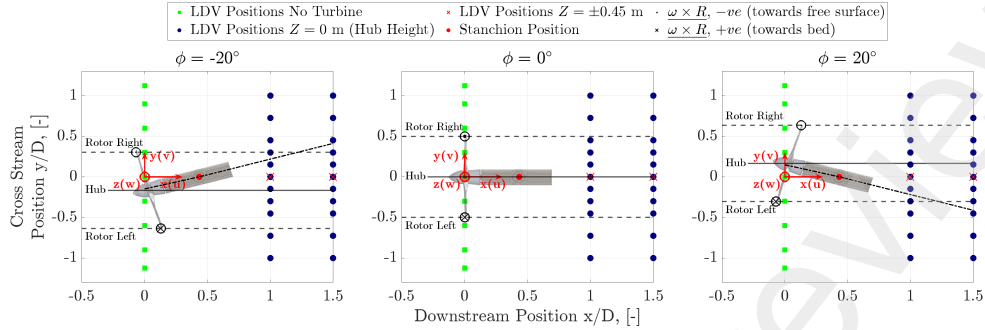
242 The recirculating flume was setup to have an upstream velocity of, $u_\infty \approx$
243 1.1 ms^{-1} and the tests were undertaken with and without honeycomb flow
244 straighteners installed to give two test cases, LTI (with straighteners, $TI \approx$
245 2%) and HTI (without flow straighteners, $TI \approx 15\%$).

246 Here, \bar{u} is the time averaged stream-wise velocity derived from the 3D
247 LDV measurements by Equation 1, where Δt is the time period between
248 subsequent samples and N is the number of sample points, finally $u(t)$ is
249 the stream-wise fluid velocity recorded via the 3D LDV at a given location
250 - similar expressions were used to average the v and w velocity components
251 to give \bar{v} and \bar{w} . The 3 dimensional Turbulence Intensity (TI) is given by
252 Equation 2.

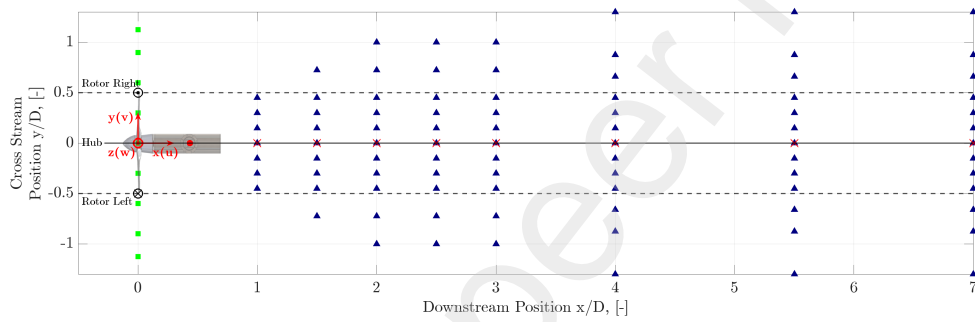
253 An estimation of the integral time scale, \mathcal{T} , was achieved by numeri-
254 cally integrating the auto-correlation function over time from 0 to T_0 , where
255 T_0 represents the time lag corresponding to the first zero-crossing of the
256 auto-correlation function [27]. Subsequently, the integral length scale can
257 be determined using the formula $\mathcal{L} = \bar{u}_\infty \cdot \mathcal{T}$. The auto-correlation and the
258 integral used to develop an estimate of \mathcal{T} are defined in Equations 3 and 4,
259 respectively.

260 The resultant inflow conditions measured at the turbine rotor plane prior
261 to installation are displayed for the LTI and HTI cases against cross-stream
262 position and vertical position in Figures 3 and 4, respectively. Table 1 shows
263 the resultant inflow characteristics averaged over the turbine rotor.

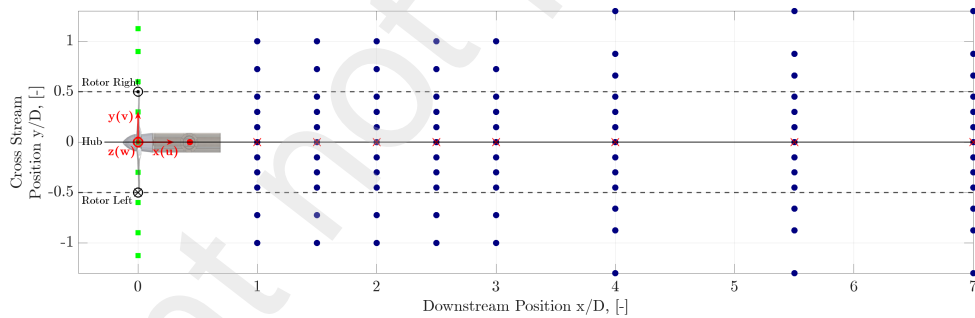
264 Figures 3 and 4 show that consistent results were achieved with the flume
265 settings yielding two test cases with two varying levels of turbulent inflow
266 as desired. However, in the process of removing the flow straighteners an
267 overall rotation is imposed on the flow - this can be seen in the flow profile
268 for \bar{w} in Figure 3b and the flow profile observed for \bar{v} in Figure 4b. The
269 rotation of the flow is of an opposite orientation to the rotational velocity
270 of the lab-scale turbine. Therefore the rotation imposed on the flow by the
271 HATT rotor will be in the same sense or direction as overall rotation of the
272 flow. As this rotation was unavoidable without excessive additional time and



(a) Turbine orientations



(b) LDV measurement positions, LTI Cases.



(c) LDV measurement positions, HTI Cases.

Figure 2: Schematic of the LDV measurement points for turbulence level and the turbine positions for each of the yawed flow cases, (a) shows the measurement positions for the LDV under the LTI cases, (b) shows the LDV measurement positions under the HTI cases and (c) shows the turbine positions for the three yaw angle setting. Note: the green squares show the rotor plane survey measurement positions, which were taken prior to the turbine installation for both the LTI and HTI cases, \odot illustrate a vector pointing out of page and \otimes illustrates vector pointing into the page.

273 expense, the testing proceeded and the impact of the overall flow rotation is
 274 discussed throughout as required.

$$\bar{u} = \frac{1}{T} \int_{t=0}^{t=T} u(t) \approx \frac{\Delta t}{2} \sum_{i=0}^{i=N} (u(t) + u(t + \Delta t)) \quad (1)$$

$$TI = \frac{\sqrt{\frac{1}{3}(u'^2 + v'^2 + w'^2)}}{\sqrt{u_\infty^2 + v_\infty^2 + w_\infty^2}} \times 100 \quad (2)$$

$$\mathcal{T} = \int_0^{T_0} R(t') \cdot dt' \quad (3)$$

275 where

$$R(t') = \frac{\overline{u'(t)u'(t-t')}}{\sigma_u^2} \quad (4)$$

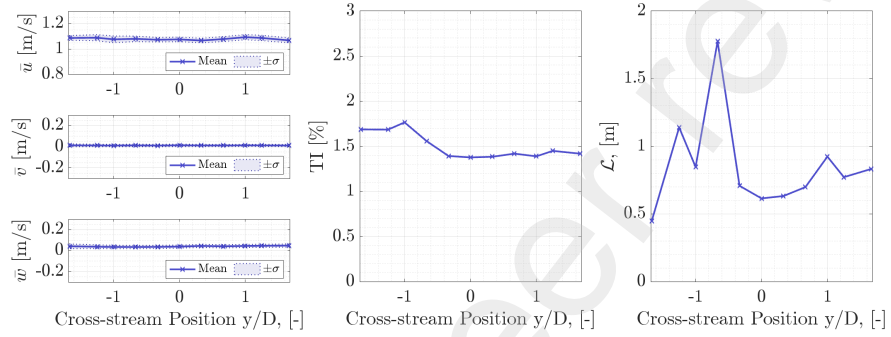
Table 1: Average flow characteristics at the rotor plane before installation of the turbine.

	LTI	HTI
$\overline{u_\infty}$	1.08 m/s	1.05 m/s
TI	1.48 %	13.06 %
\mathcal{L}	0.83 m	0.58 m

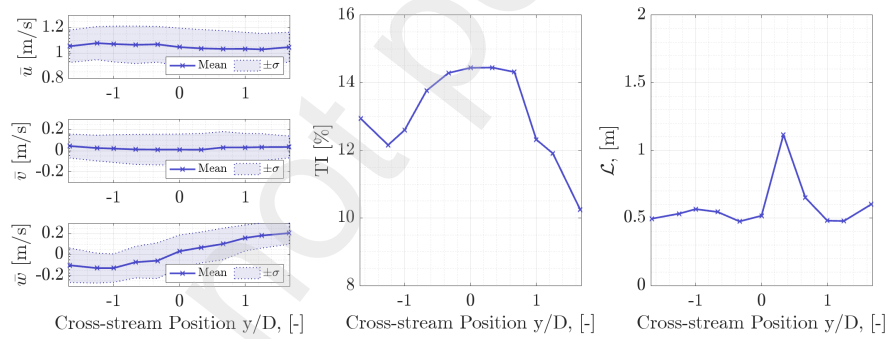
276 2.5. Reynolds, Froude Number Scaling and Blockage Ratio

277 In developing lab-scale experiments care should be taken when scaling dy-
 278 namic, kinematic and geometric aspects, to ensure the results can be utilised
 279 at full-scale. To this end both the Reynolds number (Re) and Froude number
 280 (Fr) were considered when developing the experiments.

281 Re quantifies the ratio of momentum forces to viscous forces within a
 282 given flow and is defined as used here in Equation 5. It is generally infeasible
 283 to achieve Re number equality between lab-scale and large scale testing,
 284 however many non-dimensional parameters have been shown to become Re
 285 independent for Re values between 0.5×10^5 to 1×10^5 , depending on the
 286 rotor geometry [28]. The Re for the experiments developed, based on the
 287 chord length at $r/R = 0.7$ was $Re_{0.7} = 0.9 \times 10^5$, which has been shown pre-
 288 viously to develop a flow regime whereby non-dimensional power and loading
 289 coefficients become independent of Re [28].

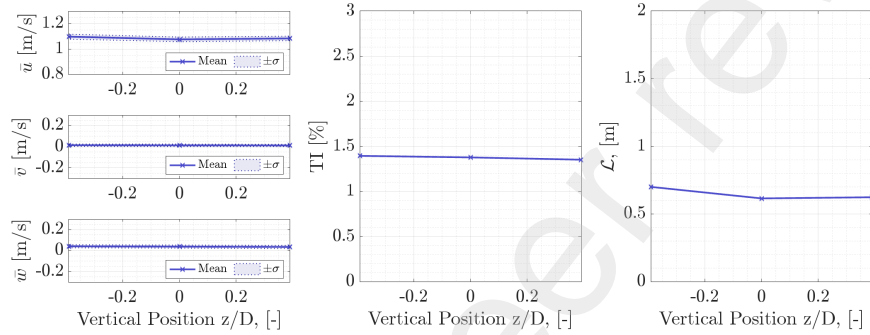


(a) Rotor plane flow characteristics with cross-stream position, LTI Cases

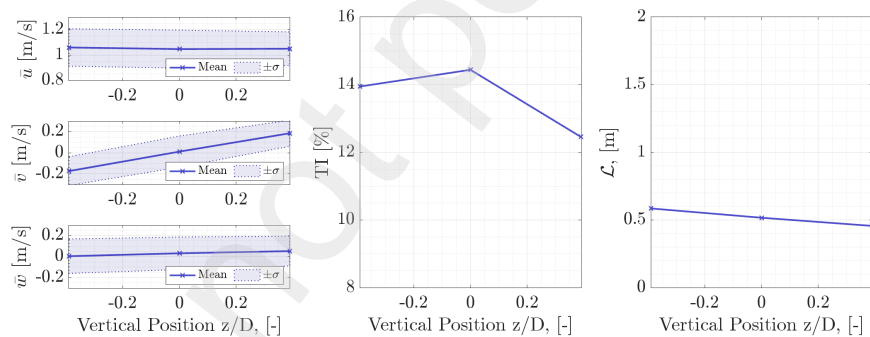


(b) Rotor plane flow characteristics with cross-stream position, HTI Cases

Figure 3: The inflow characteristics (U , TI and \mathcal{L}) recorded at the Rotor plane of the lab-scale tidal turbine plotted against cross-stream position.



(a) Rotor plane flow characteristics with vertical position, LTI Cases



(b) Rotor plane flow characteristics with vertical position, HTI Cases

Figure 4: The inflow characteristics (U , TI and \mathcal{L}) recorded at the Rotor plane of the lab-scale tidal turbine, plotted against vertical position.

$$Re = \frac{\rho \overline{u_\infty} C_{0.7}}{\mu} \quad (5)$$

290 Fr quantifies the relative impact of inertial forces relative to gravitational
 291 forces and is calculated here using the hub depth and free-stream velocity
 292 using Equation 6. Fr for the testing setup described was $Fr_{Lab} = 0.25$. This
 293 is compared to Fr numbers ranging between 0.15 and 0.21 for large scale
 294 tidal devices operating in flows of 3 ms^{-1} at hub depths between 20 and 40
 295 m. This is inline with the values reported in literature ($Fr = 0.143$ in [29])
 296 and together with the Re number achieved, the test are thought to balance
 297 of the dynamics of inertial, viscous and gravitational forces appropriately.

$$Fr = \frac{v}{\sqrt{g \cdot D}} \quad (6)$$

298 The blockage ratio, B is utilised to acknowledge that in flume testing
 299 the flow is constrained by the flume walls when compared to the open ocean.
 300 Numerous studies have shown that more fluid is forced through the rotor, as it
 301 is unable to freely pass around the rotor due to the flume walls, and therefore
 302 elevated power and loading values are recorded [30][31][32]. Similarly, the
 303 effect of the flow being constrained also impacts the development of flow in
 304 the wake of a given HATT - however for wind turbines it has been shown
 305 that blockage ratios of < 0.09 have minimal impact wake mixing whereas
 306 blockage ratios if 0.2 had a significant impact on mean stream-wise velocity
 307 in the wake [33]. The blockage ratio used here is given by Equation 7, where
 308 A_t is the swept area of the turbine rotor and A_F is the cross-sectional area
 309 of the flume. For the testing detailed herein a blockage ratio of $B = 0.08$ or
 310 8% was achieved.

$$B = \frac{A_t}{A_F} \quad (7)$$

311 2.6. Thrust Coefficient and Blockage Correction

312 As the blockage level of, 8%, was fixed and unavoidable, blockage cor-
 313 rection of the rotor non-dimensional thrust coefficient, C_T , was considered.
 314 Here, the C_T is defined in Equation 8, where u_∞ is the rotor plane velocity
 315 prior to turbine installation. The blockage correction developed by Bahaj
 316 et al was used [2] to correct the thrust coefficient to represent the turbine
 317 performance in an unbounded flow. The C_T values, both measured and cor-
 318 rected are given in Table 2. Here, the authors note that the levels will have

319 an effect on the wake progression, where observable these effects are noted
 320 throughout.

$$C_T = \frac{T}{\frac{1}{2}\rho A_t u_\infty^2} \quad (8)$$

Table 2: Thrust Coefficient, C_T , before and after blockage correction for both LTI and HTI cases.

	Measured C_T	Blockage Corrected C_T
LTI, $\Phi = 0^\circ$	0.988	0.900
LTI, $\Phi = 20^\circ$	0.924	0.854
LTI, $\Phi = -20^\circ$	0.926	0.860
HTI, $\Phi = 0^\circ$	0.993	0.903
HTI, $\Phi = 20^\circ$	0.917	0.849
HTI, $\Phi = -20^\circ$	0.935	0.862

321 3. Wake Characteristics

322 3.1. Time-Average Flow Velocities

323 This section presents a summary of the wake structures measured in terms
 324 of the average velocity and velocity deficit in the principle orientations of in-
 325 terest, x (u) and z (w). The results of applying Equation 1 to the measured
 326 flow data are presented in Figures 5 to 8 for a horizontal plane, positioned
 327 vertically at the rotor hub centre. Figure 5 shows the velocity deficit de-
 328 rived from the average stream-wise fluid velocity measured, the deficit u^* is
 329 given by Equation 9. Figure 7 shows the specific time-averaged vertical flow
 330 velocity \bar{w} , for each wake transect undertaken. In both Figure 5 and 7 the
 331 results of the rotor plane map measurements, as detailed in Figure 2, are
 332 shown in black. Figures 6 and 8 show linearly interpolated wake maps with
 333 the turbine orientation and rotational axis highlighted.

334

$$u^*(x, y) = 1 - \frac{\overline{u(x, y)}}{u_\infty} \quad (9)$$

335 Firstly, we consider the stream-wise velocity wake structure, which is of
336 significance for array spacing, see [8] [34] [9]. An obvious finding shown in
337 both Figure 5 and 6 is that the ambient turbulence has a profound effect on
338 the wake recovery rate. During the LTI cases the wake deficit was found to
339 be significant at $x/D = 7$, whereas this deficit was heavily diminished at the
340 same point for the HTI cases, this can be clearly seen for each yaw angle
341 case. This finding is inline with previous work undertaken in the area [9].
342 This finding is relatively intuitive as the increased ambient turbulence levels
343 in the HTI cases led to greater mixing of the slowed wake flow and accel-
344 erated bypass flow at the wake extremes which leads quicker wake recovery.
345 Yawing the turbine was not found to have a significant impact on the wake
346 recovery rate in the stream-wise direction for either turbulence intensity case.

347
348 In the LTI cases the asymmetry in blockage introduced under the yawed
349 cases led to a very slight asymmetry in the bypass flow where a marginally el-
350 evated value of the bypass flow was observed on the side with the smaller tip
351 clearance to the flume side created in the yawing process. For example, in the
352 $\phi = 20^\circ$, LTI case, on the left hand side the deficit was $u^* = -0.06$, whereas
353 on the more constrained right hand side the deficit was $u^* = -0.09$, where
354 negative values show an increase in flow velocity. In the HTI cases, there was
355 an overall asymmetry to the bypass flow regime, which persists regardless of
356 the yaw angle and associated asymmetrical flow constraint - for example in
357 the $\phi = 0^\circ$ case, on the left of the turbine $u^* = 0.09$ was observed and on
358 the right $u^* = 0.04$. The left hand bias in this case maybe related to asym-
359 metry in the upstream flow introduced by the removal of flow straighteners
360 which has already been observed in the Section 2.3. The accelerated bypass
361 flow diminished more quickly in the HTI case compared to the LTI due to
362 the increased mixing in the shear layer enabled by the turbulence in the flow.

363
364 In the near wake, $1 < x/D < 2$, the particular contributions of the hub
365 and blade tips to the wake can be seen, particularly at $\phi = 0^\circ$. These struc-
366 tures have been shown to be heavily dependent on the λ -value setting for the
367 experiments. The structures from these individual effects were observed for
368 larger x/D in the LTI cases ($x/D = 2.5$) and were less persistent in the HTI
369 cases, where they were less prominent at the $x/D = 1.0$ and were destroyed
370 by $x/D = 1.5$. In the yawed cases the effects of the turbine hub/nacelle
371 and the blade tip structures, created when shadowing the hub/nacelle, were
372 merged. This led to a combined larger deficit at the tank centre line (between

373 hub and right blade extreme for $\phi = -20^\circ$ and between hub and left blade
374 extreme for $\phi = 20^\circ$). These merged artefacts were dissipated over similar
375 distances as the separate blade and hub deficits in the $\phi = 0^\circ$ yaw case - this
376 was true of both the LTI and HTI cases, where rotor/nacelle specific deficits
377 were destroyed by $x/D = 2.5$ and $x/D = 1.5$, respectively.

378

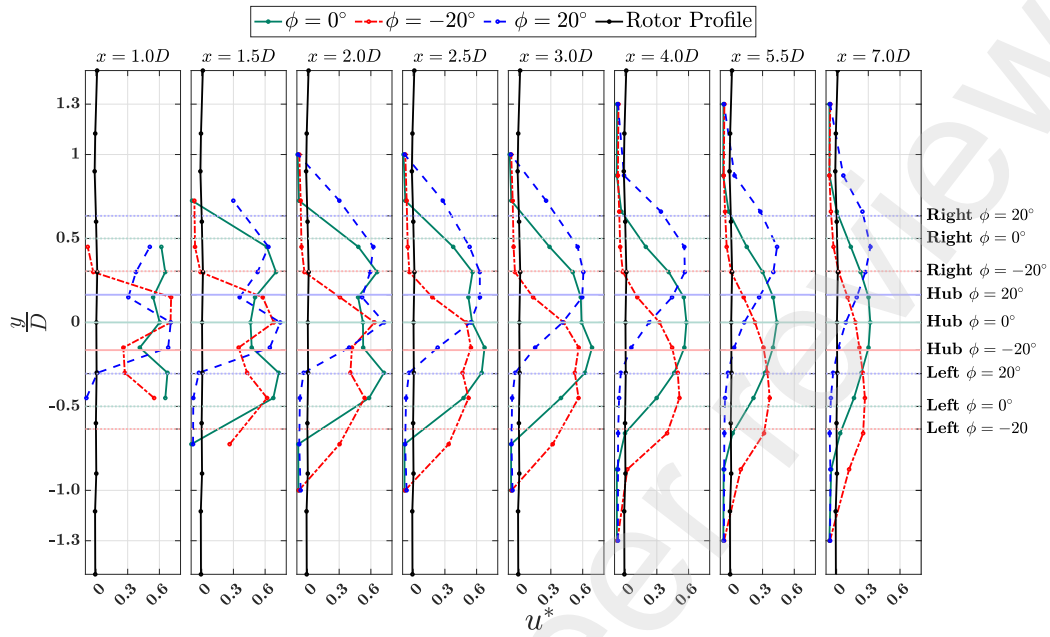
379 The stream-wise velocity map shown in Figure 6 highlights the degree of
380 wake skew observed in the experiments. Here, wake skew is defined as the
381 angle between the turbine rotational axis and the wake axis. The wake maps
382 show that for the LTI cases the orientation of the wake skew was generally
383 found to be inline with momentum theory, i.e. the wake axis is deviated from
384 the stream-wise flow axis in the opposite direction to the turbine yaw direc-
385 tion [35]. Here this deviation can be explained by the reaction force applied
386 by the turbine rotor to the flow which will be equal and opposite to the load
387 developed on the turbine - the load developed on the device acts along the
388 turbine rotational axis in a positive sense and the load applied to the fluid
389 acts along the same axis but in the negative sense, skewing the wake in the
390 opposite direction to the turbine yaw angle.

391

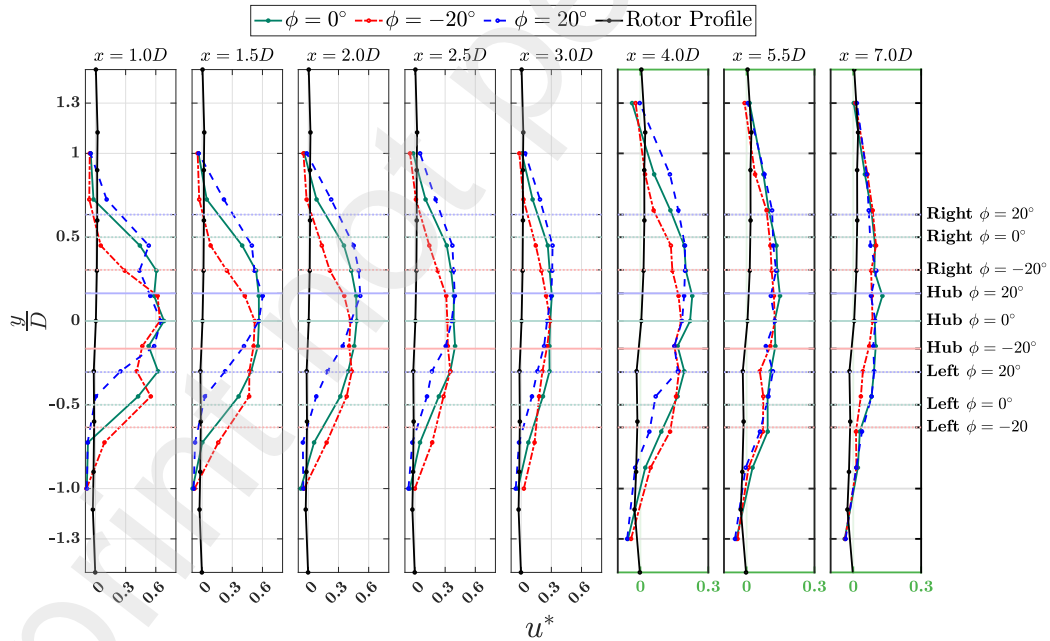
392 The clear agreement with the theoretical approaches noted above was not
393 as strong for the HTI where a more complex wake development was observed
394 for the yawed turbine cases. Initially, the wake is most drastic behind the
395 hub/nacelle and blade tip regions, as noted above, then the position of the
396 maximum deficit meanders to the opposite side of the turbine centre-line (at
397 differing positions for either yaw case) and finally the position of the largest
398 deficit returns to the nacelle-blade combined side creating a curved wake
399 path. The persistence of the merged effect of the hub/nacelle and blade tip
400 extreme was more pronounced in the $\phi = -20^\circ$ case. This finding suggests
401 that ambient turbulence intensity may impact the wake skew under yawed
402 flows; although, as can be seen in Figure 3b, a significant depth wise velocity
403 profile in the HTI case make this finding difficult to generalise.

404

405 Figures 7 and 8 show the depth-wise velocity for a horizontal plane
406 through the centre of the turbine hub. In the LTI case the depth-wise ve-
407 locities recorded show that the wake is rotating in an opposite direction to
408 the turbine rotation as expected, with the centre of rotation shifted to the
409 position of the turbine hub for the $\phi = -20^\circ$ and $\phi = 0^\circ$ cases. However,
410 for the $\phi = 20^\circ$ case the centre of rotation is shifted towards the right hand

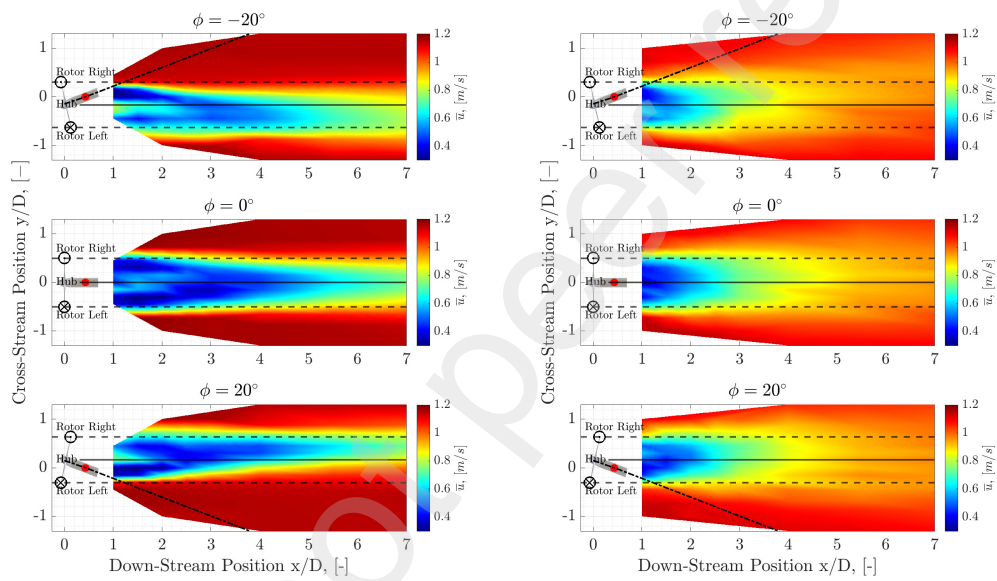


(a) Stream-wise Velocity Deficit, LTI Cases



(b) Stream-wise Velocity Deficit, HTI Cases

Figure 5: Time averaged stream-wise flow velocity with downstream and cross-stream position for the (a) LTI cases and (b) HTI cases. NOTE: Coloured x-axis show differing x-limits to better display data.



(a) Stream-wise Velocity Wake Map, LTI Cases

(b) Stream-wise Velocity Wake Map, HTI Cases

Figure 6: Time averaged stream-wise flow velocity wake map with downstream and cross-stream position for the (a) LTI cases and (b) HTI cases. Dotted line projecting from the turbine sketch shows the turbine axis.

411 side of the turbine and shifts increasingly in this direction with downstream
412 distance. In the yawed cases the extent of the induced depth-wise velocity is
413 reduced slightly for the two yawed cases ($\phi = -20^\circ$ and $\phi = 20^\circ$) in compar-
414 ison to the non-yawed case ($\phi = 0^\circ$).

415

416 As shown in Figure 7b, developing a clear picture of the depth-wise ve-
417 locity generated by the device is difficult due to the inherent rotation in the
418 ambient flow as discussed in Section 2.3. However, in the near wake, at
419 $x/D = 1$, an increased magnitude of depth-wise velocity can be seen. As in
420 the LTI case the opposing direction either side of a centre of rotation, where
421 $w = 0$ m/s, show that the wake rotates in the opposite direction to the tur-
422 bine rotation, to a greater degree than the ambient flow. Very quickly, by
423 $1.5 \leq x/D \leq 2$, the depth-wise velocities induced by the rotor are dissipated
424 towards the inherent depth-wise profile observed in the ambient flow. Figure
425 8 shows that the effect of wake skew can be seen in the depth-wise velocity
426 maps, particularly for the LTI cases.

427

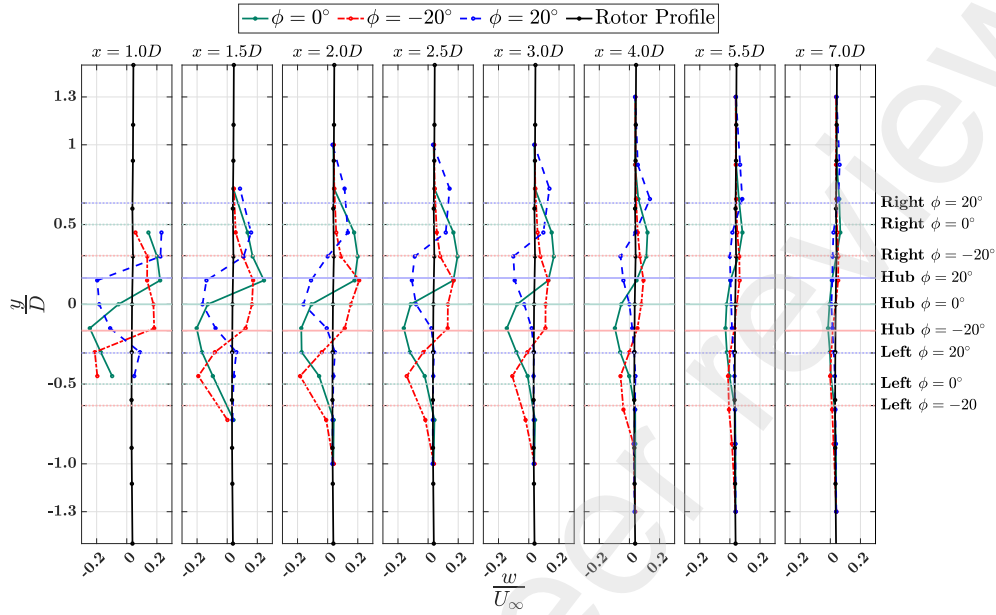
428 3.2. Self Similarity and Wake Centre

429 In this section we consider the structure of the wake from the point of
430 view of its progression and expansion under effective normalisation. Such a
431 manner of viewing wake development is known as a self similarity which, if
432 present, means that the structure of the wake is independent of downstream
433 position when appropriately normalised. Self similarity is well understood
434 for the development of plane wakes behind foundational geometries such as
435 spheres [36], airfoils [37], and disks [38] and we use these formulations with
436 some minor modifications to account for the yawed inflow conditions.

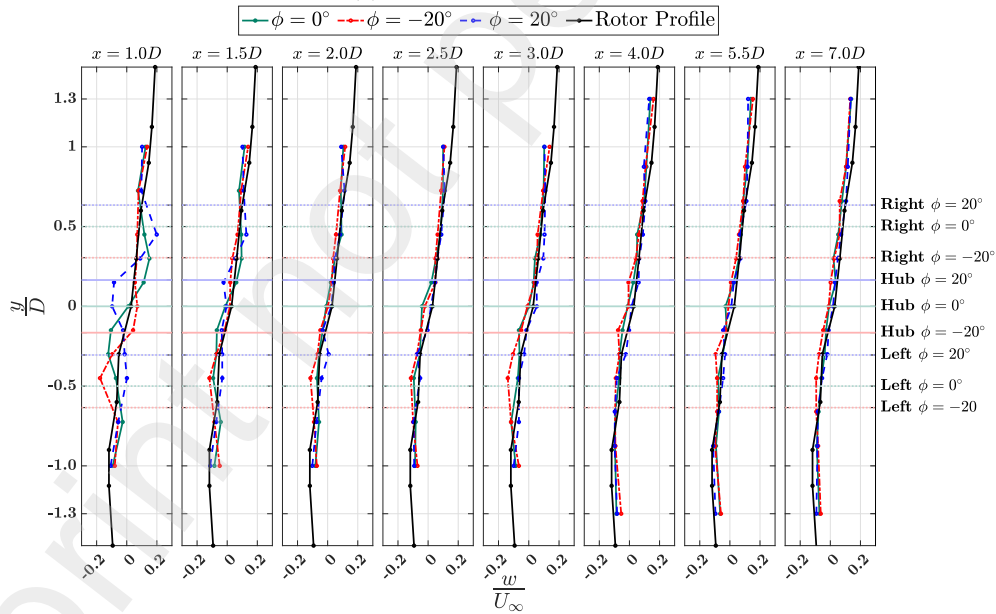
437

438 Self similarity has been used previously in the context of tidal energy [39]
439 and is the source of on-going research in the pursuit of improved optimisation
440 of the array structures of tidal farms [34] [40]. The advantage of imposing self
441 similarity on wake development is that the stream-wise velocity field in the
442 wake of tidal stream turbines can be expressed as a set of ordinary differential
443 equations rather than the full Navier-Stokes equations. Such a simplification
444 can be incredibly useful when trying to optimise array structures in realistic
445 channels.

446



(a) Vertical Velocity, LTI Case



(b) Vertical Velocity, HTI Case

Figure 7: Time averaged vertical flow velocity with downstream and cross-stream position for the (a) LTI cases and (b) HTI cases.

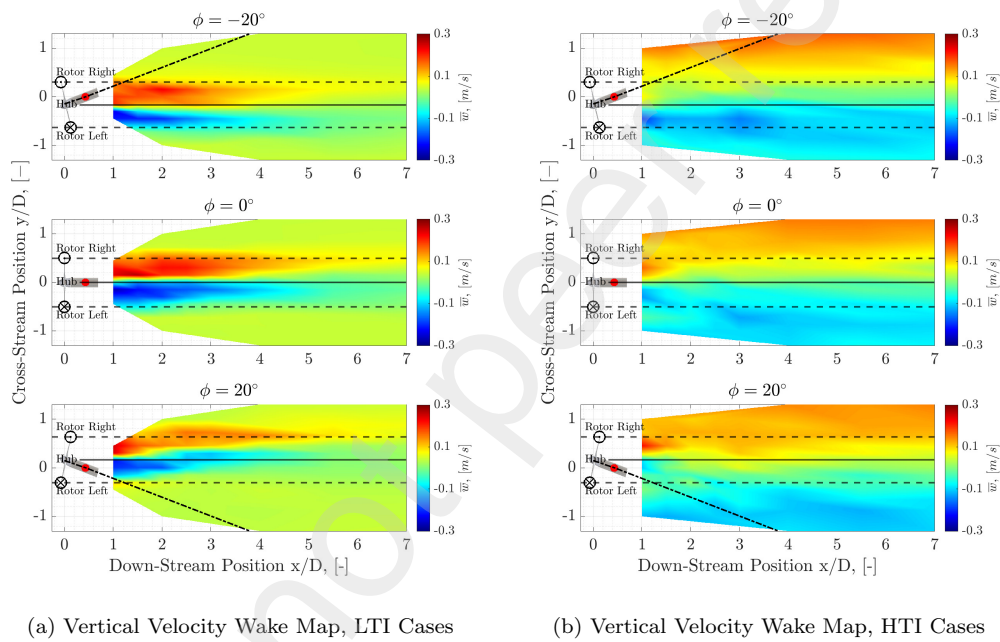


Figure 8: Time averaged Cross-stream flow velocity wake map with downstream and cross-stream position for the (a) LTI cases and (b) HTI cases. Dotted line projecting from the turbine sketch shows the turbine axis.

447 Here, we consider similarity for the stream-wise velocity field and consider
 448 the velocity field at a given horizontal plane (in this case the one vertically
 449 aligned with the hub centre). Self similarity is confirmed, if the stream-wise
 450 velocity, which is a function of x and y co-ordinates, can be normalized in
 451 such a way that it can be written as a function of a single non-dimensional
 452 cross-stream co-ordinate ξ .

453
 454 Equations 10 to 13 show the definitions required to normalise a plane or
 455 axisymmetric wake [41]. In Equation 10 the non-dimensional scale parame-
 456 ter, ξ , for the cross-stream position is defined. The parameter is generated by
 457 normalising the cross stream position by the distance from the cross-stream
 458 position associated with a characteristic velocity deficit to the point where
 459 the velocity deficit is half of the given characteristic deficit - Equation 11
 460 defines this characteristic length $y_{\frac{1}{2}}$. In case of axis-symmetric wakes the
 461 characteristic deficit is the deficit relative to the stream velocity, u_{∞} that
 462 occurs at y_{cl} as shown in Equation 12. Finally, we define the self similar
 463 velocity deficit in Equation 13, which is purely a function of ξ and has no
 464 dependence on x .

465

$$\xi = \frac{y}{y_{\frac{1}{2}}(x)} \quad (10)$$

$$\overline{u(x, \pm y_{\frac{1}{2}}, 0)} = u_{\infty} - \frac{1}{2}u_s(x) \quad (11)$$

$$u_s(x) = u_{\infty} - \overline{u(x, y_{cl}, 0)} \quad (12)$$

$$f(\xi) = \frac{u_{\infty} - \overline{u(x, y, 0)}}{u_s(x)} = e^{-\alpha\xi^2} \quad (13)$$

466 In developing a self similar description of the wakes measured, we note
 467 that the goal of the analysis is to search for approximate self similarity which
 468 may not be observable in the near wake - indeed some studies show consis-
 469 tent self similarity only occurs after $x/D = 8$, [34]. Furthermore, as a novel
 470 contribution and particular to yawed inflow, we define y_{cl} using a simple mo-
 471 mentum theory treatment of wake development under constant or steady yaw
 472 to give an estimate of the wake skew angle and following this the position of
 473 y_{cl} utilised in Equation 12. In this regard we use the results from Glauert's

474 analysis of the momentum through an auto-gyro to give the expression in
475 Equation 14 for the wake skew angle [42].

476

$$\chi = \phi(1 + 0.3(1 - \sqrt{1 - C_T})) \quad (14)$$

477 The results of this process are presented in Figure 9, where the upper
478 charts show the self similar deficit against normalised cross-stream position
479 for the LTI Cases and the lower charts show the same data for the HTI
480 cases. Each chart is compared with the analytical expression for the self
481 similar deficit as shown on the right hand side of Equation 13.

482

483 In both sets of charts the data shows some reasonable adherence to the
484 underlying self similar function given by the right hand side of Equation 13.
485 In the case of the LTI in-flow turbulence setting, it can be seen that the
486 largest deviations from self similarity were in the near to mid wake tran-
487 sects ($< 3x/D$) and in the bypass flow regions. The former statement shows
488 that the wake structure in the near to mid wake region is still dominated by
489 the turbine geometry and therefore y_{cl} is not necessarily the largest deficit
490 recorded as would be the case for purely plane or axisymmetric wakes - this
491 is why we observe values of $f(\xi) > 1$ for portions of these transects. The
492 second statement shows that, as expected and discussed in Section 2.6, the
493 constraints of the recirculating flume have an impact on the self similarity
494 findings outside of the turbine swept area where accelerated bypass flow leads
495 to deviations from the expected curve.

496

497 Perhaps surprisingly, relatively good adherence for all yaw angles was ob-
498 served for mid to far wake transects at the LTI setting. As expected very
499 good agreement with the normal self similar curve was found for the far wake
500 under the $\phi = 0^\circ$ case, particularly for $|\xi| < 1$. In the yawed flow cases there
501 where more deviations from the theoretical curves and the self similar wake
502 showed significant skewing in the bypass flow regions. It was also observed
503 that, the greatest adherence to the theoretical curve was for $x/D = 4.0$ and
504 $x/D = 5.5$ rather than $x/D = 7.0$ as expected - this is likely to show that
505 the wake centre-line deviates from the theoretical position based on the ex-
506 pression in Equation 14.

507

508 In the HTI inflow cases, the non-yawed case showed very good agreement

509 with the theoretical self-similar wake shape, other than in the outer regions,
510 again suggesting the impact of the upstream flow characteristics, the block-
511 age effect and the associated bypass flow acceleration. Furthermore, in the no
512 yaw ($\phi = 0$) case the approximate adherence to self similarity was observed
513 for all transects. Similarly, both yawed inflow cases exhibited deviation from
514 the expected curves at value of $|\xi| > 1$ due to the bypass flow.

515

516 Interestingly, under both yawed flow cases there is poor agreement or a
517 lack of self similarity for transects in the near wake and far wake, but good
518 agreement or similarity in the mid wake. The authors suggest that this is
519 due to the dominance of the turbine geometry in the flow structure in the
520 near wake and deviation of the wake centre from $y_{cl} = x \times \tan(\chi - \phi)$ in the
521 far wake, as predicted by Equation 14.

522

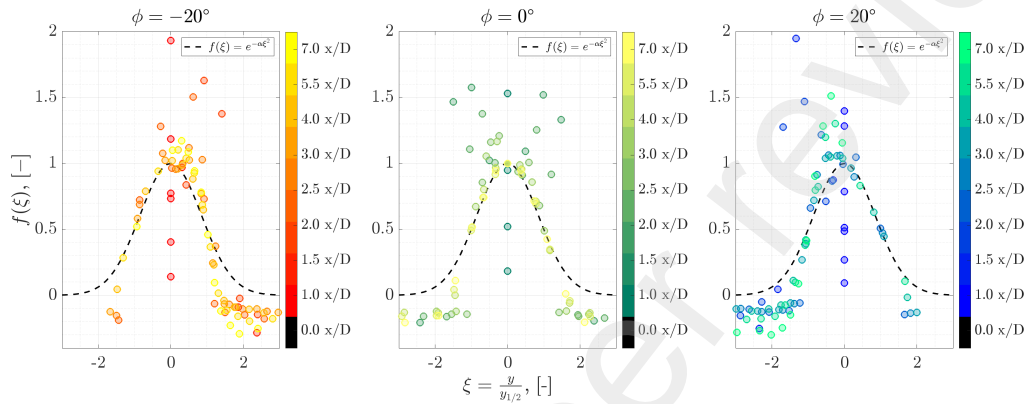
523 To study the wake progression in more detail and to understand the re-
524 gions where self similarity was less consistent, the value of y_{cl} was optimised
525 to give the best adherence to the analytical function given in Equation 13.
526 This was done by searching the cross-stream position to set y_{cl} to give the
527 smallest sum of squared error between the wake measurement points and the
528 aforementioned analytical function. The results of this process are shown in
529 Figure 10. Again the upper charts show the LTI cases and the lower charts
530 show the HTI cases. Under this new approach all cases show a good degree
531 of self similarity for transects in the mid to far wake regions i.e. $x/D > 3$ -
532 with the exception of the points in the outer wake, $|\xi| > 1$, again as a result
533 of the by-pass flow. The development of the centre-line of the wakes, as cal-
534 culated utilising the momentum theory and via the optimisation described
535 is presented in Figure 11.

536

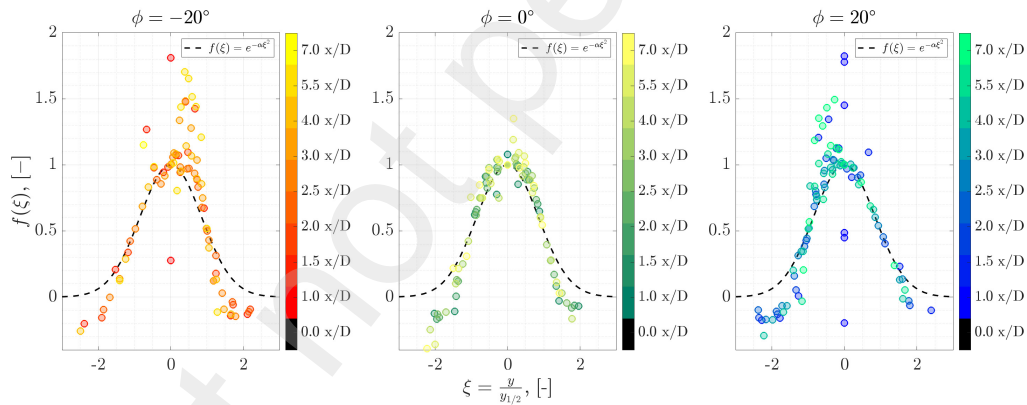
537 Figure 11 shows that the momentum theory predicts that the wakes
538 should have a slightly greater skewed angle for the LTI cases. This increase
539 in the wake skew angle for LTI is resultant from the dependence on C_T in
540 Equation 14 which is higher under the LTI cases - here we note the agreement
541 with other findings where the average C_T value is inversely impacted by the
542 turbulence level in the flow, [27].

543

544 The optimised traces immediately show that viewing the wake centre from
545 the point of optimised self-similarity reveals that the extent of the wake pro-
546 gression along the skew angle is curtailed by the bounded flow and resultant



(a) Normalised velocity deficit against normalized wake width, LTI Cases



(b) Normalised velocity deficit against normalized wake width, HTI Cases

Figure 9: Self Similar wake parameters for the stream-wise deficit recorded in the wake of the turbine under high and low turbulence intensities, in this case the value of y_{cl} was generated using momentum theory to define the wake skew angle, as detailed in Equation 14.

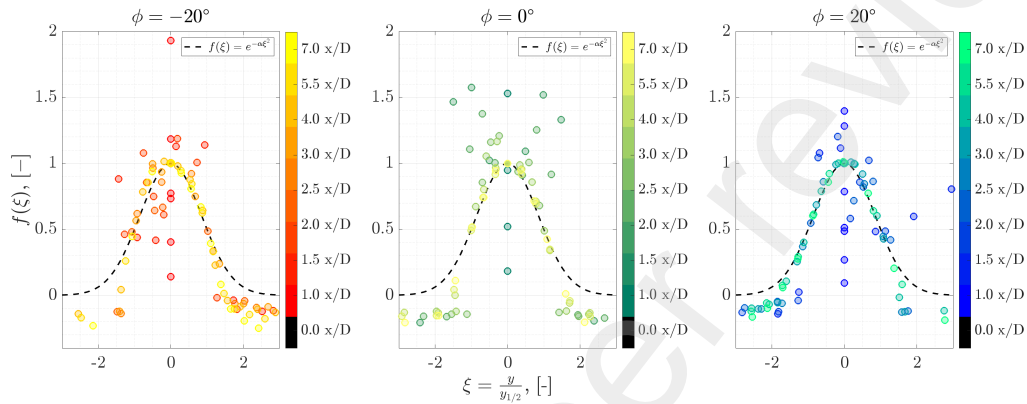
547 bypass flow. In the LTI cases, the overall wake progression, after the near
548 wake region, agrees with the theoretical wake centre-line progression in over-
549 all direction but not in magnitude. Again for the LTI cases, the wake centre
550 position estimates, generated through the optimisation process, are depend-
551 ent on downstream position and plateau to a fixed wake offset of $\pm 0.45y/D$.
552 The wake centre line progression calculated via the optimised self-similarity
553 approach shows an overall symmetry about the flume centre under the LTI
554 inflow conditions.

555
556 Under the HTI cases a more complex picture was observed via the self
557 similarity optimisation, where the wake centre meanders across the turbine
558 centre in the far wake such that the final position of wake centre line esti-
559 mated was contrary to the momentum theory position in both direction and
560 extent. In the HTI cases this complexity is compounded by an asymmetry
561 observed between the $\phi = 20^\circ$ and $\phi = -20^\circ$ cases, where in the $\phi = -20^\circ$
562 case we see the wake centre, calculated via the optimisation, meander across
563 the turbine centre line at a lower downstream position ($\frac{x}{D} = 3$) than the
564 $\phi = 20^\circ$ case ($\frac{x}{D} = 5.5$). Whilst it is not clear why this wake meander-
565 ing occurs the authors suggest that the impact of the constrained flow is a
566 prominent aspect in "straightening" the wake and potentially reversing its
567 lateral trajectory. The authors also note that an imbalance in the by-pass
568 flow shown in Figure 5, where in the HTI cases, a higher by-pass flow was
569 observed on the left hand side of the rotor. This is most evident in the $\phi = 0^\circ$
570 plot but can be seen in both the $\phi = 20^\circ$, where the imbalance is exacerbated
571 by the yawed device and the $\phi = -20^\circ$ where the expected symmetry with
572 the $\phi = 20^\circ$ was not observed.

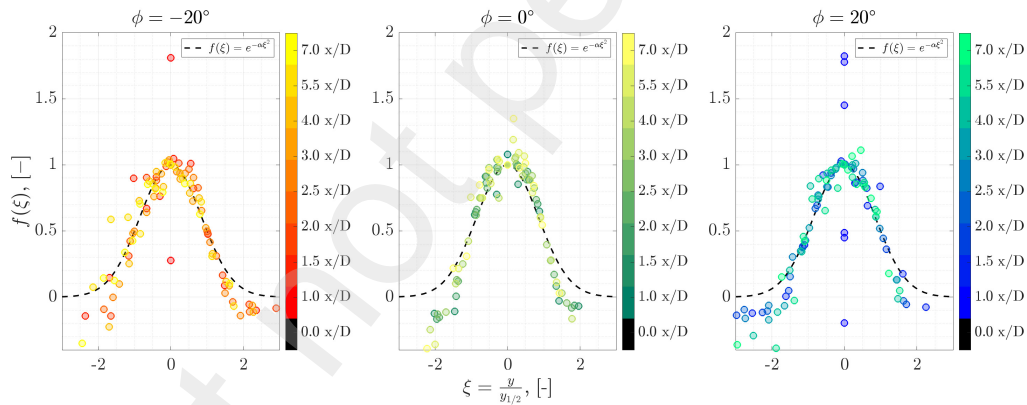
574 3.3. Turbulent Kinetic Energy and Dissipation Rate

575 In the following subsection we discuss the nature of the turbulence gen-
576 erated within the turbine wake, the turbulence in the wake will help further
577 mixing of the by-pass flow and has implications for the use of turbulence
578 closure models utilised in CFD.

579 Firstly we consider the Turbulent Kinetic Energy (k) development in the
580 wake. Turbulent kinetic energy, k , is defined in Equation 15, where the
581 overline signifies taking the time average for a given measurement position
582 as given in Equation 1. The balance of the kinetic energy of the turbulent
583 fluctuations induced by the turbine rotor, as described by the value k , is



(a) Normalised velocity deficit against normalized wake width, LTI Cases



(b) Normalised velocity deficit against normalized wake width, LTI Cases

Figure 10: Self Similar wake parameters for the stream-wise deficit recorded in the wake of the turbine under high and low turbulence intensities, in this case the value of y_{cl} was generated using by minimising the sum of squared error between the theoretical value of $f(\xi)$ and the measured values.

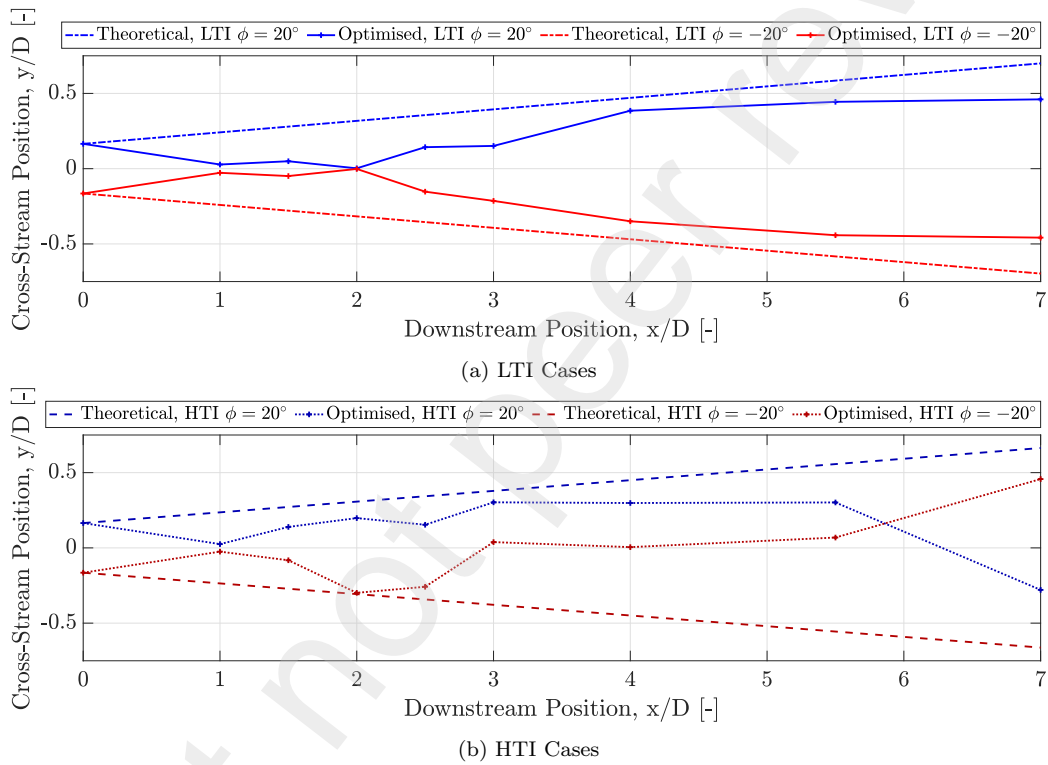


Figure 11: Chart showing the wake centre-lines as calculated using the momentum theory and by optimising the adherence of the wake to the self similar function $f(\xi) = e^{-\alpha\xi}$.

584 crucial to understanding the physical processes in the wake and particularly
585 important in turbulence modelling. Figure 12 shows the turbulent kinetic
586 energy, k , measured via the 3-component 3D LDV for each of the cases tested,
587 Figure 12a shows the LTI cases and Figure 12b shows the HTI cases - where k
588 is shown with cross-stream position (y/D , x-axis) and downstream position
589 (x/D , plot colours).

$$k = \frac{1}{2} \overline{(u'^2 + v'^2 + w'^2)} \quad (15)$$

590 Figure 12a shows the measured k for each LDV measurement position
591 and yaw case, immediately apparent, and unsurprisingly, the turbine rotor
592 is converting energy from the inflow to turbulent energy in the wake which
593 is not fully dissipated by $x/D = 7$ - as seen by comparing the black trace
594 to coloured traces. In the near wake, $x/D = 1$, the highest level of k is not
595 reached meaning there is further turbulence production moving downstream
596 of the rotor. In this region ($1 \leq x/D \leq 1.5$) the peak level of turbulent
597 kinetic energy produced was dependent on the yaw angle and peak k values
598 of $0.035 \text{ m}^2/\text{s}^2$, $0.047 \text{ m}^2/\text{s}^2$ and $0.070 \text{ m}^2/\text{s}^2$ were observed for the yaw
599 cases, $\phi = -20^\circ$, $\phi = 0^\circ$ and $\phi = 20^\circ$, respectively. In this region the early
600 wake appeared to be dominated by the merged effects of the nacelle and
601 shadowed blade-tip as discussed in Section 3.1 - the aforementioned peaks
602 in k values occur between the hub and right blade for the $\phi = -20^\circ$ case,
603 between the hub and left blade for the $\phi = 20^\circ$ case and finally directly
604 behind the hub for the $\phi = 0^\circ$ case. It should be noted here that the reduced
605 set of measurement positions for the LTI cases may have meant increases in
606 k due to high shear regions around blade tips which may have been missed
607 in this near wake region - particularly for the $\phi = 0^\circ$ case, where no increase
608 in k at the blade tip regions was found.

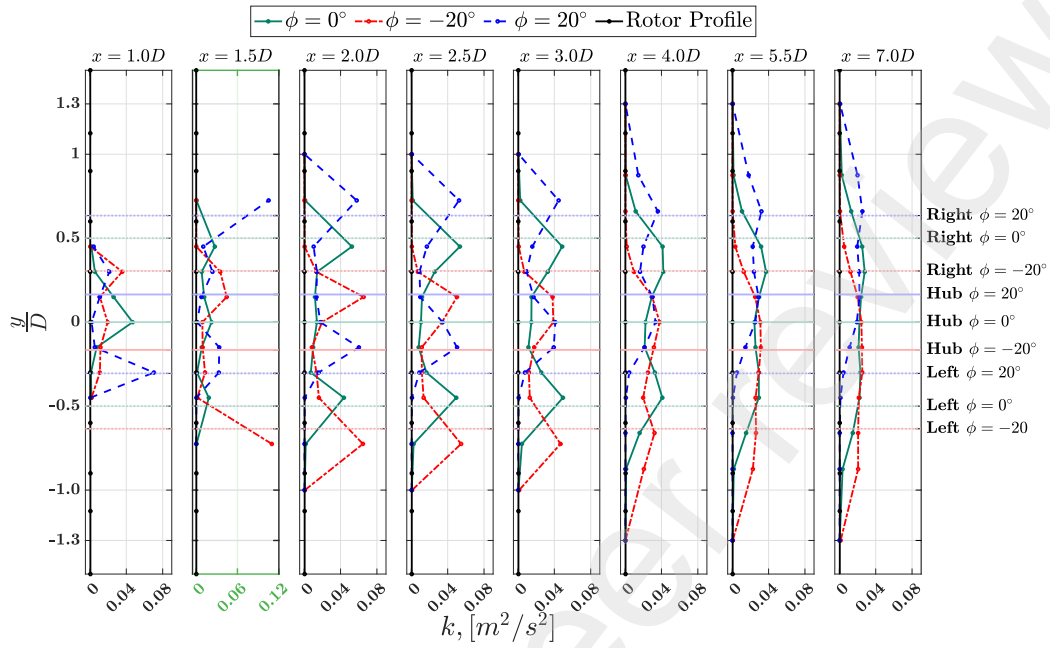
609 Further downstream, the k in the wake is highest in the high shear regions
610 as expected, this resulted in three peaked regions at the hub/nacelle and
611 blade tips (left and right) for the $\phi = 0^\circ$ case. The peaks associated with
612 the right blade and the nacelle for the $\phi = -20^\circ$ case are merged as were
613 the peaks associated with left blade and the nacelle for the $\phi = 20^\circ$ case. In
614 each of the yaw cases the k value associated with the non-shadowing blade
615 position at $x/D = 1.5$ exhibited the highest k value - this peak was dissipated
616 or diffused by $x/D = 2$. In each of the yaw cases the turbulent kinetic energy
617 distribution is asymmetric with the position of the blade peaks moving from
618 outside the turbine radius towards the respective blade tip positions (left:

619 $\phi = -20^\circ$ & right: $\phi = 20^\circ$). Finally, at $x/D = 7$ slightly higher peak
620 turbulent kinetic energy was observed for the no yaw case, relative to the
621 yawed cases.

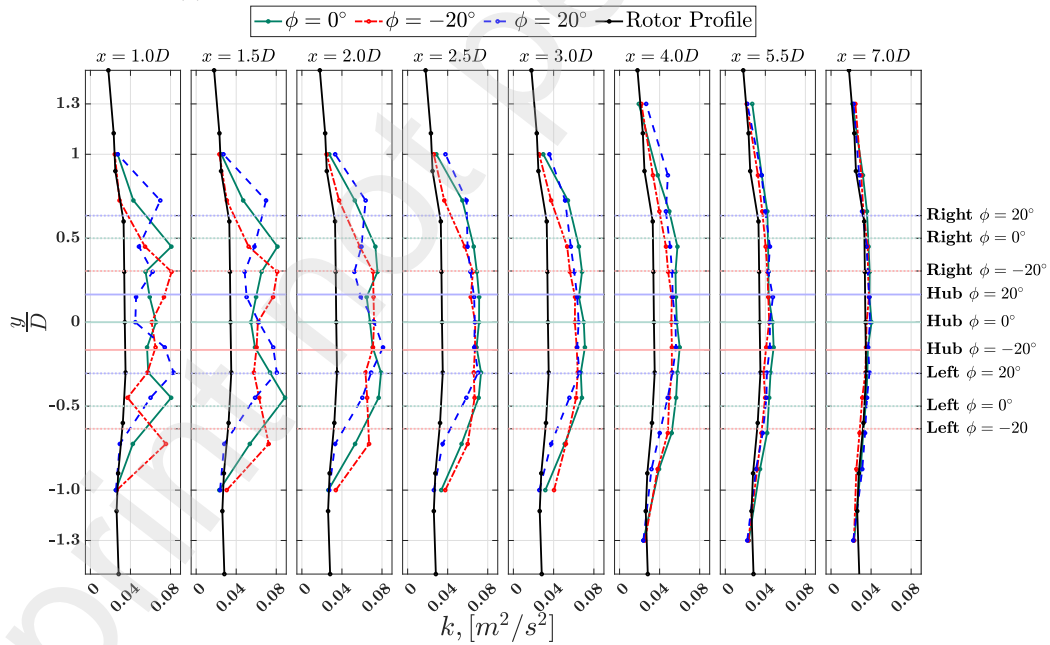
622 Figure 12b shows that in the HTI case the turbine also converted some
623 of the energy in the free stream flow into turbulent kinetic energy and in
624 the HTI case this energy was found to have mostly dissipated by $x/D = 7$ -
625 albeit towards a higher ambient k -value than the LTI cases, as expected. In
626 the near wake, $x/D < 2$, the peaks associated with the higher shear regions
627 created by the blade tips and hubs/nacelles can be seen, along with the
628 aforementioned merging of the shadowed blade tip and hub/nacelle effects in
629 both yawed cases, these effects are dissipated by $x/D = 4$ in all cases. The
630 generally higher levels of k , observed for the HTI cases, coincides with the
631 findings that the effects of shear structures and vertices are broken down at
632 a faster rate when the ambient or free stream turbulence is higher. Finally,
633 whilst the obvious structures associated with the blade tip regions and nacelle
634 are broken down more quickly in the HTI case, there is an overall asymmetry
635 in the turbulence kinetic energy introduced by yawing the turbine. These
636 asymmetries were found to persist for all downstream positions measured.
637 For each yaw case, k was asymmetric in the sense of higher relative k values
638 observed in the wake of the combined nacelle and blade extreme position -
639 i.e. $y/D = 0.3$ in the $\phi = -20^\circ$ case and $y/D = -0.3$ in the $\phi = 20^\circ$ case.

640 Closely related to the level of turbulent kinetic energy, k , throughout the
641 wake is the rate at which turbulence is converted to heat through molecular
642 interactions - the turbulence dissipation rate ϵ . Generally, experiments have
643 shown that ϵ is correlated to the k - indeed frequently it is considered that the
644 rate of turbulence production is equal to the rate of turbulence dissipation
645 for turbulent flows in equilibrium [43] [44]. Furthermore, assumptions on the
646 nature and transport of both k and ϵ are utilised in CFD models to generate
647 relationships for closing the Navier-Stokes relationships, for example in the
648 commonly used k - ϵ approach to modelling turbulence which is very popular
649 in modelling tidal stream devices [44].

650 The turbulence dissipation rate, ϵ , is calculated from LDV measurements
651 by assuming the turbulence spectra measured are of the form $C_0 \mathcal{K}^\beta$, as shown
652 by the approximate relation in Equation 16. Here, \mathcal{K} is the wave number
653 given by $\mathcal{K} = \frac{2\pi}{l}$, where l is an arbitrary length scale of the turbulent flow
654 and in the case of the spectra proceeds from larger length scales to smaller
655 length scales with increasing \mathcal{K} - in this case the wave number index for the
656 spectra were generated using Taylor's hypothesis that motion of each length



(a) Time-Averaged turbulent kinetic energy recorded for the LTI Cases



(b) Time-Averaged turbulent kinetic energy recorded for the HTI Cases

Figure 12: Turbulent kinetic energy, k , with downstream and cross-stream position for the (a) LTI cases and (b) HTI cases.

657 scale past a given point is driven by the global average flow velocity, $\overline{U_\infty}$.
 658 This gives $\mathcal{K} = \frac{f}{\overline{U_\infty}}$, where f is the standard frequency index for a developed
 659 spectra.

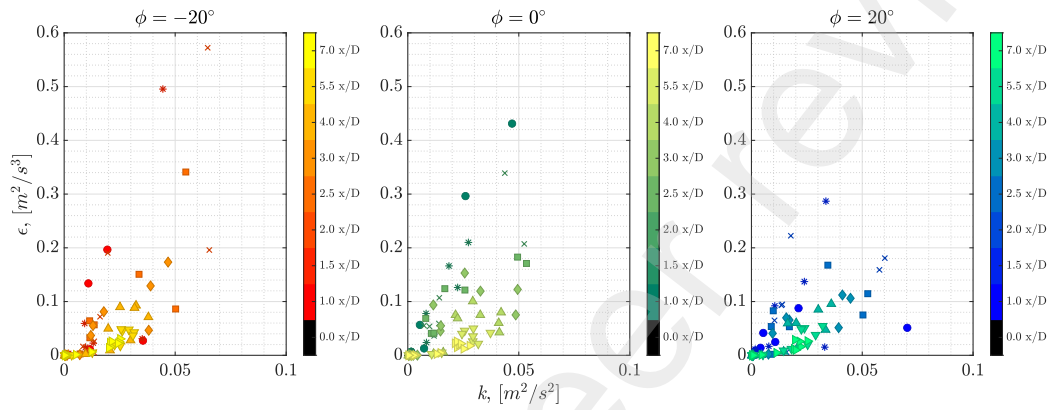
660 Then, using least squares regression the assumed function, $C_0\mathcal{K}^\beta$, is fitted
 661 to the data in the inertial range portion of the spectra yielding an estimate
 662 of C_0 assuming $\beta \approx -5/3$. The spectrum is related to ϵ by Kolomogoroff's
 663 2nd similarity hypothesis, which states that statistics of motions on scales
 664 within the inertial subrange are uniquely determined by ϵ . Such a hypothesis
 665 is expressed by the equality in Equation 16, where $C = 1.5$ as observed in
 666 experiments [45][46]. The right hand side of Equation 16 can be equated
 667 to the fitted form of the spectra using the value of C_0 generated through
 668 the least square regression and rearranged to give an estimate of ϵ given by
 669 Equation 17.

$$E(\mathcal{K}) \approx C_0\mathcal{K}^\beta = C\epsilon^{2/3}\mathcal{K}^\beta \quad (16)$$

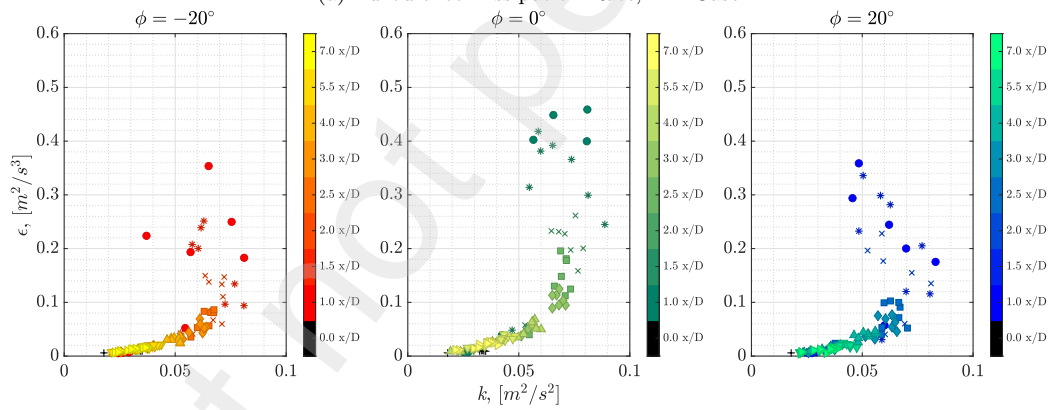
$$\epsilon = \left(\frac{C_0}{C}\right)^{3/2} \quad (17)$$

670 Figure 13a shows the relationship between k and ϵ and downstream position
 671 for the wake measurements made at LTI, whereas Figure 13a shows the
 672 same relationship for the HTI case. Figure 13a shows a reasonable correlation
 673 between the observed k values and the estimated dissipation rate, ϵ , as
 674 expected. In the near mid-wake the correlation is a little steeper indicating
 675 higher dissipation rates in regions in close proximity to the turbulence gener-
 676 ated by the turbine rotor, for these cases there were also slight variations
 677 between the yaw angle cases with a particularly steep gradient observed for
 678 the $\phi = 0^\circ$ case. In the far wake very similar results were observed across
 679 all cases and generally there was a reduction in the correlation between the
 680 turbulent energy and dissipation level in the flow with both tending towards
 681 zero for the transects measured furthest downstream.

682 Figure 13b, again, shows a clear correlation between turbulent energy
 683 and dissipation, this relationship is consistent for the far wake measurements
 684 ($\frac{x}{D} > 4$) and shows a minimal level of turbulent dissipation, inline with
 685 the turbulence generated by the overall flow regime. In the near to mid
 686 wake region ($1 < \frac{x}{D} < 4$) a non-linear relationship develops. In the $\phi = 0^\circ$
 687 case there is an almost asymptotic relationship whereby small increases in
 688 turbulent energy result in large increases in dissipation rate, showing the local



(a) Turbulence Dissipation Rate, LTI Case



(b) Turbulence Dissipation Rate, HTI Case

Figure 13: Turbulent energy dissipation rate, ϵ , with downstream and cross-stream position for the (a) LTI cases and (b) HTI cases.

689 adaptation of dissipation to maintain equilibrium. The highest dissipation
690 rates were observed in the $\phi = 0^\circ$ case. Accompanying the lower dissipation
691 rates observed in the yawed inflow cases, are regions of high dissipation rate
692 with relatively low turbulent energy, creating an almost inverse trend - it is
693 not clear why this relationship develops but it is potentially caused by a lack
694 redistribution of turbulent energy via diffusion or convection. Finally, the
695 authors note the broad adherence within the figures to a polynomial form
696 which may be related to an expression that can be developed via dimensional
697 considerations relating, k and ϵ - such an expression is given in Equation 18,
698 where ν_T is the eddy viscosity which to generate the expression it is assumed
699 that, $\nu_T = f(\epsilon, k)$.

$$\epsilon = \frac{C_\mu k^2}{\nu_T} \quad (18)$$

700 This finding is more clearly adhered to at greater distances downstream
701 of the turbine. where as one moves downstream the return to isotropic tur-
702 bulence can be observed. Clearly then, the geometric structures associated
703 with the turbulence generated via the turbine rotor are broken down, when
704 moving away from the turbine rotor.

705 3.4. Integral Length Scale

706 To further integrate the structure of the turbulence generated in the tur-
707 bine wake, an assessment of the integral length scale, denoted as \mathcal{L} , linked
708 to the stream-wise velocity fluctuations was conducted. This analysis aimed
709 to gain insights into how the large-scale turbulent structures evolved with
710 cross-stream and downstream position.

711 Figure 14a and 14b shows the \mathcal{L} for the LTI and HTI cases, respectively for
712 each of the yaw cases. In the LTI case, Figure 14a, a clear reduction in length
713 scale can be viewed in the near wake which is of a similar size ($\mathcal{L} < 0.1m$)
714 for each of the yaw cases. The length scale then grows with downstream
715 distance to $\mathcal{L} \approx 0.2m$ at $\frac{x}{D} = 7.0$, where the length scale remains significantly
716 smaller than observed upstream of the turbine. Here, the consistent growth
717 of turbulence length scale with downstream distance is in agreement with
718 other similar studies and shows the complex interaction of unfurling vortical
719 structures, the inertial sub-range where eddies are broken down and the
720 ultimate destruction of smaller eddies via viscous dissipation. In the bypass
721 flow, ($-0.5 < \frac{y}{D} < 0.5$), length scales of a similar size $0.5 < \mathcal{L} < 1$ can be
722 observed.

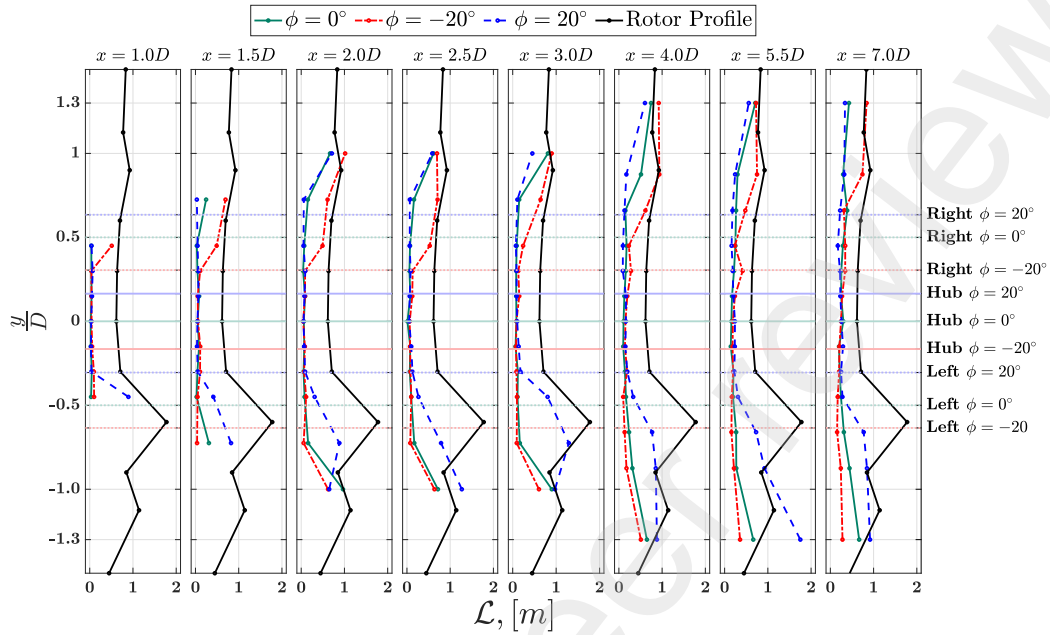
723 Under the HTI cases, a reduction in length scale can be seen in the
724 transects below $\frac{x}{D} = 2$, where in the $\frac{x}{D} = 1.0$ transects, length scale in the
725 region of $0.25m$ were observed. The higher ambient turbulence throughout
726 the flow has sped up the elongation of measured lengths scale via more rapid
727 unfurling and mixing of larger flow structures. Very quickly a chaotic picture
728 ensues with lengths scales in the order of the those found at the rotor plane,
729 this is true for all yaw cases after $\frac{x}{D} > 2.0$.

730 3.5. Turbulence Isotropy

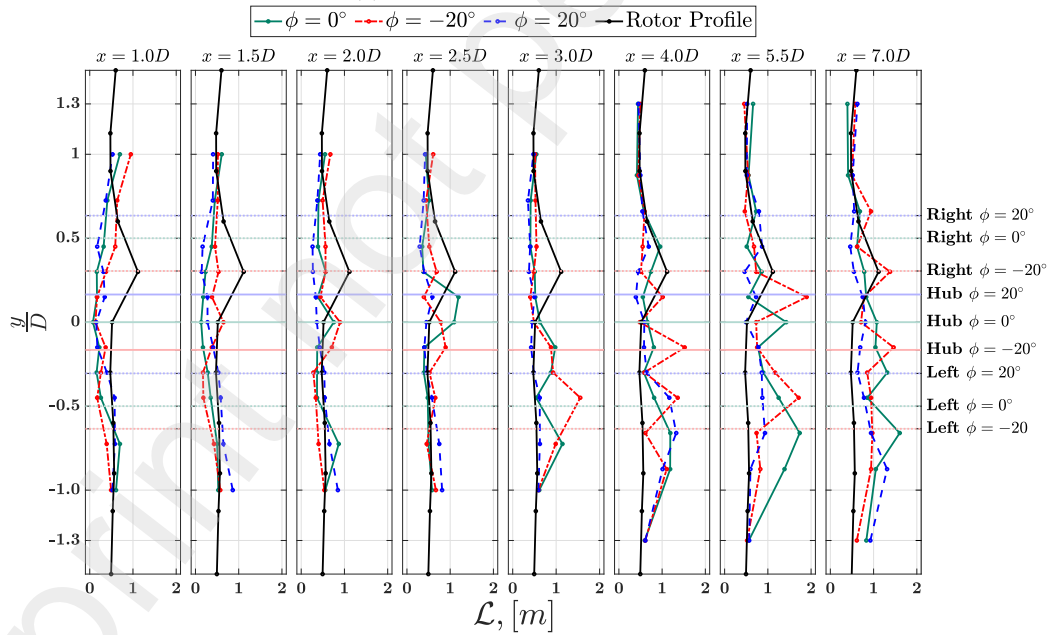
731 The final aspect of the analysis presented is to consider the anisotropy
732 of turbulence in the wake under the influence of yawed inflow at the LTI
733 and HTI test cases. The anisotropy is studied here utilising the 'Lumley
734 Triangle', developed by Choi and Lumley [47], which has been utilised in the
735 description of the turbulence in the wake of HATTs previously [12][48][49].
736 Anisotropy, can show the presence of coherent structures in turbulent flows
737 [50] and has an impact on the turbulence methods utilised within CFD mod-
738 els which often assume isotropic turbulence [12]. Understanding the return
739 to isotropy under the test cases measured will help develop an understanding
740 of how to appropriately model turbulence in the wake of HATTs and on the
741 propagation of coherent structures such as tip vertices.

742 The turbulence triangle is derived by considering the anisotropy of the
743 Reynolds stresses, $\overline{u'_i u'_j}$, which can be characterised via the anisotropy tensor,
744 b_{ij} given by Equation 19, where k is defined as shown previously in Equation
745 15 and the indices i and j indicate differing Cartesian directions x, y, z asso-
746 ciated with the velocity components (u, v, w) and finally, δ_{ij} is the Kronecker
747 delta where $\delta_{ij} = 1$, if $i = j$ and $\delta_{ij} = 0$, otherwise. Here, b_{ij} is a 2nd rank ten-
748 sor, which is symmetric and has zero trace and tends to zero when turbulence
749 becomes isotropic [47]. The anisotropy tensor being of rank 2, has a set of
750 three principle invariants which by definition do not change under a change
751 of basis, these invariants (I , II , and III), are defined in Equation 20, where
752 λ_1 , λ_2 and λ_3 are the eigenvalues associated with the eigenvector-eigenvalue
753 decomposition of the anisotropy tensor.

754 These invariants can then be utilised to define the boundaries of the
755 turbulent triangle via variables ξ and η defined in Equation 21. Figure 15,
756 adapted from [49], shows how the values of ξ and η can be interpreted.
757 The curved 'base' of the Lumley triangle adheres to the following non-linear
758 relationship, $\eta^2 = 1/27 + 2\xi^3$ and shows where turbulent structures become 2-
759 dimensional tending towards 1-dimensional at $\eta = \xi = 1/3$. Moving from the



(a) Integral Length Scale, LTI Case



(b) Integral Length Scale, HTI Case

Figure 14: Integral length scale, \mathcal{L} , with downstream and cross-stream position for the (a) LTI cases and (b) HTI cases.

760 origin, the sides of the triangle show axisymmetric turbulence where $\xi = -\eta$
 761 (tending towards oblate or squashed) and $\xi = \eta$ (tending towards prolate or
 762 elongated structures).

$$b_{ij} = \frac{\overline{u'_i u'_j}}{2k} - \frac{1}{3} \delta_{ij} \quad (19)$$

$$\begin{aligned} I &= 0 = \lambda_1 + \lambda_2 + \lambda_3 \\ II &= \frac{-b_{ij} b_{ji}}{2} = \lambda_1^2 + \lambda_1 \lambda_2 + \lambda_2^2 \\ III &= \frac{b_{ij} b_{jk} b_{ki}}{3} = -\lambda_1 \lambda_2 (\lambda_1 + \lambda_2) \end{aligned} \quad (20)$$

$$\begin{aligned} \xi^3 &= \frac{III}{2} \\ \eta^2 &= \frac{-II}{3} \end{aligned} \quad (21)$$

763 Figures 16a and 16b show the Lumley triangles for the wake measured
 764 under each condition. In the LTI case the inflow turbulence is minimal and
 765 can be seen to have a prolate spheroid shape tending towards 2-D axisym-
 766 metric turbulence. The effect of the rotor was to generate oblate spheroid
 767 structures which were not present in the flow at the rotor plane prior to in-
 768 stallation. No clear trajectory towards isotropic turbulence can be observed
 769 and with downstream distance from the rotor one can see the flow tends
 770 towards 2-D asymmetric elongated turbulent structures as observed prior to
 771 turbine installation - however this tendency is not consistent across the differ-
 772 ing yaw cases. The finding would suggest that the inlet turbulence requires
 773 far greater mixing to developed isotropic turbulence at the inlet to the rotor.

774 Under the HTI case (Figure 16b) a more clear trajectory tending towards
 775 isotropic turbulence was observed with downstream distance, however perfect
 776 isotropic turbulence was not observed, neither prior to installation or in the
 777 far wake. In the $\phi = -20^\circ$ case the yawed rotor has generated clearly 2-
 778 D axisymmetric elongated turbulence which is dissipated with downstream
 779 position. Similar structures were introduced by the rotor in the $\phi = 0^\circ$ case
 780 and the $\phi = 20^\circ$ case but not to the same degree. In all instances a minority of
 781 measurement positions exhibited 2-D axisymmetric structures of the oblate

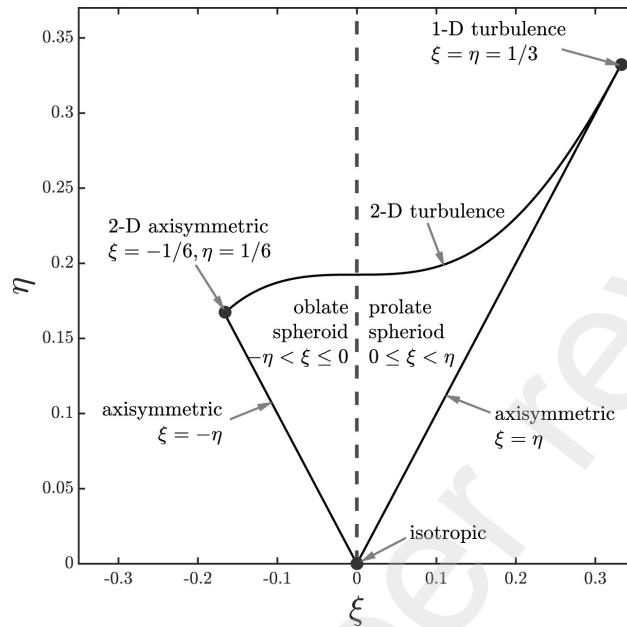
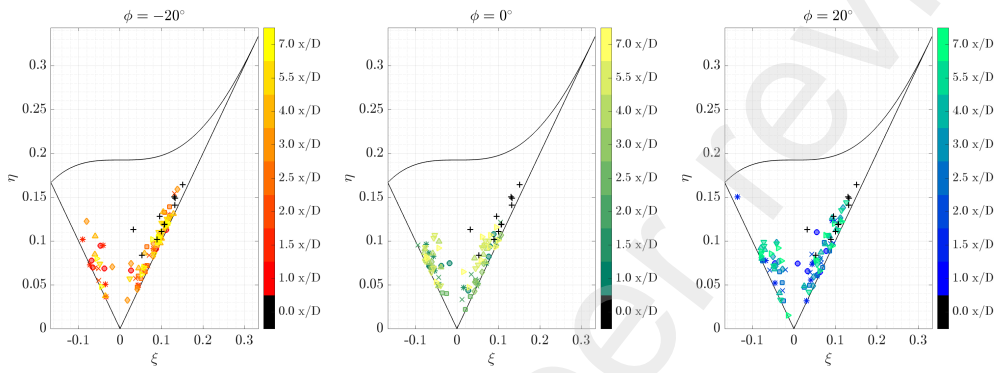


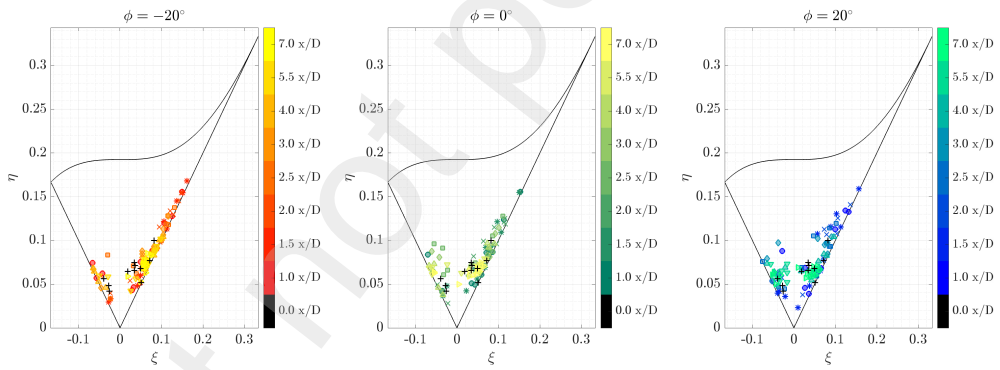
Figure 15: Schematic of the interpretation of the Lumley triangle utilised to analyse the shape of the turbulent structures in the wake of the HATT. [49]

782 spheroid shape, this was more pronounced in the $\phi = 0^\circ$ and $\phi = 20^\circ$ cases.
 783 As expected, due to observations in [12] [48] and [49], in the near wake
 784 elongated axis symmetric turbulence is generated by the rotor. Again, inline
 785 with previous studies no fully isotropic turbulence was observed by $x/D =$
 786 7.0. However, the authors note the varying levels and characteristics of the
 787 anisotropy observed between the three experiments and those presented here,
 788 suggesting that the inflow characteristics do have an impact on the findings.

789 To understanding in more detail the return to anisotropy, Figures 17 and
 790 18 show the principle values of the anisotropy tensor, b_{ii} - the upper charts
 791 shows the values inline with the rotor right side, the central charts show the
 792 values inline with hub and the lower charts show the values aligned with the
 793 left side of the rotor. The principle values show the degree of anisotropy in
 794 each principle direction (x, y and z) and sum to zero, as the trace of the
 795 anisotropy tensor is zero - Equation 20. Figures 17 and 18 show comparable
 796 values to both [47] and [48]. We immediately see that the development of
 797 the anisotropy with downstream position in the turbulent wakes is a function
 798 of inflow conditions (difference between Figures 17 and 18), cross-stream
 799 position (difference between upper, middle and lower charts) and yaw angle



(a) LTI Case.



(b) HTI Case.

Figure 16: Lumley triangles showing anisotropy of turbulence structures with downstream position.

800 (difference between differing coloured markers).

801 Generally, the trajectory of the principles of the anisotropy tensor are
802 rather chaotic, this is particularly true under the LTI cases. Under the LTI
803 cases, each principle axis shows significant levels of anisotropy and one can
804 only roughly observe the tendency for each principle value to converge to a
805 fixed level with downstream position. Under the HTI cases more clear tra-
806 jectories can be observed - each principle value seems to converge to a given
807 level. Furthermore, in the HTI case the third principle, b_{33} (z-direction), has
808 a markedly smaller level than b_{11} and b_{22} .

809 Whilst the variations in the figures are rather chaotic, there is consis-
810 tency across both inflow conditions for the $\phi = 0^\circ$ cases - here, the data
811 at either blade extreme is similar across each principle direction (upper and
812 lower charts are similar for all three lateral charts), suggesting a level of
813 symmetry about the rotational axis. This is in contrast with the principle
814 values recorded behind the hub and nacelle (central row of charts) which
815 interestingly show almost opposite tendencies to the blade extremes (this is
816 particularly clear in Figure 18).

817 In the $\phi = -20^\circ$ and $\phi = 20^\circ$ cases, very similar trajectories for the
818 anisotropy in all principle directions was observed behind the hub. Whereas
819 almost inverted roles were observed at the blade extremes - i.e. for b_{11} (x-
820 direction) the upstream blade generates a greater level of anisotropy such
821 that in the upper charts for $\phi = -20^\circ$ higher values were observed and the
822 opposite in the lower charts.

823 4. Conclusions

824 This paper has discussed measurements of the flow velocity in the near to
825 mid wake of a HATT under laboratory conditions utilising the re-circulating
826 flume at IFREMER, Boulogne-sur-Mer, France. A 3D LDV system was used
827 to take point measurements of the flow at a pre-configured series of positions
828 in the turbine wake. Here we analyse the wake structure in the horizontal
829 plane aligned with the turbine rotor. The wakes were measured for three
830 turbine yaw angles as defined in Figure 2 for two differing TI inflow settings.

831 The flow velocity deficit in the stream-wise direction showed that in-
832 deed the yawing of the turbine introduced a skewed wake. In the LTI cases
833 marginally faster recovery was observed for both the yawed cases but this
834 was not observed in the HTI cases. The rate of wake recovery was signif-
835 icantly higher in the HTI cases than in the LTI cases which is inline with

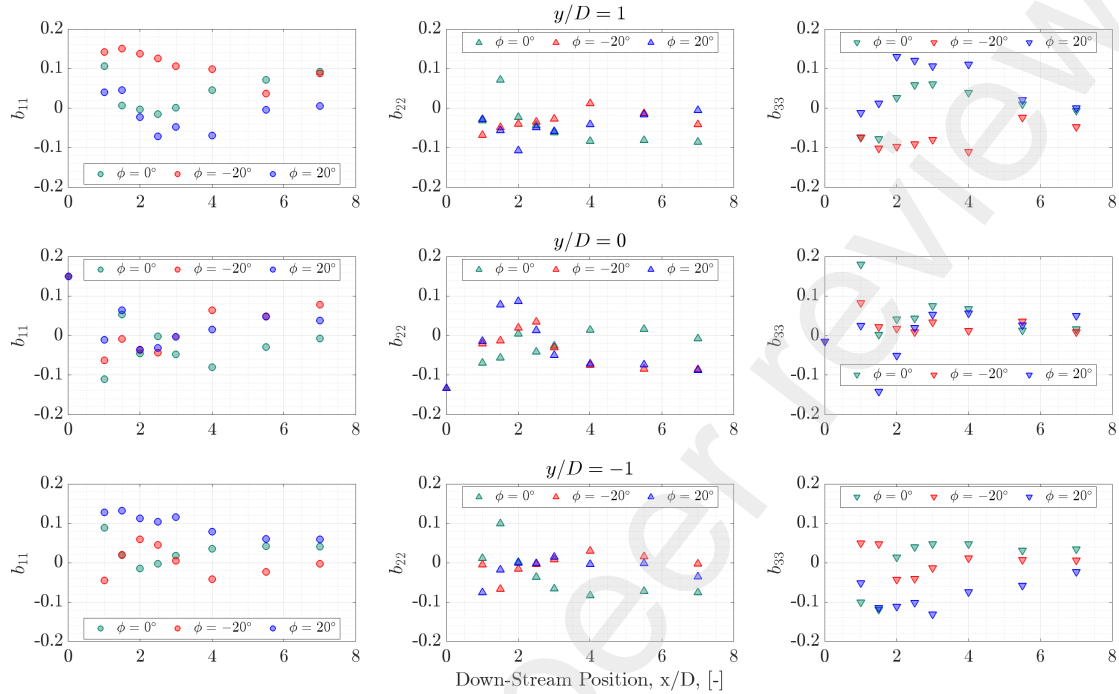


Figure 17: Lumley triangle showing anisotropy of turbulence structures with downstream and cross-stream position for the LTI cases.

836 other findings. The degree of deficit in the wakes observed was inline with
 837 other experimental results also [12][9][48]. Consideration of the vertical ve-
 838 locity induced by the rotor, structures similar to Rankine or Lamb-Oseen
 839 [51] vortexes were observed, these were particularly clear for the LTI but
 840 were obscured by a flume wide vorticity which masked these findings for
 841 measurement positions $x/D > 1.5D$ in the HTI case.

842 A degree of self similarity was observed for all wakes when combining self-
 843 similarity theory and skew angle predictions developed by Glauert [42]. The
 844 self similarity curves generated were generally very consistent for the no yaw
 845 cases ($\phi = 0^\circ$) under both TI settings. Under the yawed cases the inclusion
 846 of the skew angle in setting the reference velocity for each transect facili-
 847 tated a self similarity analysis of the skewed wakes which gave reasonable
 848 results. An optimisation was performed to improve the adherence to the self
 849 similar wake shape by selecting the wake centre-line (and reference) position

850 (and value) that gave the best agreement to the aforementioned wake shape.
 851 This approach gave a method of inspecting the wake centre-line with down-
 852 stream position which was compared with the skewed wake centre predicted
 853 by Glauert. This process uncovered a complex picture showing wake mean-
 854 der - in some cases across the hub centre - but ultimately showed that the
 855 inability of the wake to fully expand, due to the flume walls, had a significant
 856 impact on the results. Even so, some evidence wake skewing and bending was
 857 observed and could be useful in array power output optimisations. However,
 858 further work is required in this regard, this work should either be undertaken
 859 numerically or by conducting tests with a smaller blockage ratio.

860 The turbulence in the wake was analysed in terms of energy levels, k ,
 861 length scale, \mathcal{L} , and anisotropy. The levels of turbulent kinetic energy in the
 862 wake k , were clearly elevated by the turbine and peaks in k were observed
 863 aligned with blade extremes. Interestingly, the peaks were larger for the

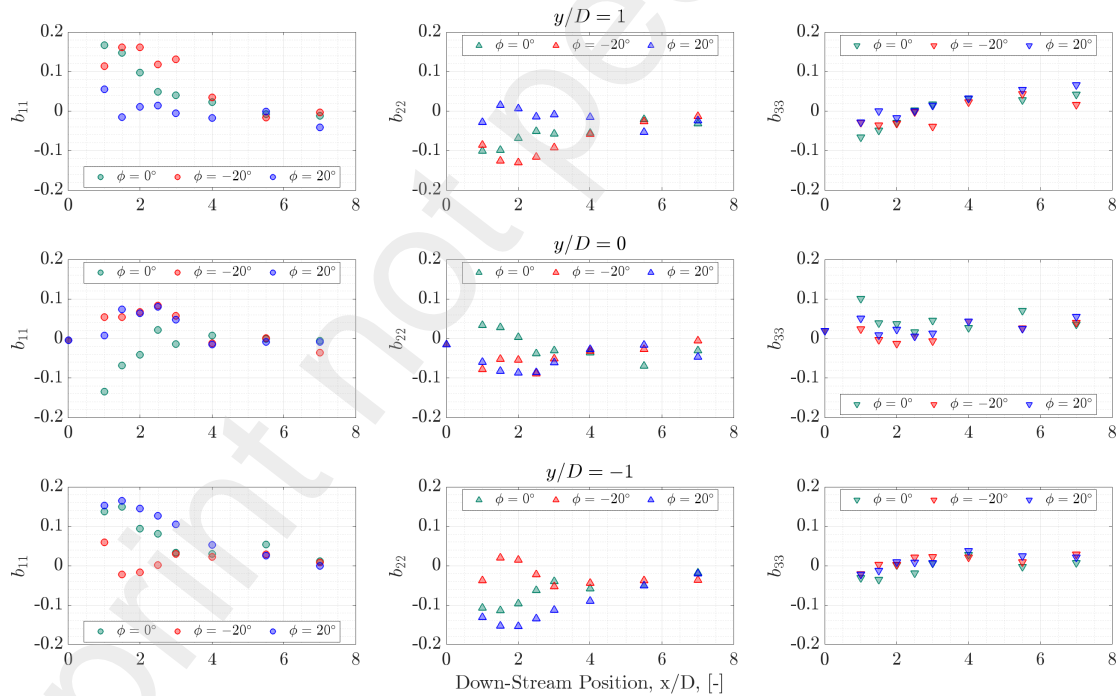


Figure 18: Lumley triangle showing anisotropy of turbulence structures with downstream and cross-stream position for the HTI cases.

864 LTI case than the HTI case with these high peak values only observed in
865 the yawed turbine cases. In all cases the peak levels of k increased between
866 $1D$ and $1.5D$ downstream of the turbine. The coherence of high levels of k
867 aligned with the blade extremes and the hub dissipated more slowly under
868 LTI inflow and quickly under HTI inflow. The relationship between, k and
869 dissipation rate ϵ was also explored and showed that in the mid wake a
870 parabolic relationship was observed but this was not the case in the near
871 wake. The parabolic relationship was observed after $x/D = 5.5$ and $x/D =$
872 2.0 for the LTI and HTI cases, respectively. This relationship was consistent
873 across all yaw settings and can be developed from dimensional reasoning [43].

874 Analysis of the length scales of the turbulent structures in the wake
875 showed evidence that the turbine generated smaller structures than the am-
876 bient inflow. In the HTI cases these smaller structures grew quickly to sizes
877 similar to the inflow turbulence. In the case of the LTI inflow the smaller
878 structure persisted for longer and full growth to the inlet scales was not ob-
879 served. Growing length scales in HATT wake turbulence has been observed
880 previously and our study supports these findings.

881 The presence of the turbine also changed the anisotropy of the turbulence
882 - the observed changes were sensitive to the level and type anisotropy ob-
883 served in the inflow. Changes in the anisotropy in the LTI cases was chaotic
884 and a return to elongated 2-D axisymmetric structures, observed in the in-
885 flow, was not clearly observed - this was further observed in the chaotic traces
886 of the anisotropy tensor principle values. More consistent development of the
887 tensor invariants and the principle values were observed in the HTI cases.

888 This work showed the complexity of the wake development and the poten-
889 tial for wake skewing and bending which maybe useful in array optimisation.
890 The turbulence within the wake is complex and dependent on the inflow tur-
891 bulence and its structure as well as the control strategy of the rotor, in this
892 case speed control. This latter point needs further investigation and may
893 well give some insight to some of the effects discussed.

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900 **Data**

901 The data underpinning this article can be accessed at: <https://doi.org/10.17035/d.2020.0124287>

902 **References**

- 903 [1] C. H. Frost, P. S. Evans, M. J. Harrold, A. Mason-Jones,
904 T. O'Doherty, D. M. O'Doherty, The impact of axial flow misalign-
905 ment on a tidal turbine, *Renewable Energy* 113 (2017) 1333–1344.
906 doi:10.1016/j.renene.2017.07.006.
- 907 [2] A. Bahaj, A. Molland, J. Chaplin, W. Batten, Power and thrust mea-
908 surements of marine current turbines under various hydrodynamic flow
909 conditions in a cavitation tunnel and a towing tank, *Renewable Energy*
910 32 (2007) 407–426. doi:10.1016/j.renene.2006.01.012.
- 911 [3] A. Bahaj, L. Myers, Shaping array design of marine current energy
912 converters through scaled experimental analysis, *Energy* 59 (2013) 83–
913 94. doi:10.1016/j.energy.2013.07.023.
- 914 [4] F. Maganga, G. Germain, J. King, G. Pinon, E. Rivoalen, Experimental
915 characterisation of flow effects on marine current turbine behaviour and
916 on its wake properties, *IET Renewable Power Generation* 4 (2010) 498.
917 doi:10.1049/iet-rpg.2009.0205.
- 918 [5] P. Mycek, B. Gaurier, G. Germain, G. Pinon, E. Rivoalen, Experimen-
919 tal study of the turbulence intensity effects on marine current turbines
920 behaviour. part i: One single turbine, *Renewable Energy* 66 (2014) 729–
921 746. doi:10.1016/j.renene.2013.12.036.
- 922 [6] B. Morandi, F. D. Felice, M. Costanzo, G. Romano, D. Dhomé, J. Allo,
923 Experimental investigation of the near wake of a horizontal axis tidal
924 current turbine, *International Journal of Marine Energy* 14 (2016) 229–
925 247. doi:10.1016/j.ijome.2016.02.004.
- 926 [7] P. Mycek, B. Gaurier, G. Germain, G. Pinon, E. Rivoalen, Experimental
927 study of the turbulence intensity effects on marine current turbines be-
928 haviour. part ii: Two interacting turbines, *Renewable Energy* 68 (2014)
929 876–892. doi:10.1016/j.renene.2013.12.048.

- 930 [8] M. Allmark, R. Ellis, T. Ebdon, C. Lloyd, S. Ordonez-Sanchez, R. Mar-
931 tinez, A. Mason-Jones, C. Johnstone, T. O'Doherty, A detailed study
932 of tidal turbine power production and dynamic loading under grid gen-
933 erated turbulence and turbine wake operation, *Renewable Energy* (12
934 2020). doi:10.1016/j.renene.2020.12.052.
- 935 [9] T. Ebdon, M. Allmark, D. O'Doherty, A. Mason-Jones, T. O'Doherty,
936 G. Germain, B. Gaurier, The impact of turbulence and turbine operating
937 condition on the wakes of tidal turbines, *Renewable Energy* 165 (2021).
938 doi:10.1016/j.renene.2020.11.065.
- 939 [10] Y. Chen, B. Lin, J. Lin, S. Wang, Experimental study of wake structure
940 behind a horizontal axis tidal stream turbine, *Applied Energy* 196 (2017)
941 82–96. doi:10.1016/J.APENERGY.2017.03.126.
- 942 [11] I. Afgan, J. McNaughton, S. Rolfo, D. Apsley, T. Stallard, P. Stansby,
943 Turbulent flow and loading on a tidal stream turbine by les and
944 rans, *International Journal of Heat and Fluid Flow* 43 (2013) 96–108.
945 doi:10.1016/j.ijheatfluidflow.2013.03.010.
- 946 [12] S. Tedds, I. Owen, R. Poole, Near-wake characteristics of a model hori-
947 zontal axis tidal stream turbine, *Renewable Energy* 63 (2014) 222–235.
948 doi:10.1016/j.renene.2013.09.011.
- 949 [13] T. Blackmore, W. M. Batten, A. S. Bahaj, Influence of turbulence on the
950 wake of a marine current turbine simulator, *Proceedings of the Royal*
951 *Society A: Mathematical, Physical and Engineering Sciences* 470 (10
952 2014). doi:10.1098/RSPA.2014.0331.
- 953 [14] Z. Zhang, Y. Zhang, J. Zhang, Y. Zheng, W. Zang, X. Lin, E. Fernandez-
954 Rodriguez, Experimental study of the wake homogeneity evolution be-
955 hind a horizontal axis tidal stream turbine, *Applied Ocean Research* 111
956 (6 2021).
- 957 [15] P. B. S. Lissaman, Energy effectiveness of arbitrary arrays of wind tur-
958 bines, *Journal of Energy* 3 (1979) 323–328. doi:10.2514/3.62441.
- 959 [16] N. Jensen, A note on wind generator interaction, no. 2411 in Risø-M,
960 Risø National Laboratory, 1983.

- 961 [17] S. P. O. R. S. L. J. H. M. T. Sten Frandsen, Rebecca Barthelmie, An-
962 analytical modelling of wind speed deficit in large offshore wind farms,
963 *Wind Energy* 9 (2006) 39–53. doi:10.1002/we.189.
- 964 [18] M. Bastankhah, F. Porté-Agel, A new analytical model for
965 wind-turbine wakes, *Renewable Energy* 70 (2014) 116–123.
966 doi:10.1016/j.renene.2014.01.002.
- 967 [19] W.-H. Lam, L. Chen, Equations used to predict the velocity distribu-
968 tion within a wake from a horizontal-axis tidal-current turbine, *Ocean*
969 *Engineering* 79 (2014) 35–42. doi:10.1016/j.oceaneng.2014.01.005.
- 970 [20] G. Pinon, P. Mycek, G. Germain, E. Rivoalen, Numerical simulation of
971 the wake of marine current turbines with a particle method, *Renewable*
972 *Energy* 46 (2012) 111–126. doi:10.1016/j.renene.2012.03.037.
- 973 [21] M. Allmark, R. Ellis, C. Lloyd, S. Ordonez-Sanchez, K. Johannesen,
974 C. Byrne, C. Johnstone, T. O’Doherty, A. Mason-Jones, The develop-
975 ment, design and characterisation of a scale model horizontal axis tidal
976 turbine for dynamic load quantification, *Renewable Energy* 156 (2020).
977 doi:10.1016/j.renene.2020.04.060.
- 978 [22] C. Lloyd, M. Allmark, S. Ordonez-Sanchez, R. Martinez, C. Johnstone,
979 G. Germain, B. Gaurier, A. Mason-Jones, T. O’Doherty, Validation of
980 the dynamic load characteristics on a tidal stream turbine when sub-
981 jected to wave and current interaction, *Ocean Engineering* 222 (2021).
982 doi:10.1016/j.oceaneng.2020.108360.
- 983 [23] S. Ordonez-Sanchez, M. Allmark, K. Porter, R. Ellis, C. Lloyd, I. Santic,
984 T. O’Doherty, C. Johnstone, Analysis of a horizontal-axis tidal turbine
985 performance in the presence of regular and irregular waves using two
986 control strategies, *Energies* 12 (2019). doi:10.3390/en12030367.
- 987 [24] R. Martinez, S. Ordonez-Sanchez, M. Allmark, C. Lloyd, T. O’Doherty,
988 G. Germain, B. Gaurier, C. Johnstone, Analysis of the effects of control
989 strategies and wave climates on the loading and performance of a labo-
990 ratory scale horizontal axis tidal turbine, *Ocean Engineering* 212 (2020)
991 107713. doi:10.1016/j.oceaneng.2020.107713.
- 992 [25] M. Allmark, R. Martinez, S. Ordonez-Sanchez, C. Lloyd, T. O’Doherty,
993 G. Germain, B. Gaurier, C. Johnstone, A phenomenological study of

- 994 lab-scale tidal turbine loading under combined irregular wave and shear
995 flow conditions, *Journal of Marine Science and Engineering* 9 (2021) 593.
996 doi:10.3390/jmse9060593.
- 997 [26] R. Ellis, M. Allmark, T. O’Doherty, A. Mason-Jones, S. Ordonez-
998 Sanchez, K. Johannesen, C. Johnstone, Design process for a scale hor-
999 izontal axis tidal turbine blade, in: 4th Asian Wave and Tidal Energy
1000 Conference, TWN, 2018.
1001 URL <https://strathprints.strath.ac.uk/65920/>
- 1002 [27] T. Blackmore, L. E. Myers, A. S. Bahaj, Effects of turbulence on
1003 tidal turbines: Implications to performance, blade loads, and condi-
1004 tion monitoring, *International Journal of Marine Energy* 14 (2016) 1–26.
1005 doi:10.1016/j.ijome.2016.04.017.
- 1006 [28] A. Mason-Jones, D. O’Doherty, C. Morris, T. O’Doherty, C. Byrne,
1007 P. Prickett, R. Grosvenor, I. Owen, S. Tedds, R. Poole, Non-
1008 dimensional scaling of tidal stream turbines, *Energy* 44 (2012) 820–829.
1009 doi:10.1016/j.energy.2012.05.010.
- 1010 [29] Y. Qian, Y. Zhang, Y. Sun, H. Zhang, Z. Zhang, C. Li, Experimental and
1011 numerical investigations on the performance and wake characteristics
1012 of a tidal turbine under yaw, *Ocean Engineering* 289 (2023) 116276.
1013 doi:10.1016/j.oceaneng.2023.116276.
- 1014 [30] J. Schluntz, R. H. Willden, The effect of blockage on tidal turbine
1015 rotor design and performance, *Renewable Energy* 81 (2015) 432–441.
1016 doi:10.1016/J.RENENE.2015.02.050.
- 1017 [31] J. I. WHELAN, J. M. R. GRAHAM, J. PEIRÓ, A free-surface and
1018 blockage correction for tidal turbines, *Journal of Fluid Mechanics* 624
1019 (2009) 281–291. doi:10.1017/S0022112009005916.
- 1020 [32] N. Kolekar, A. Vinod, A. Banerjee, On blockage effects for a
1021 tidal turbine in free surface proximity, *Energies* 12 (17) (2019).
1022 doi:10.3390/en12173325.
1023 URL <https://www.mdpi.com/1996-1073/12/17/3325>
- 1024 [33] H. Sarlak, T. Nishino, L. A. Martínez-Tossas, C. Meneveau, J. N.
1025 Sørensen, Assessment of blockage effects on the wake characteristics

- 1026 and power of wind turbines, *Renewable Energy* 93 (2016) 340–352.
1027 doi:10.1016/J.RENENE.2016.01.101.
- 1028 [34] P. Stansby, T. Stallard, Fast optimisation of tidal stream turbine posi-
1029 tions for power generation in small arrays with low blockage based on
1030 superposition of self-similar far-wake velocity deficit profiles, *Renewable*
1031 *Energy* 92 (2016) 366–375. doi:10.1016/j.renene.2016.02.019.
- 1032 [35] T. Burton, N. Jenkins, D. Sharpe, E. Bossanyi, *Wind Energy Handbook*,
1033 Wiley, 2011. doi:10.1002/9781119992714.
- 1034 [36] M. S. Uberoi, P. Freymuth, Turbulent energy balance and spectra of
1035 the axisymmetric wake, *The Physics of Fluids* 13 (1970) 2205–2210.
1036 doi:10.1063/1.1693225.
- 1037 [37] I. Wygnanski, F. Champagne, B. Marasli, On the large-scale structures
1038 in two-dimensional, small-deficit, turbulent wakes, *Journal of Fluid Me-*
1039 *chanics* 168 (1986) 31. doi:10.1017/S0022112086000289.
- 1040 [38] S. Cannon, F. Champagne, A. Glezer, Observations of large-scale struc-
1041 tures in wakes behind axisymmetric bodies, *Experiments in Fluids* 14
1042 (1993) 447–450. doi:10.1007/BF00190199.
- 1043 [39] T. Stallard, T. Feng, P. Stansby, Experimental study of the mean wake
1044 of a tidal stream rotor in a shallow turbulent flow, *Journal of Fluids and*
1045 *Structures* 54 (2015) 235–246. doi:10.1016/j.jfluidstructs.2014.10.017.
- 1046 [40] C. Jordan, D. Dundovic, A. K. Fragkou, G. Deskos, D. S. Coles, M. D.
1047 Piggott, A. Angeloudis, Combining shallow-water and analytical wake
1048 models for tidal array micro-siting, *Journal of Ocean Engineering and*
1049 *Marine Energy* 8 (2022) 193–215. doi:10.1007/s40722-022-00225-2.
- 1050 [41] S. B. Pope, *Turbulent Flows*, Cambridge University Press, 2000.
1051 doi:10.1017/CBO9780511840531.
- 1052 [42] H. Glauert, *A general theory of the autogyro* (1926).
- [43] H. Schlichting, K. Gersten, *Turbulent Boundary Layers Without Cou-
pling of the Velocity Field to the Temperature Field*, Springer Berlin
Heidelberg, 2017, pp. 557–610. doi:10.1007/978-3-662-52919-5_18.

- [44] E. Jump, A. Macleod, T. Wills, Review of tidal turbine wake modelling methods, *International Marine Energy Journal* 3 (2020) 91–100. doi:10.36688/imej.3.91-100.
- [45] O. D. Medina, F. G. Schmitt, R. Calif, G. Germain, B. Gaurier, Turbulence analysis and multiscale correlations between synchronized flow velocity and marine turbine power production, *Renewable Energy* 112 (2017) 314–327. doi:10.1016/j.renene.2017.05.024.
- [46] A. Kolmogorov, Dissipation of energy in the locally isotropic turbulence, *Proceedings of the Royal Society of London. Series A: Mathematical and Physical Sciences* 434 (1991) 15–17. doi:10.1098/rspa.1991.0076. URL <https://royalsocietypublishing.org/>
- [47] K. S. Choi, J. L. Lumley, The return to isotropy of homogeneous turbulence, *Journal of Fluid Mechanics* 436 (2001) 59–84. doi:10.1017/S002211200100386X. URL <https://www.cambridge.org/core/journals/journal-of-fluid-mechanics/article/re>
- [48] Z. Zhang, Y. Zhang, J. Zhang, Y. Zheng, W. Zang, X. Lin, E. Fernandez-Rodriguez, Experimental study of the wake homogeneity evolution behind a horizontal axis tidal stream turbine, *Applied Ocean Research* 111 (2021) 102644. doi:10.1016/J.APOR.2021.102644.
- [49] Y. Zhang, W. Zang, J. Zheng, L. Cappietti, J. Zhang, Y. Zheng, E. Fernandez-Rodriguez, The influence of waves propagating with the current on the wake of a tidal stream turbine, *Applied Energy* 290 (2021) 116729. doi:10.1016/J.APENERGY.2021.116729.
- [50] P. K. Raushan, S. K. Singh, K. Debnath, Turbulent anisotropy and length scale variation over multiple shaped structure, *Journal of Offshore Mechanics and Arctic Engineering* 145 (12 2023). doi:10.1115/1.4057007/1159795. URL <https://dx.doi.org/10.1115/1.4057007>
- [51] E. Branlard, *Theoretical Foundations for Flows Involving Vorticity*, Springer, 2017, pp. 11–82. doi:10.1007/978-3-319-55164-7₂.